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APPENDIX 1

DERIVATION OF INDUCTION MOTOR STEADY STATE CHARACTERISTIC

In steady state condition, the electrical torque T_e and mechanical torque T_m is equal. Thus we can write

$$T_e - T_m = 0 \quad (A1.1)$$

where

$$T_e = \frac{\frac{\omega_e}{\omega_b} X_M^2 r_s |V_s|^2}{\left[r_s r_r + s \left(\frac{\omega_e}{\omega_b} \right)^2 (X_M^2 - X_{ss} X_{rr}) \right]^2 + \left(\frac{\omega_e}{\omega_b} \right)^2 (r_r X_{ss} + s r_s X_{rr})^2} \quad (A1.2)$$

$$T_m = a + b * w_r + c * w_r^{**2} \quad (A1.3)$$

or in terms of slip

$$T_m = a + b * (1-s) + c * (1-s)^{**2} \quad (A1.4)$$

Substitute (A1.2) and (A1.4) into (A1.1) and solving (A1.1) we can derive operating slip at given voltage and frequency. The method used here for solving (A1.1) is False Position Method. This method is used to find a real root of an equation written in the form :

$$f(x) = 0 \quad (A1.5)$$

First we must find x_1 and x_2 that give opposite

sign of $f(x_1)$ and $f(x_2)$, then calculate x_3 which is closer to the root by :

$$x_3 = x_1 + f(x_1)/(f(x_1)-f(x_2)) * (x_2-x_1) \quad (\text{A1.6})$$

Equation (A1.6) is then applied again with x_3 substituting either x_1 or x_2 which give $f(x)$ that is the same sign as $f(x_3)$. Repeating until $f(x_3)$ is less enough, thus x_3 is an approximate root of (A1.5).

To apply this method to (A1.1), we use the slip at maximum torque s_m and $\text{slip} = 0$, which will give opposite sign of (A1.1).

The slip at maximum torque can be written as :

$$s_m = r_r G$$

$$G = \left[\frac{(\omega_e/\omega_b)^{-2} r_s^2 + X_{ss}^2}{(X_M^2 - X_{ss} X_{rr})^2 \left(\frac{\omega_e}{\omega_b} \right)^2 + r_s^2 X_{rr}^2} \right]^{1/2}$$

Once a slip at given voltage and frequency is derived, we can calculate the input impedance of the equivalent circuit shown in Figure A1.1 by the following equation :

$$Z = \frac{\frac{r_s r_r}{s} + \left(\frac{\omega_e}{\omega_b} \right)^2 (X_M^2 - X_{ss} X_{rr}) + j \frac{\omega_e}{\omega_b} \left(\frac{r_r}{s} X_{ss} + r_s X_{rr} \right)}{\frac{r_r}{s} + j \frac{\omega_e}{\omega_b} X_{rr}}$$

Finally, the active and reactive power can be calculated as :

$$P = \text{Re} [(V^{**2}/Z^*)]$$

$$Q = \text{Im} [(V^{**2}/Z^*)]$$

APPENDIX 2

DERIVATION OF INDUCTION MOTOR DYNAMIC CHARACTERISTIC

Neglecting stator transients the per unit voltage equations in d, q-axis are :

$$v_{qs} = R_s * X_{rr} / D * F_{qs} + w_e / w_b * F_{ds} - R_s * X_m / D * F_{qr} \quad (A2.1)$$

$$v_{ds} = R_s * X_{rr} / D * F_{ds} - w_e / w_b * F_{qs} - R_s * X_m / D * F_{dr} \quad (A2.2)$$

$$\begin{aligned} v_{qr} &= R_r * X_{ss} / D * F_{qr} + p F_{qr} / w_b - R_r * X_m / D * F_{qs} \\ &+ (w_e - w_r) / w_b * F_{dr} \end{aligned} \quad (A2.3)$$

$$\begin{aligned} v_{dr} &= R_r * X_{ss} / D * F_{dr} + p F_{dr} / w_b - R_r * X_m / D * F_{ds} \\ &- (w_e - w_r) / w_b * F_{qr} \end{aligned} \quad (A2.4)$$

where

$$X_{ss} = X_s + X_m$$

$$X_{rr} = X_r + X_m$$

$$D = X_{ss} * X_{rr} - X_m^{**2}$$

The equation of motion is :

$$2 * H * p_w r / w_b = T_e - T_m \quad (A2.5)$$

where

$$T_e = X_m/D * (F_{qs} * F_{dr} - F_{qr} * F_{ds})$$

$$T_m = a + b * w_r + c * w_r^{**2}$$

The corresponding equations are :

$$F_{qs} = X_s * i_{qs} + X_m * (i_{qs} + i_{qr}) \quad (A2.6)$$

$$F_{ds} = X_s * i_{ds} + X_m * (i_{ds} + i_{dr}) \quad (A2.7)$$

$$F_{qr} = X_r * i_{qr} + X_m * (i_{qs} + i_{qr}) \quad (A2.8)$$

$$F_{dr} = X_r * i_{dr} + X_m * (i_{ds} + i_{dr}) \quad (A2.9)$$

and

$$v_{qs} = R_s * i_{qs} + w_e / w_b * F_{ds} \quad (A2.10)$$

$$v_{ds} = R_s * i_{ds} - w_e / w_b * F_{qs} \quad (A2.11)$$

$$v_{qr} = R_r * i_{qr} + (w_e - w_r) / w_b * F_{dr} + p F_{qr} / w_b \quad (A2.12)$$

$$v_{dr} = R_r * i_{dr} - (w_e - w_r) / w_b * F_{qr} + p F_{dr} / w_b \quad (A2.13)$$

To develop induction motor model we assume that

- stator transients are negligible
- rotor is short circuit, thus $v_{dr} = v_{qr} = 0$
- resistances and reactances are constant.

The differential equations of the rotor circuits and the differential equation of motion are solved in the

present work by numerical method based on the trapezoidal rule. At each time interval these equations together with the stator voltage equations can be solved iteratively beginning with known initial condition in the steady state. Note that the zero position of the reference frame is selected so that $v_{ds} = 0$.

Initial condition can be derived from active and reactive power which can be calculated by the method presented in Appendix 1. Beginning with the active and reactive power in the steady state P_0, Q_0 the following initial values can be derive :

$$i_{qs0} = (P_0 - v_{ds0} * i_{ds0}) / v_{qs0} \quad (A2.14)$$

$$i_{ds0} = (Q_0 + v_{ds0} * i_{qs0}) / v_{qs0} \quad (A2.15)$$

From (A2.14) - (A2.15) and (A2.1) - (A2.9) with neglecting the differential terms due to steady state condition one can find the initial values of $F_{qs0}, F_{ds0}, F_{qro}, F_{dro}$, and w_{ro} .

At each time step i_{ds} and i_{qs} are calculated by (A2.10) and (A2.11) and then the active and reactive power can be calculated as :

$$P = v_{qs} * i_{qs} + v_{ds} * i_{ds} \quad (A2.16)$$

$$Q = v_{qs} * i_{ds} - v_{ds} * i_{qs} \quad (A2.17)$$

APPENDIX 3

MOTOR PARAMETERS APPROXIMATION

The method used here is similar to that used in ref. [15]. However, it is different in finding leakage reactances. The purpose is to make the method simple as possible.

The input and output power of an induction motor in per unit is :

$$P_{in} = V_s * I_s * \cos \phi \quad (A3.1)$$

$$P_o = T(1-s) \quad (A3.2)$$

$$T = (I_r ** 2) * Rr/s \quad (A3.3)$$

According to ref.[15], we make the approximation that :

$$I_r = I_s * \cos \phi \quad (A3.4)$$

Input and output power are related by the following equation :

$$(I_s ** 2) * (\cos^2 \phi) * (Rr/s) * (1-s) / n = V_s * I_s * \cos \phi \quad (A3.5)$$

In steady state, we assume that $V_s = 1.0$ pu. and $I_s = 1.0$ pu.

From these equations we can obtain :

$$R_r = \eta * s / [(1-s) * \cos \phi]$$

$$X_m = \eta / [(1-s) * \sin \phi]$$

$$R_s = (\cos \phi) * [1 - \eta / (1-s)]$$

To find X_s and X_r we assume that the two values are equal. The starting current I_{st} is assumed by :

$$I_{st} = V_s / (X_s + X_r)$$

where I_{st} is obtained from manufacturer data. Thus

$$X_s = X_r = V_s / (2 * I_{st})$$

The inertia constant is :

$$H = (0.5 * J * w_s ** 2) / P_B$$

where

J is moment of inertia

w_s is synchronous speed

P_B is base power

APPENDIX 4

COMPUTER SIMULATION RESULTS

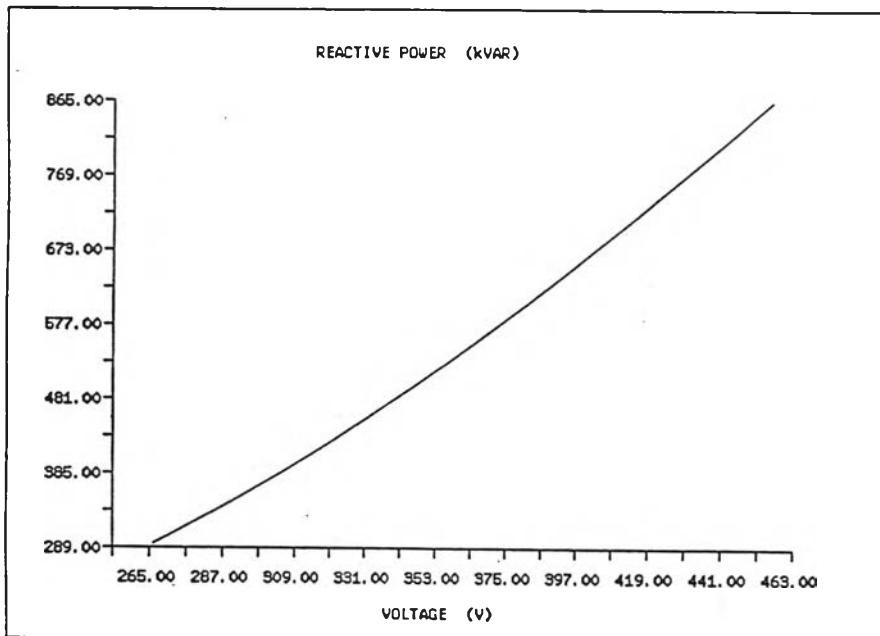
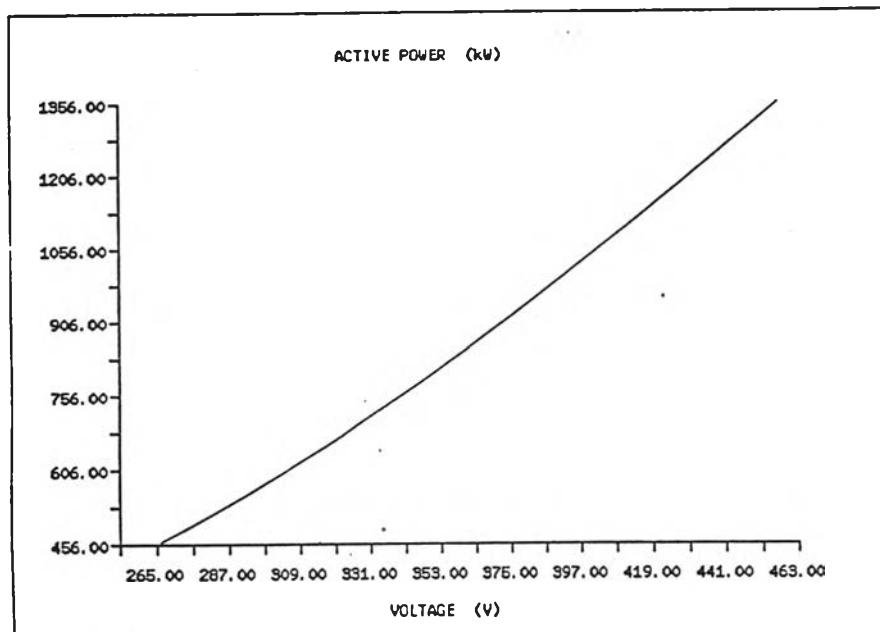


Fig. A4.1 Steady state voltage characteristic of
load case 1

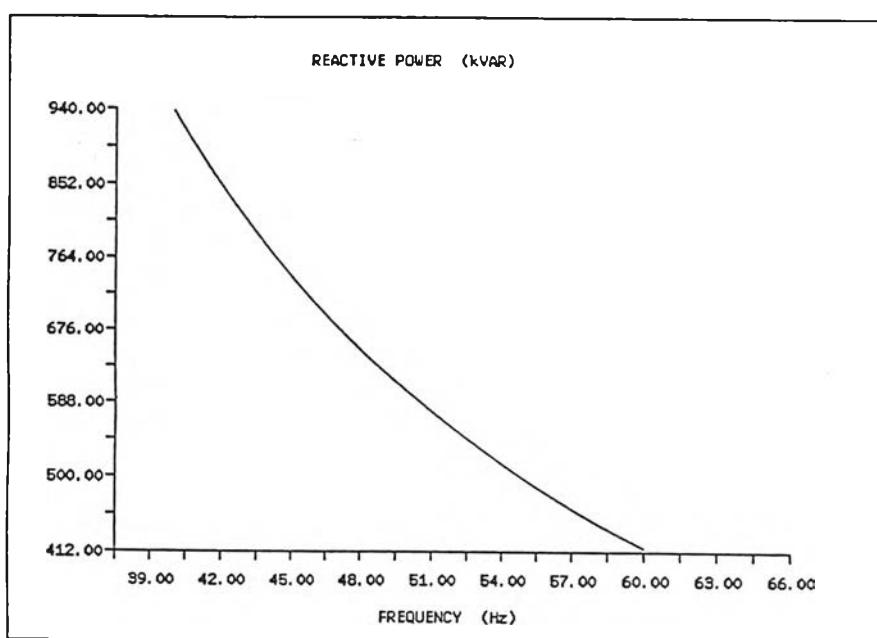
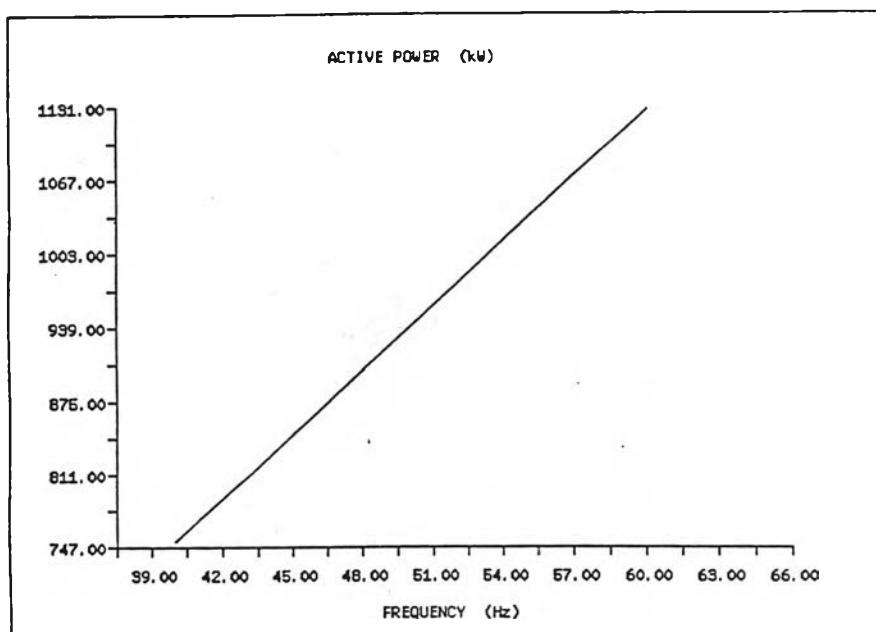


Fig. A4.2 Steady state frequency characteristic
of load case 1

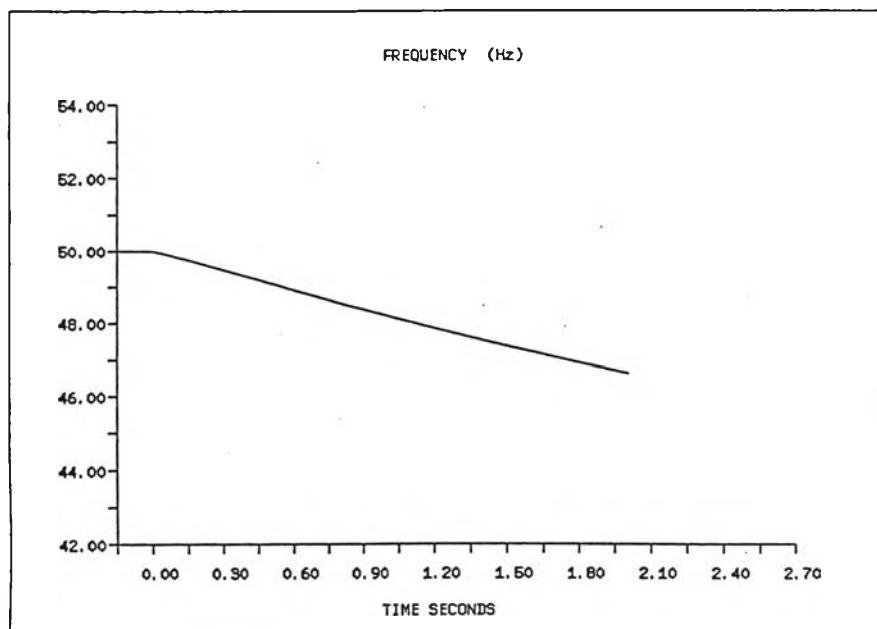
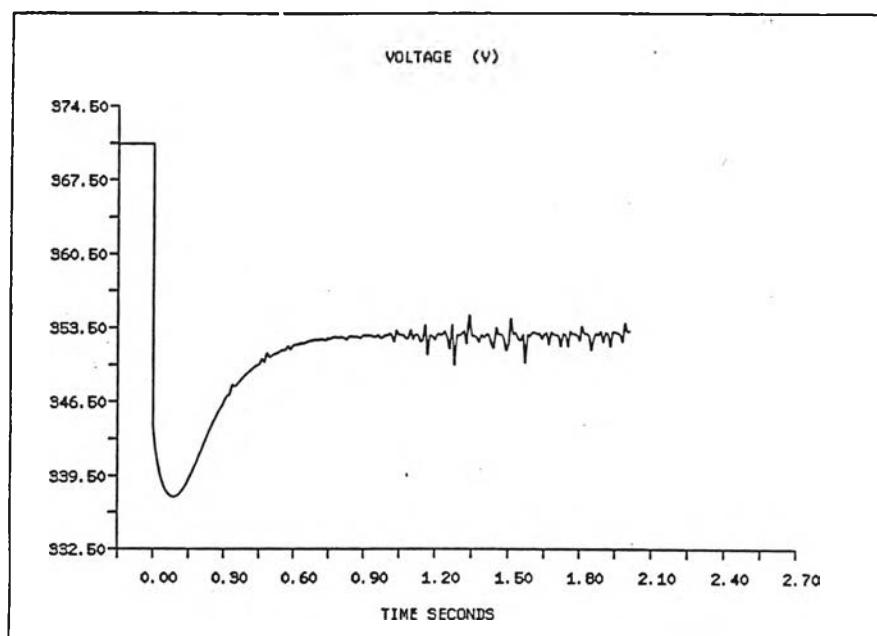


Fig. A4.3 Voltage and frequency at bus 3 during
gen. 2 disconnected

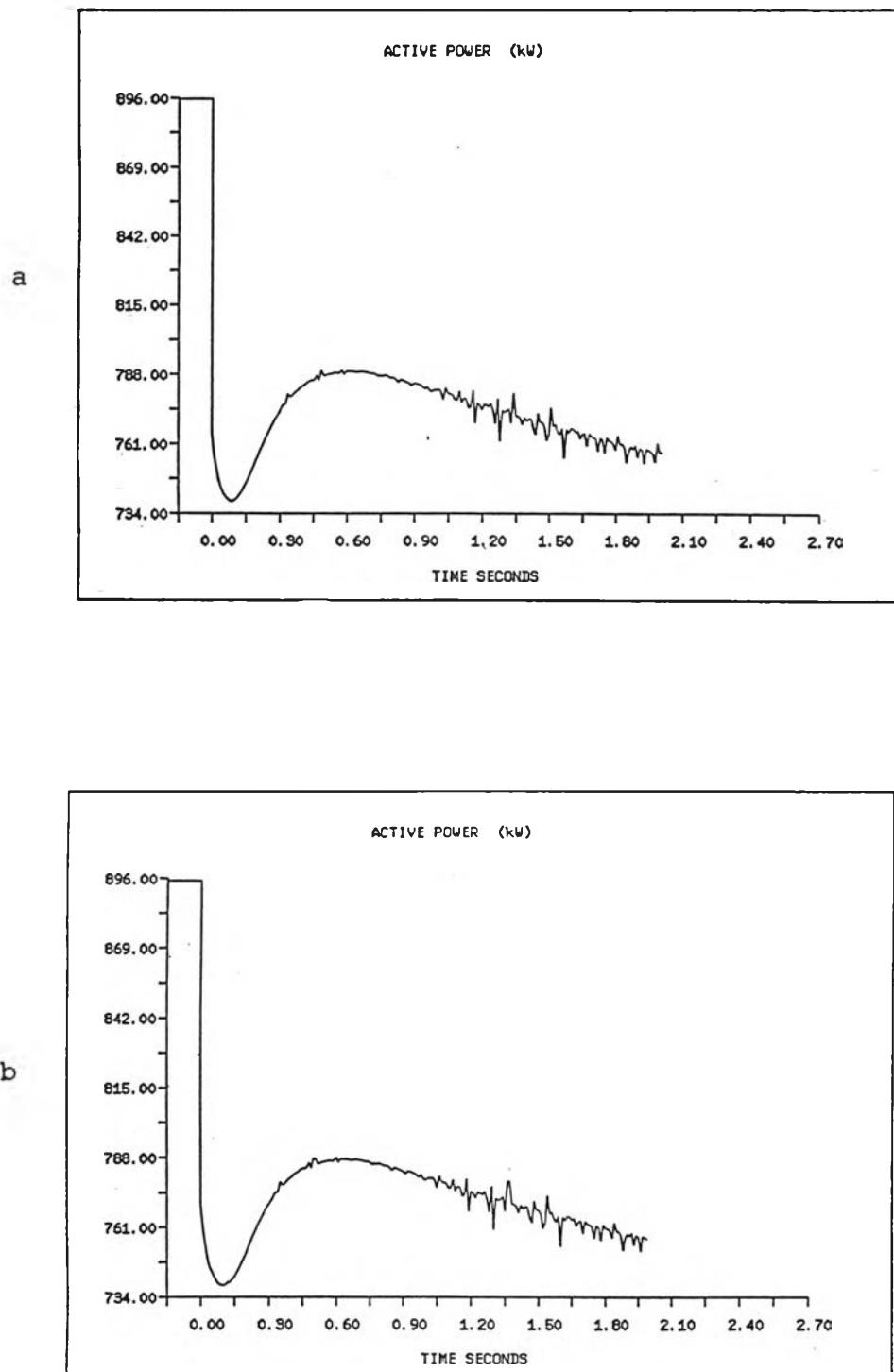


Fig. A4.4 Active power of load case 1 during
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- a. from SIMPOW
- b. from developed programme

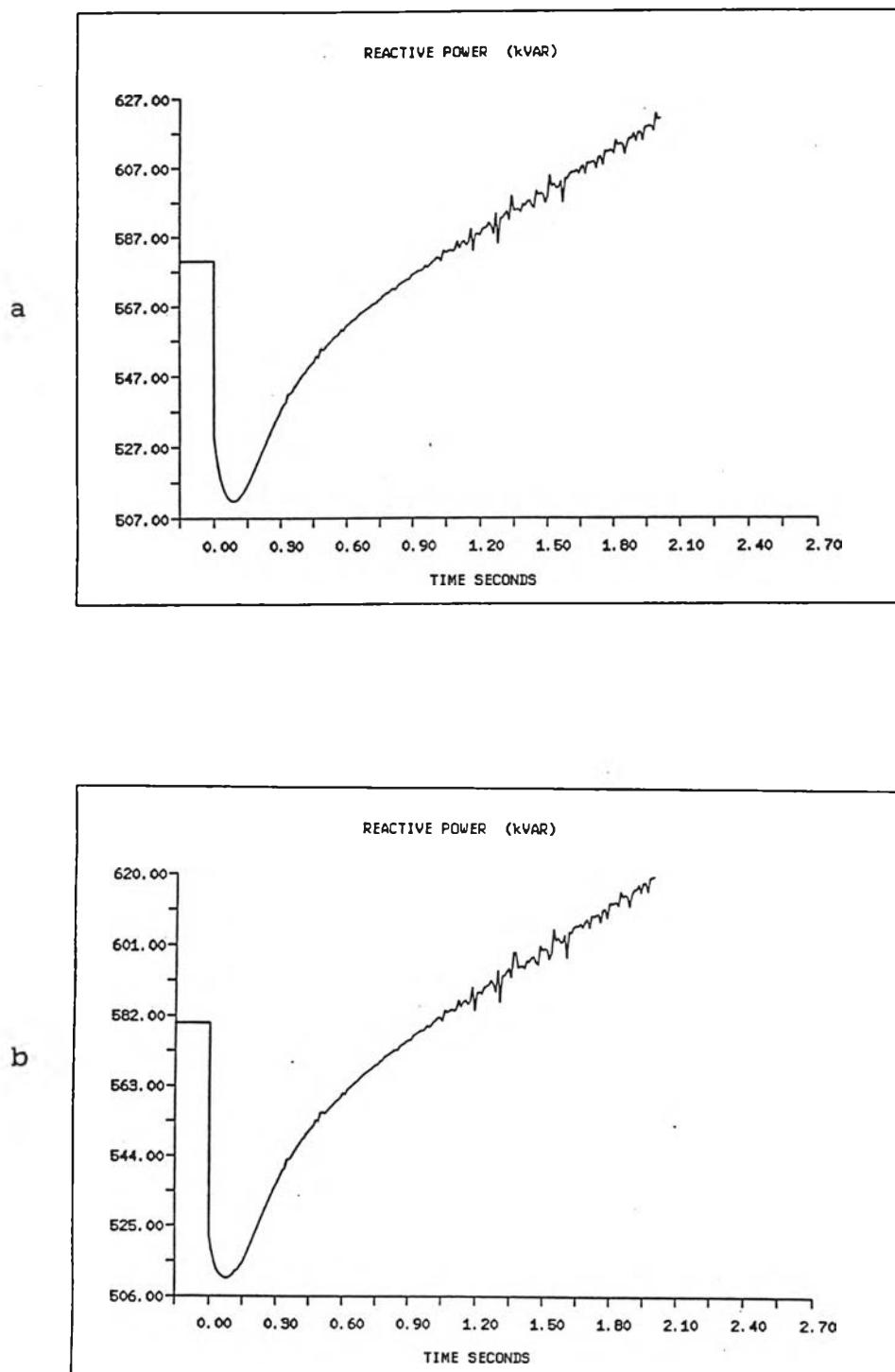


Fig. A4.5 Reactive power of load case 1 during
gen. 2 disconnected

- a. from SIMPOW
- b. from developed programme

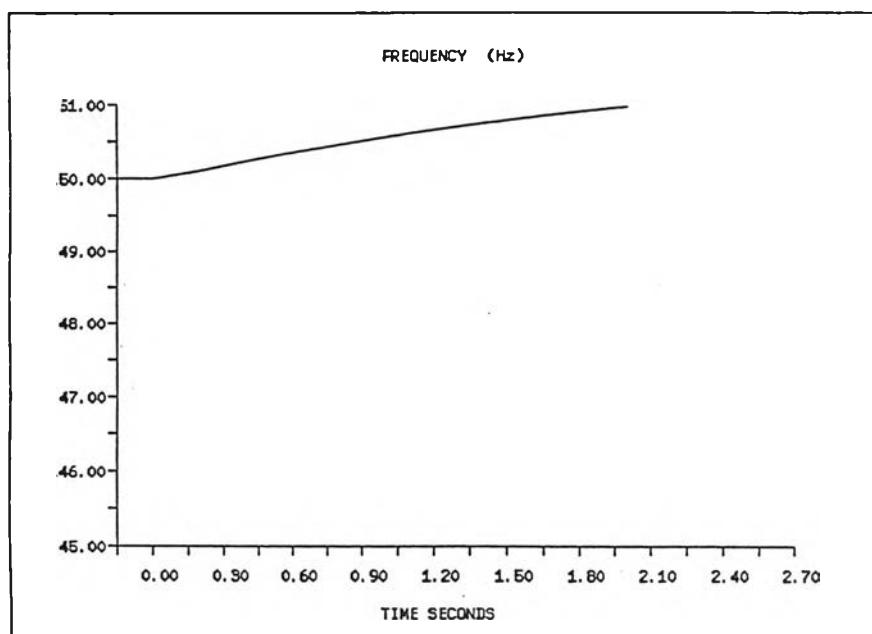
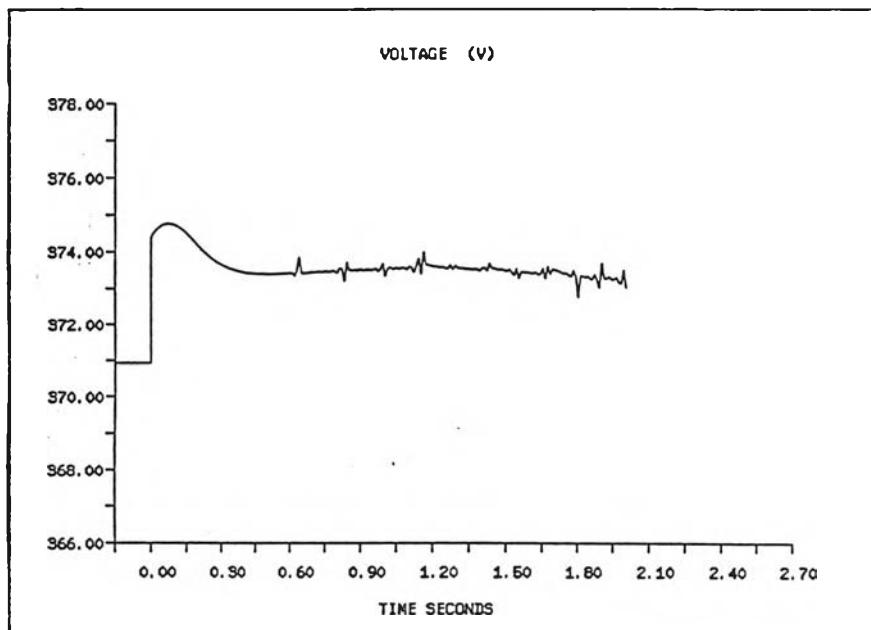


Fig. A4.6 Voltage and frequency at bus 3 during load at bus 2 disconnected

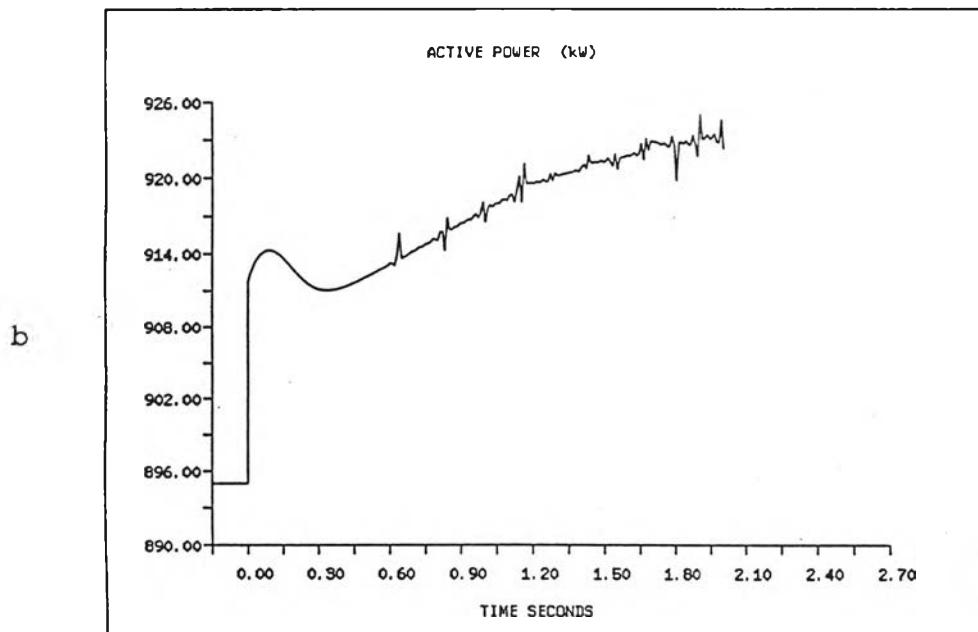
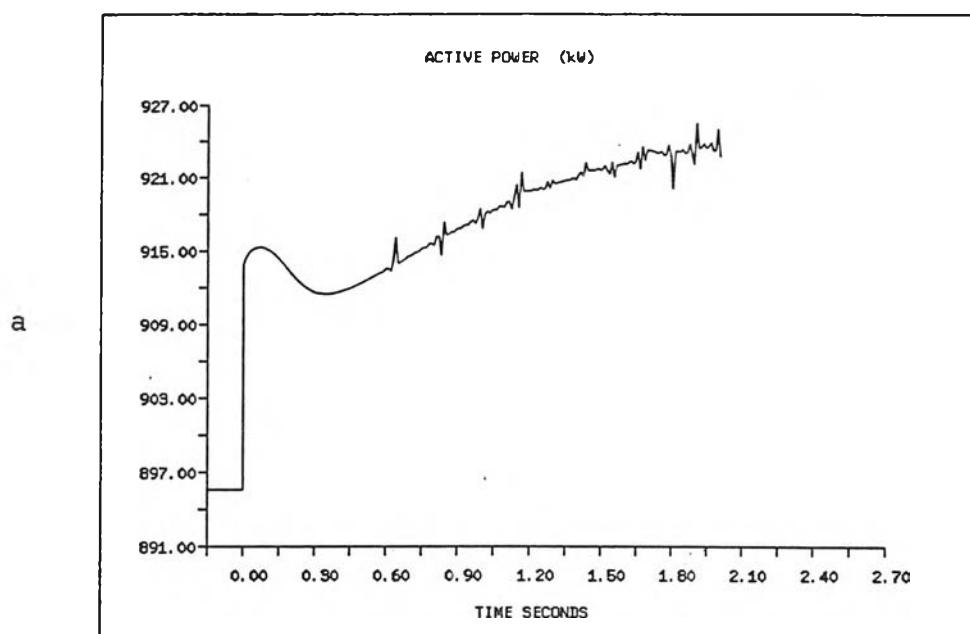


Fig. A4.7 Active power at bus 3 during load at bus 2 disconnected

- a. from SIMPOW
- b. from developed programme

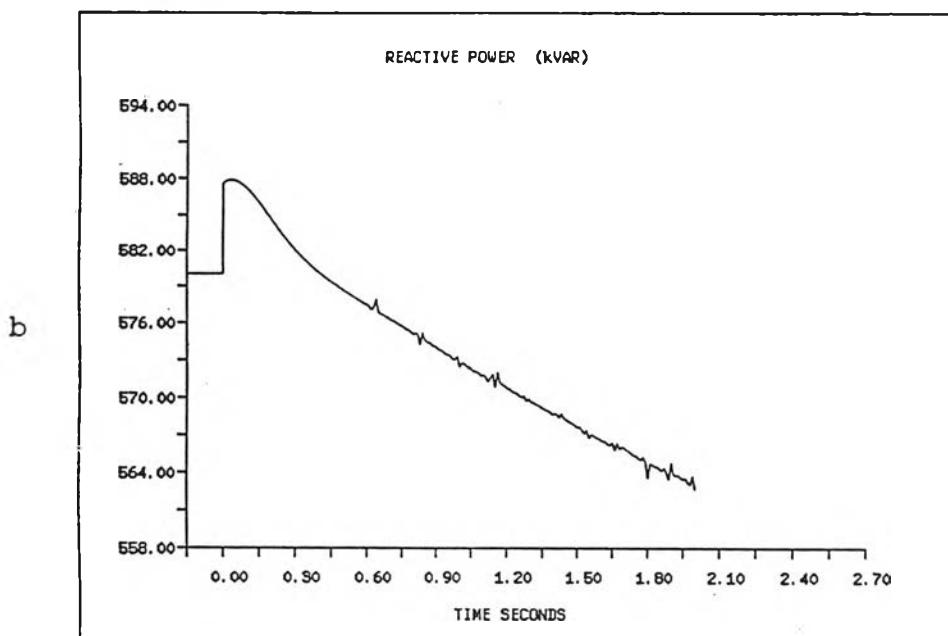
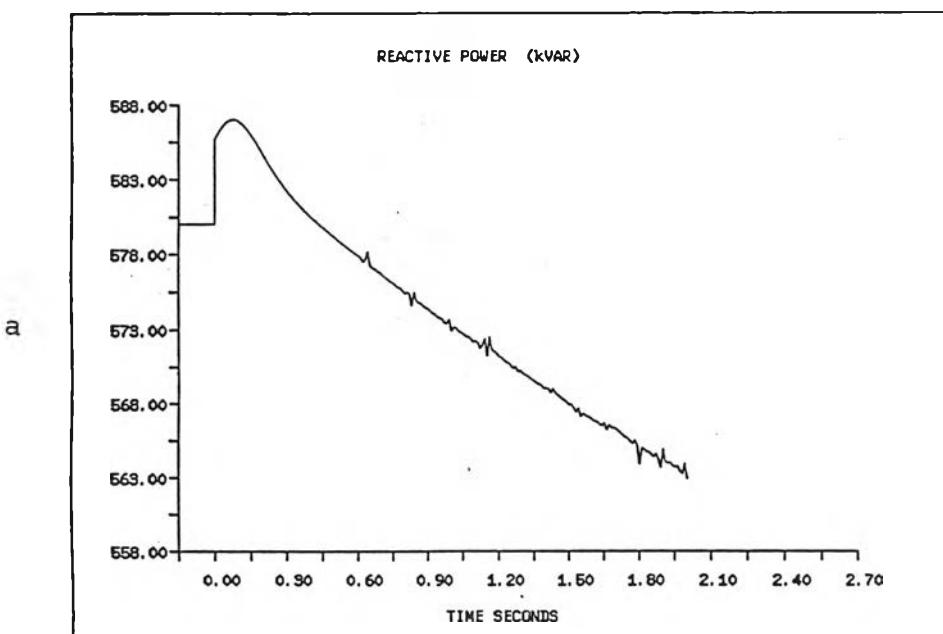


Fig. A4.8 Reactive power at bus 3 during load
at bus 2 disconnected

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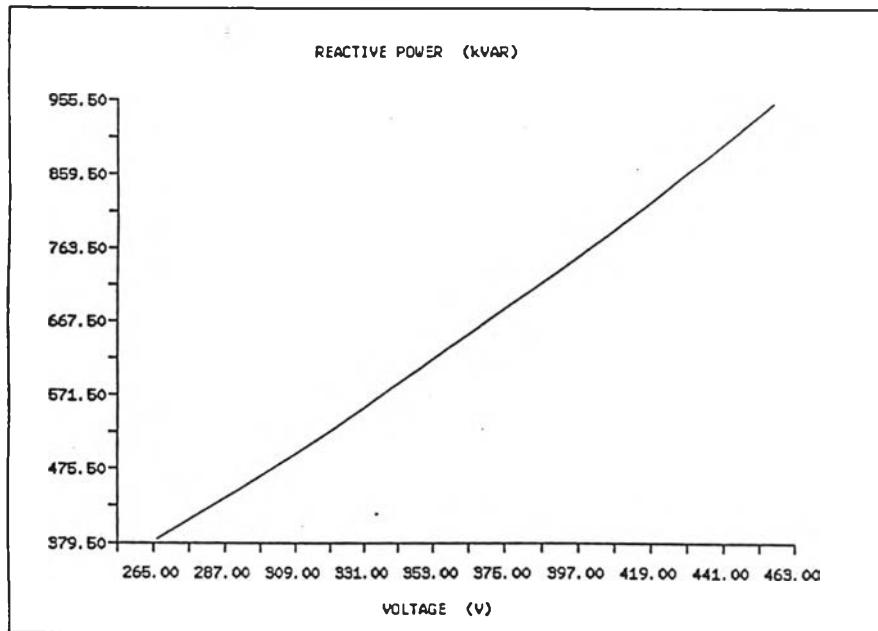
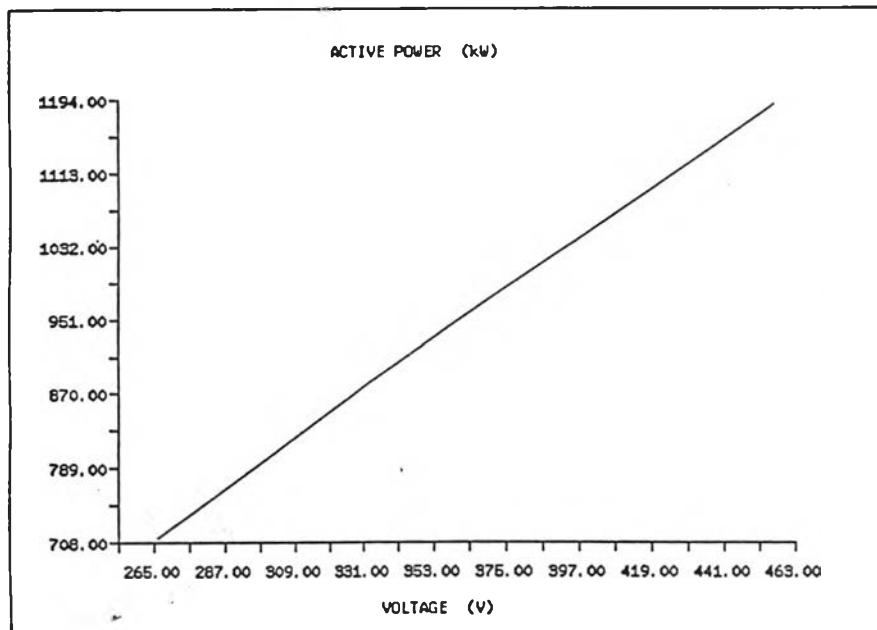


Fig. A4.9 Steady state voltage characteristic of
load case 2 used aggregate model

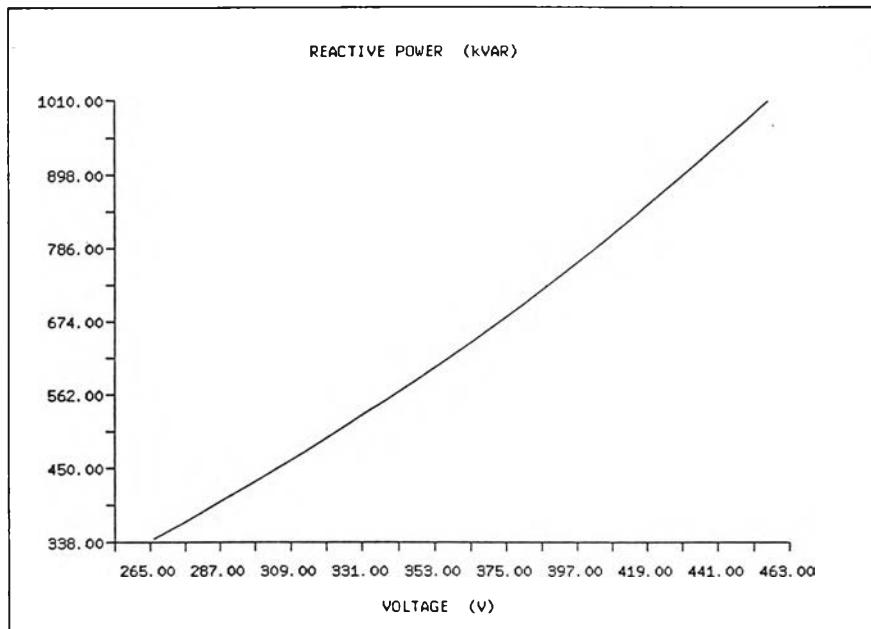
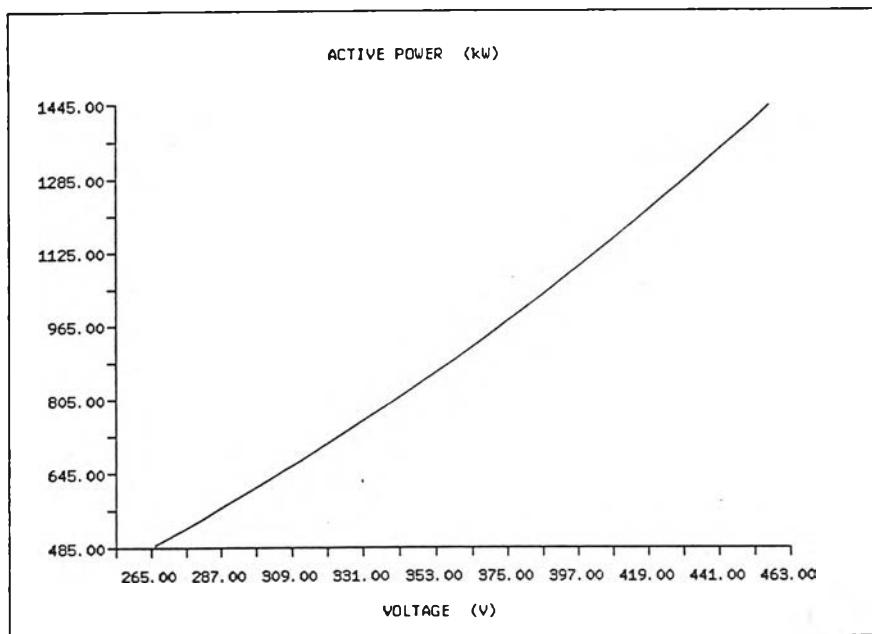


Fig.A4. 10 Steady state voltage characteristic of
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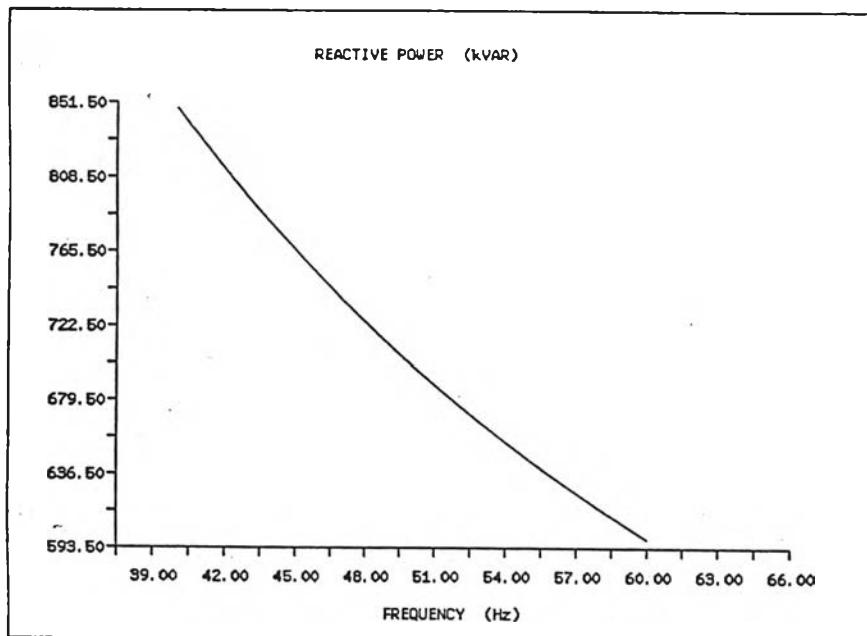
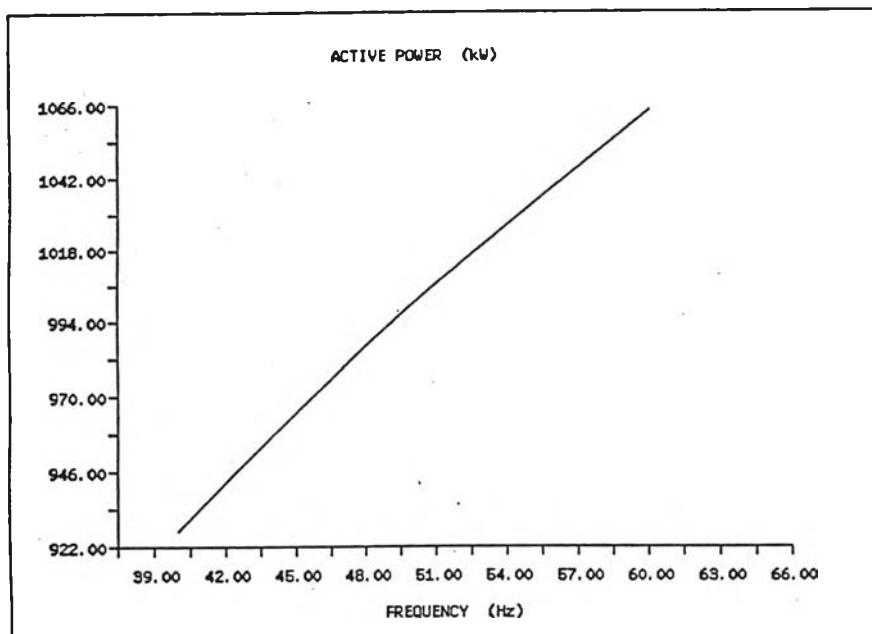


Fig. A4.11 Steady state frequency characteristic
of load case 2 used aggregate model

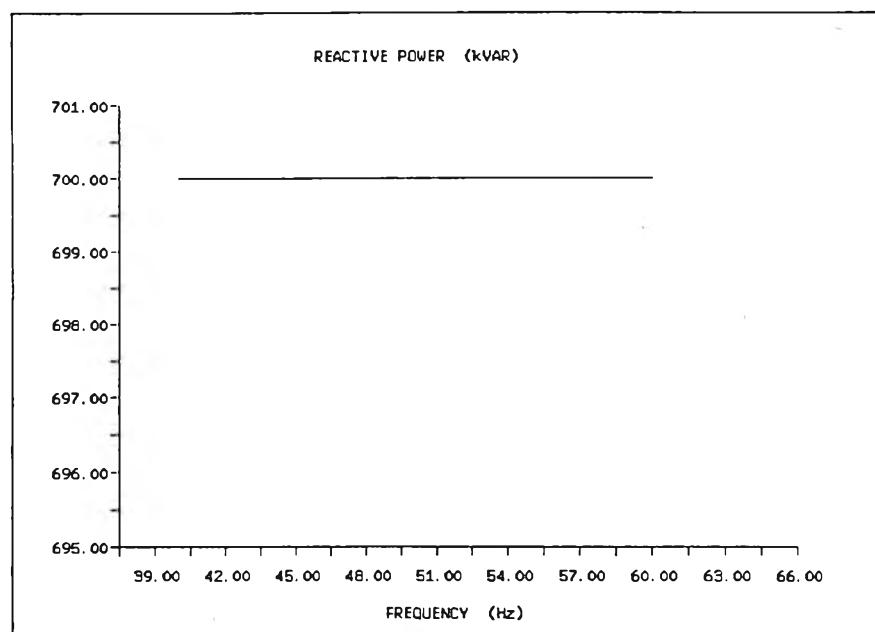
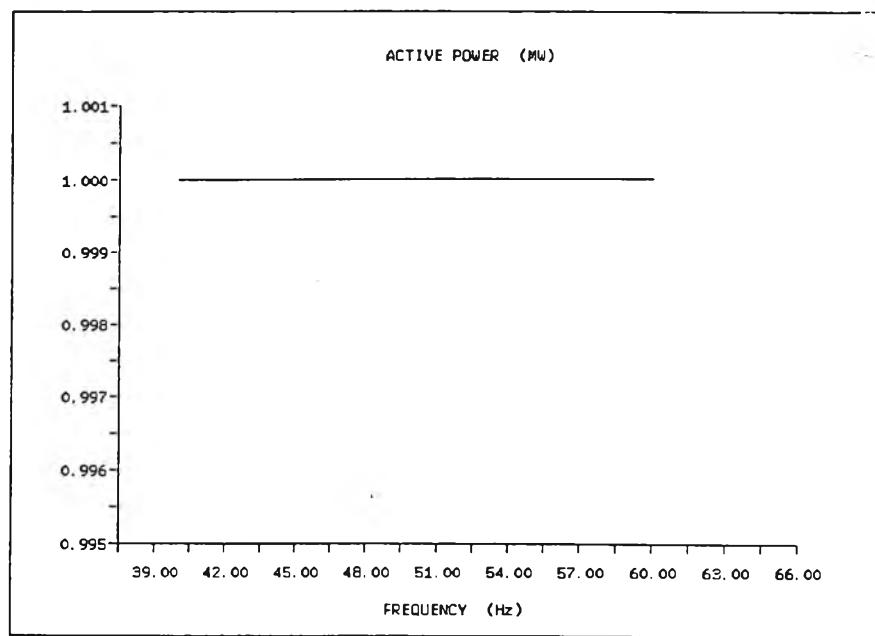


Fig. A4.12 Steady state frequency characteristic
of load case 2 used constant impedance
model

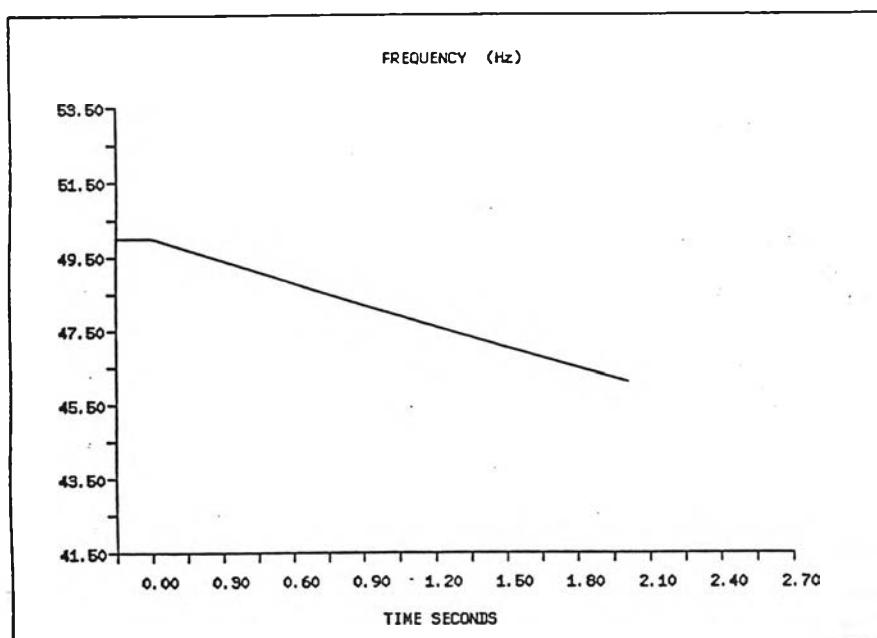
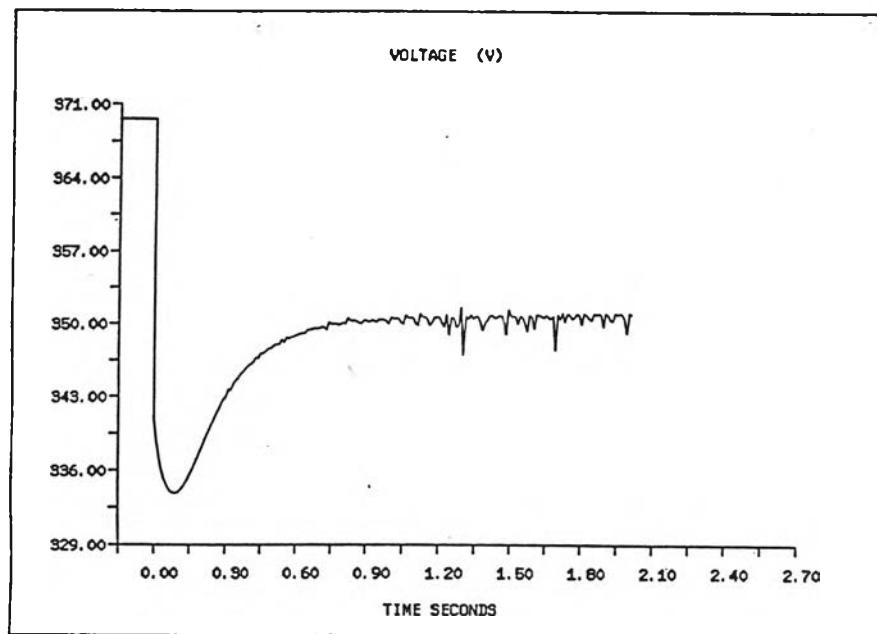


Fig. A4.13 Voltage and frequency at bus 3 during gen. 2 disconnected

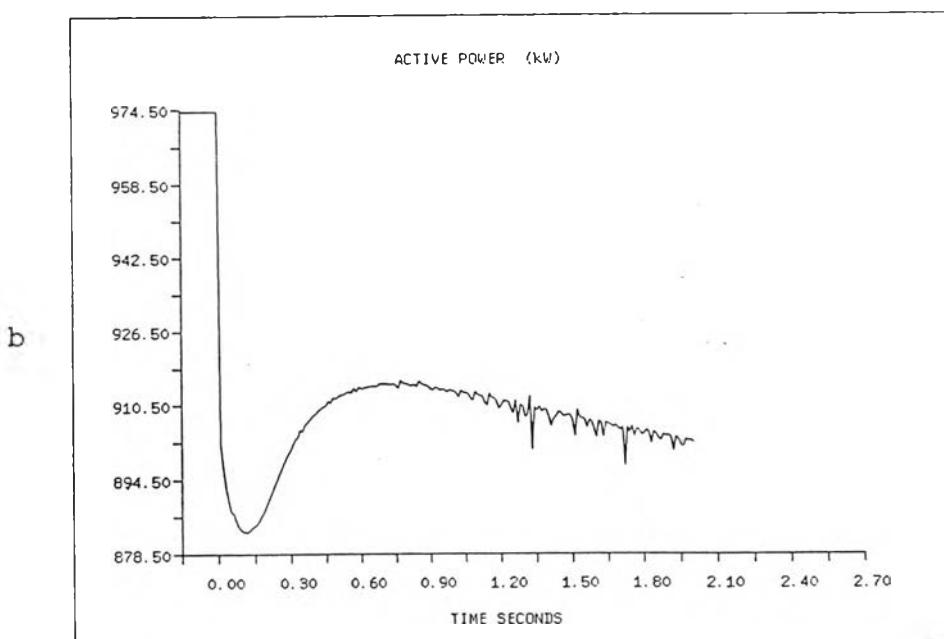
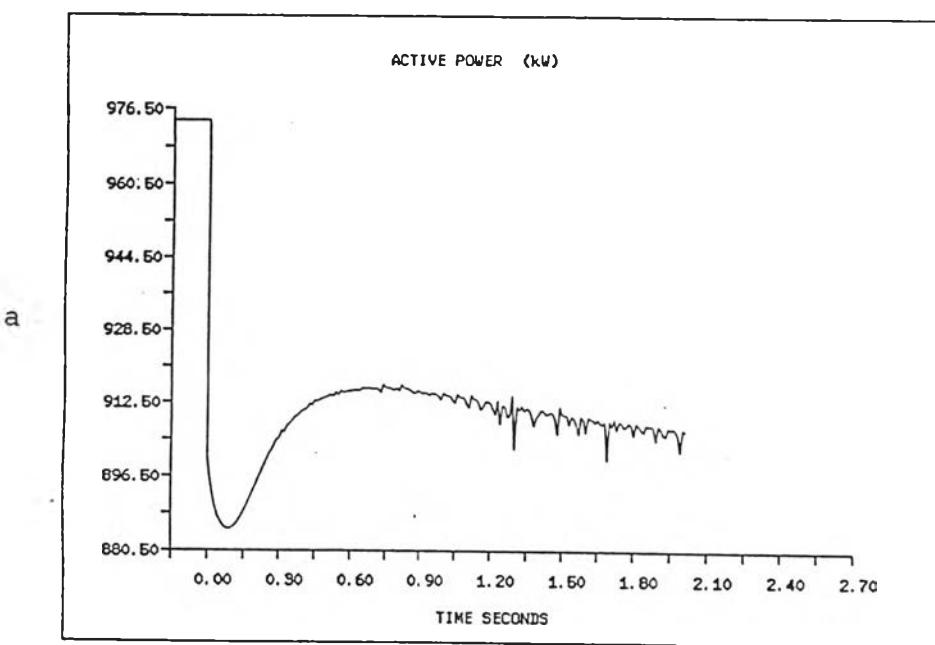


Fig. A4.14 Active power at bus 3 during gen. 2 disconnected

- a. from SIMPOW based on individual representation
- b. from developed programme based on aggregate model

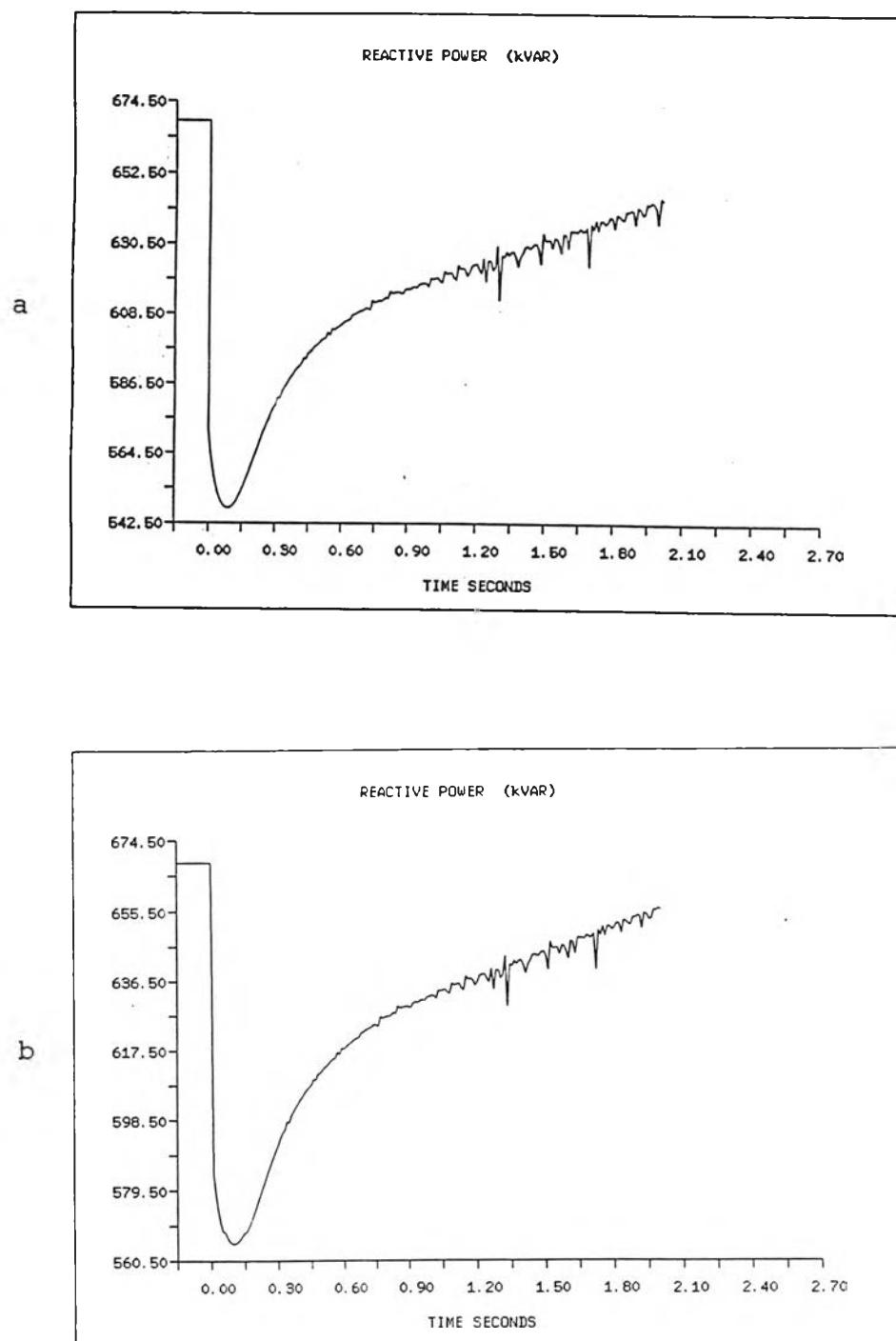


Fig. A4.15 Reactive power at bus 3 during
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- a. from SIMPOW based on individual representation
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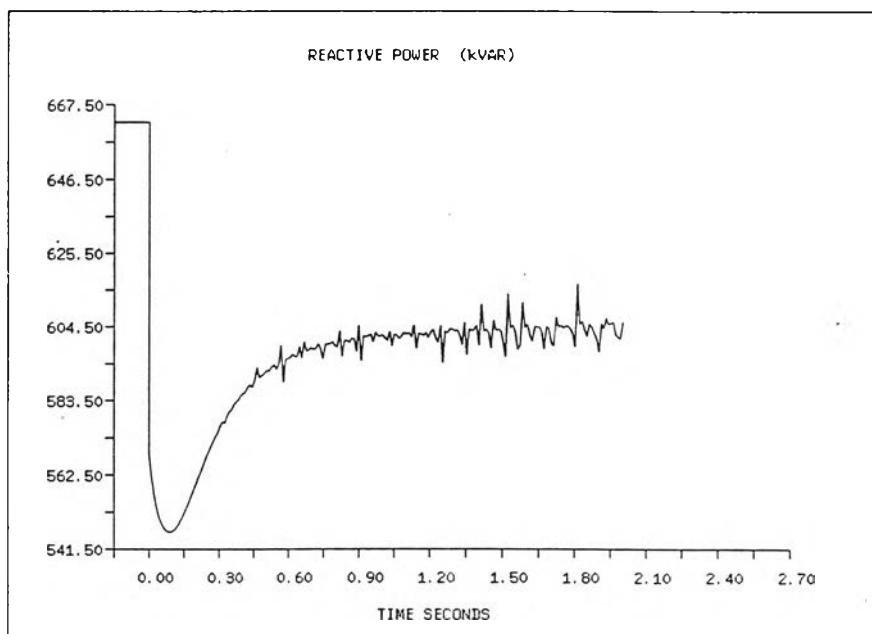
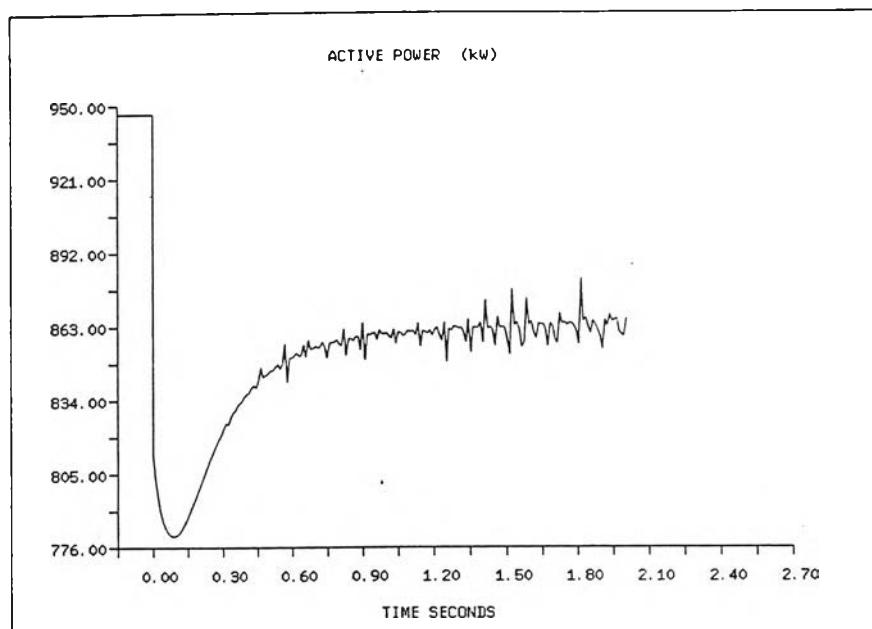


Fig. A4.16 Active and reactive power during gen. 2
disconnected used constant impedance model

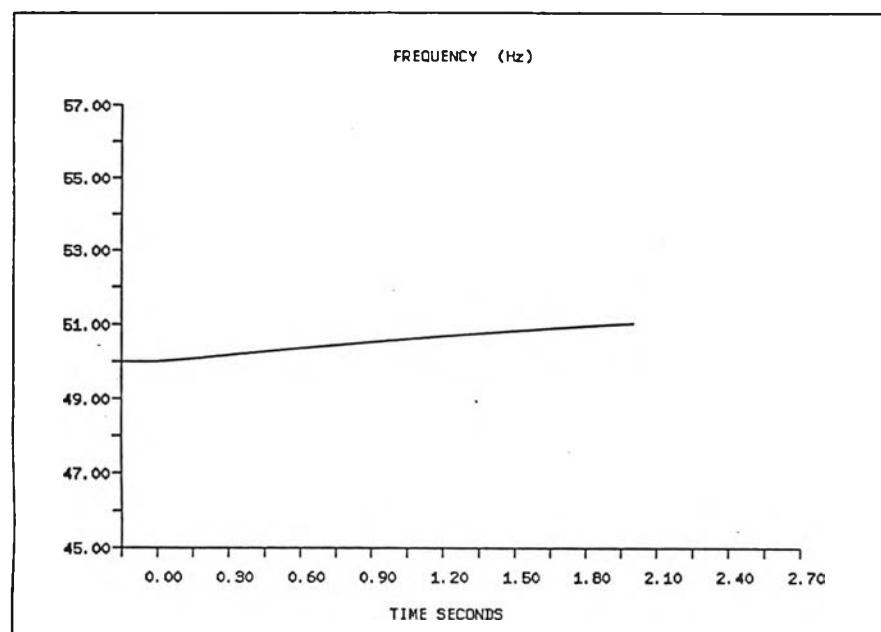
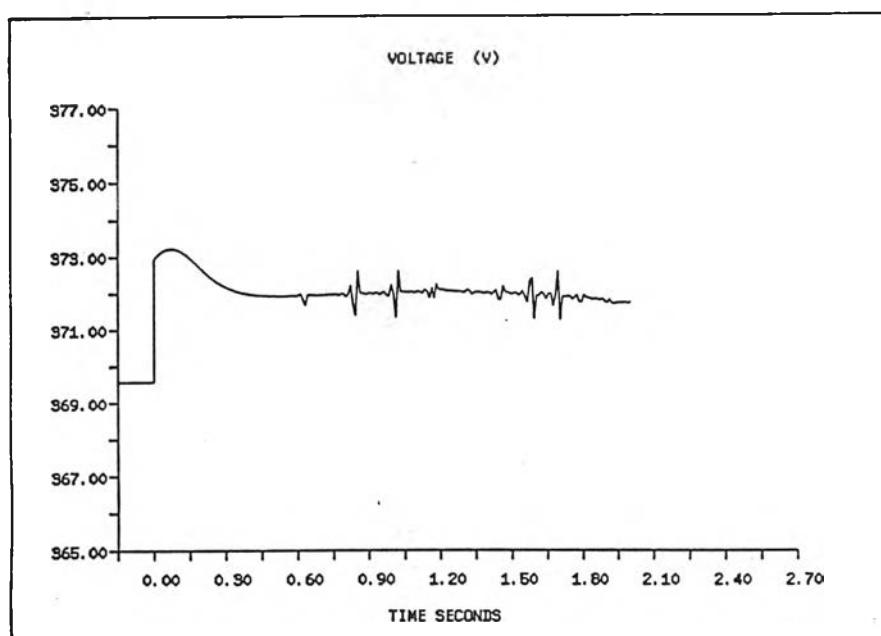


Fig. A4.17 Voltage and frequency at bus 3 during load bus 2 disconnected

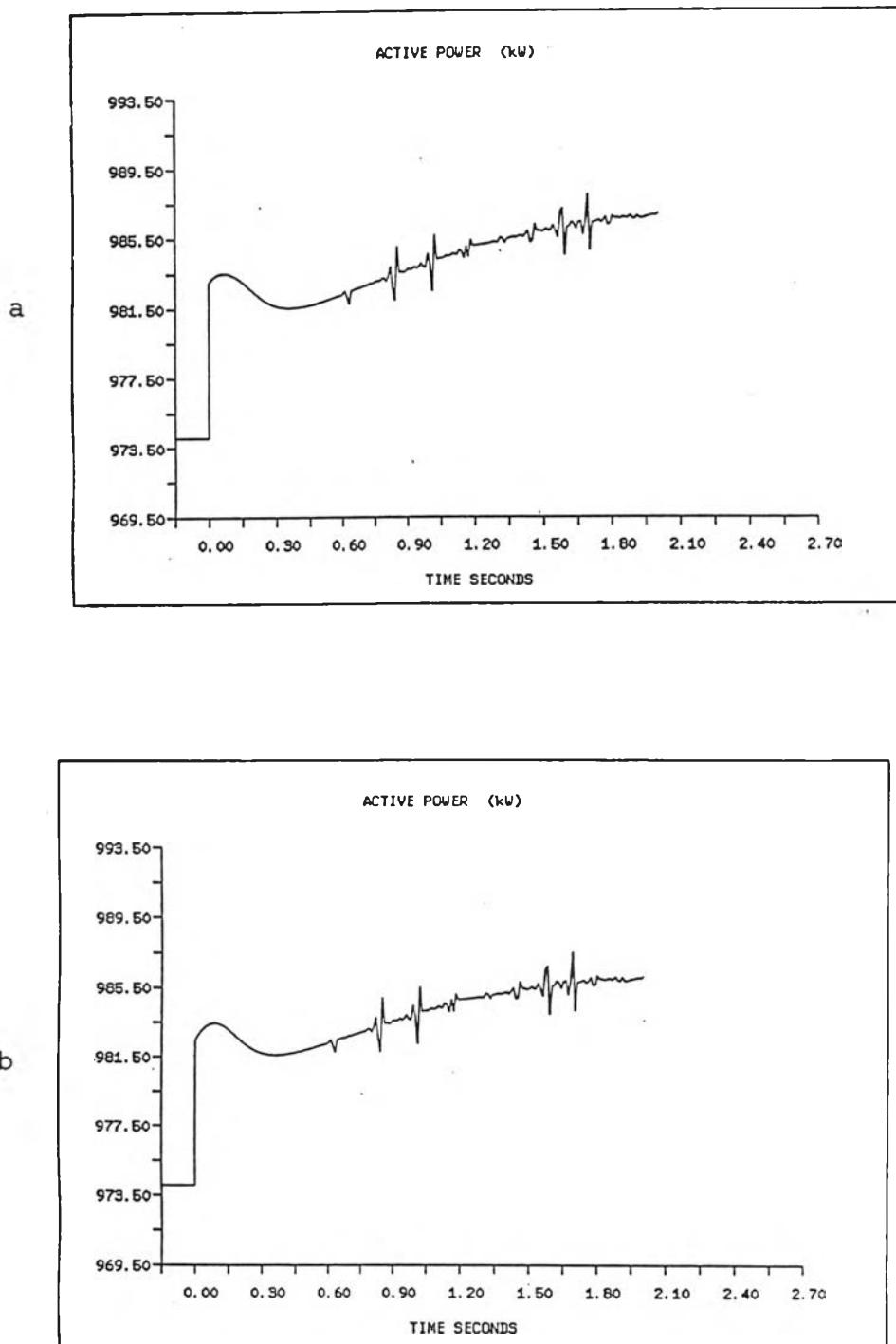


Fig. A4.18 Active power at bus 3 during load bus 2 disconnected

- a. from SIMPOW based on individual representation
- b. from developed programme based on aggregate model

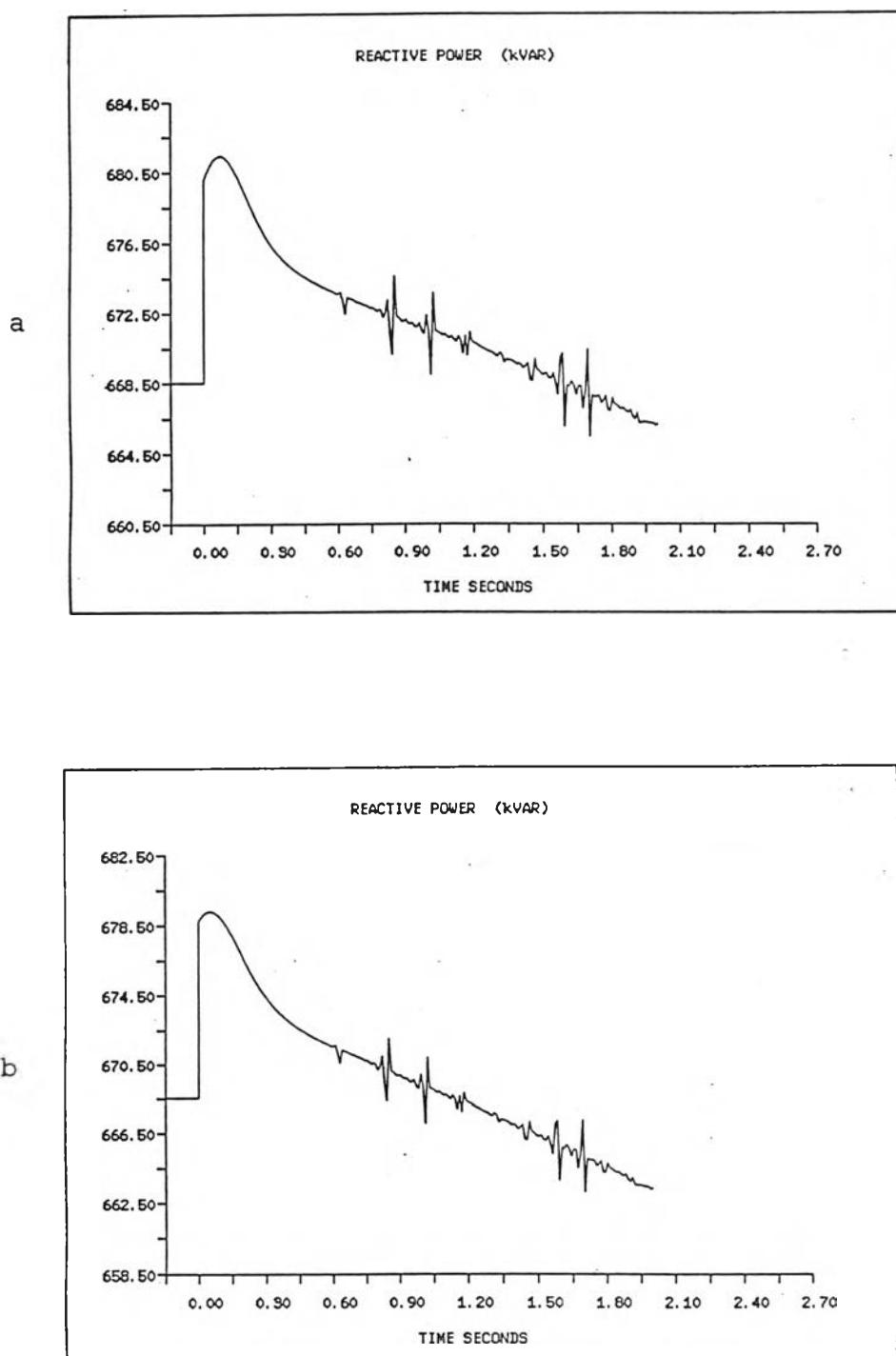


Fig. A4.19 Reactive power at bus 3 during load bus 2 disconnected

- a. from SIMPOW based on individual representation
- b. from developed programme based on aggregate model

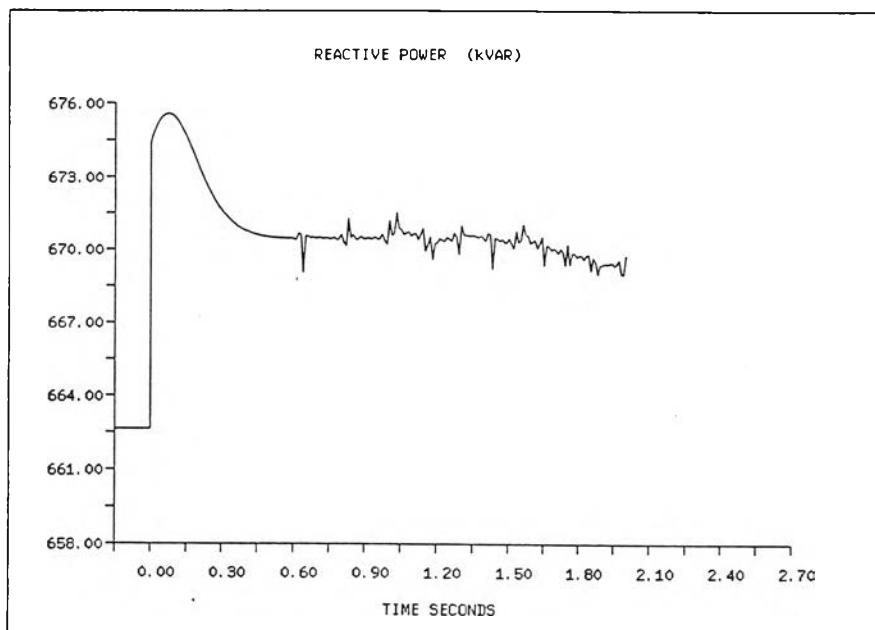
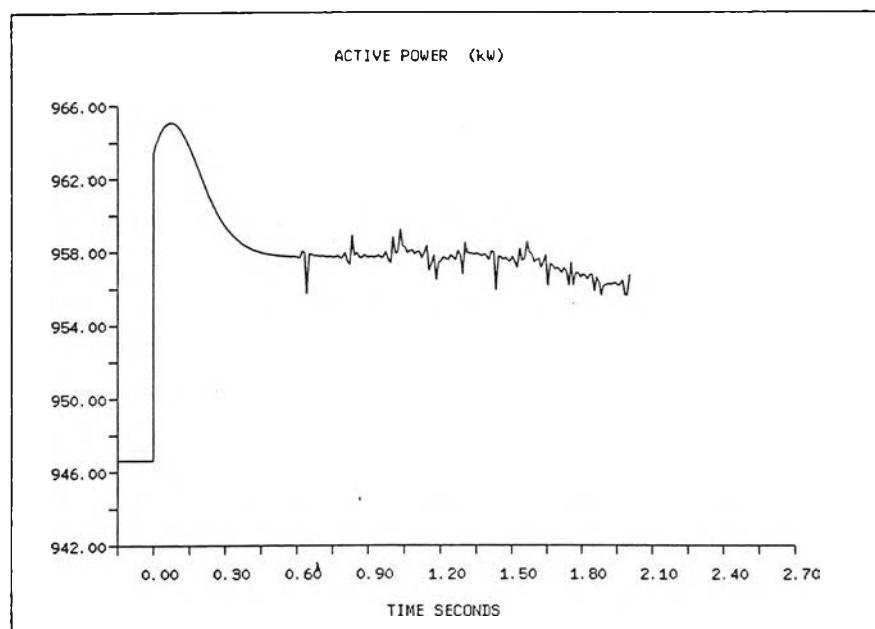


Fig. A4.20 Active and reactive power during load bus 2 disconnected used constant impedance model

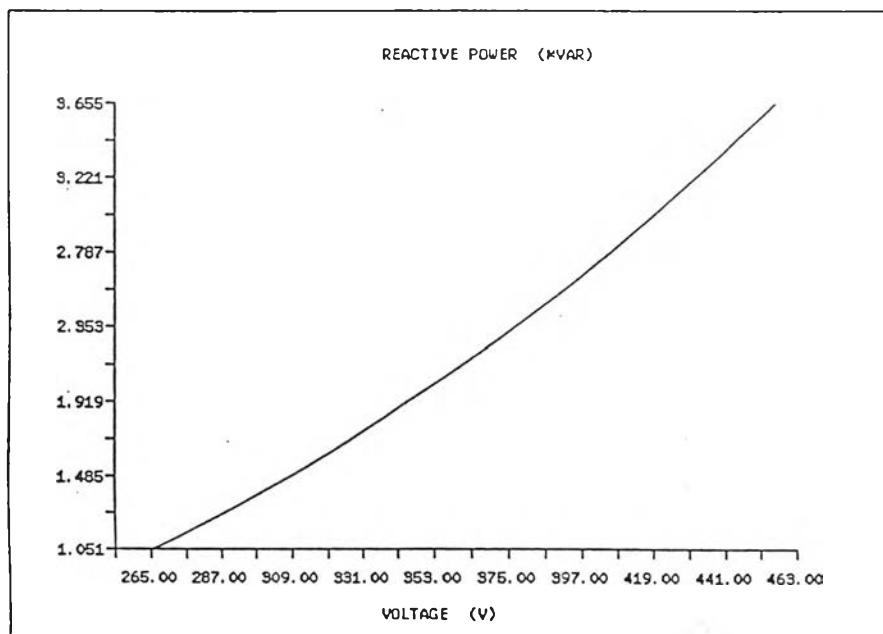
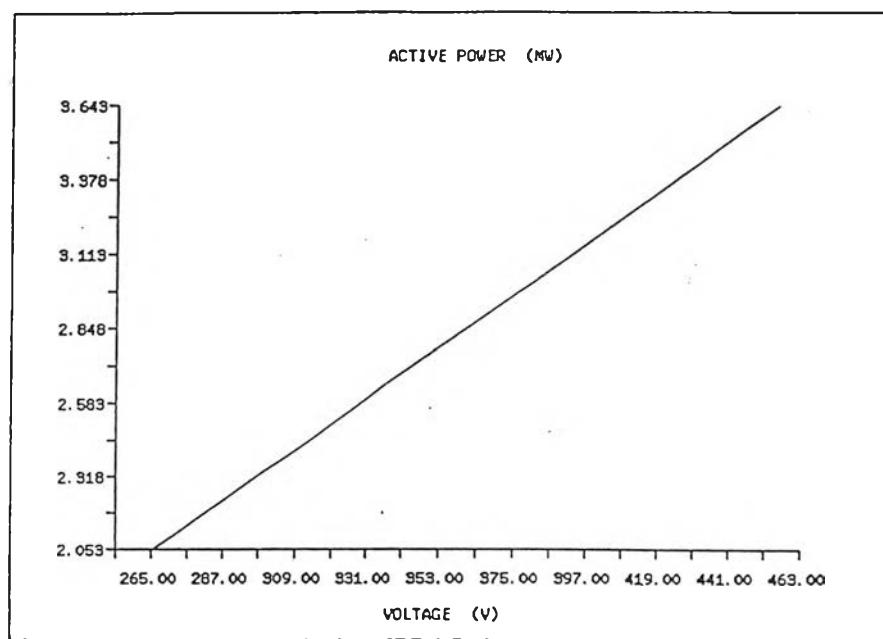


Fig. A4.21 Steady state voltage characteristic of load case 3 used aggregate model

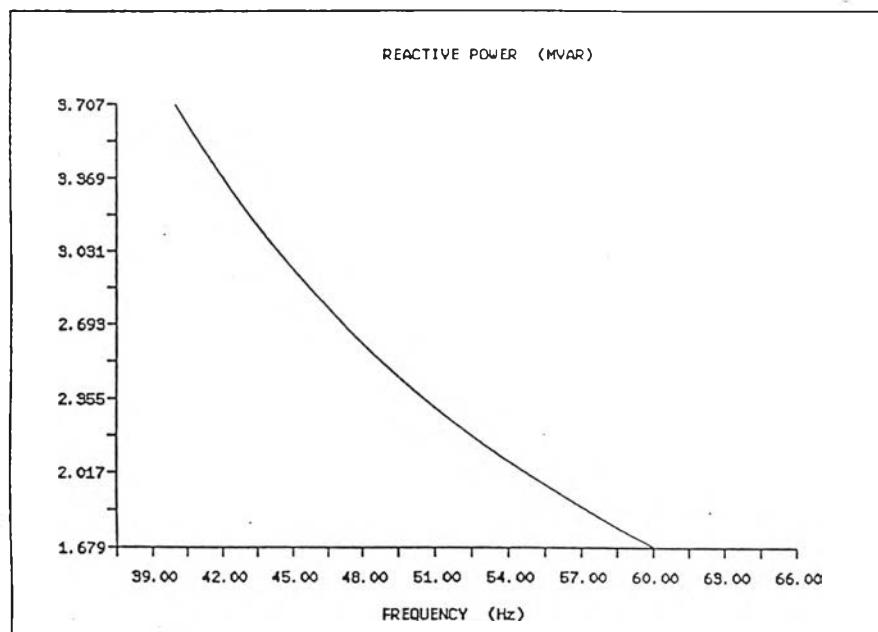
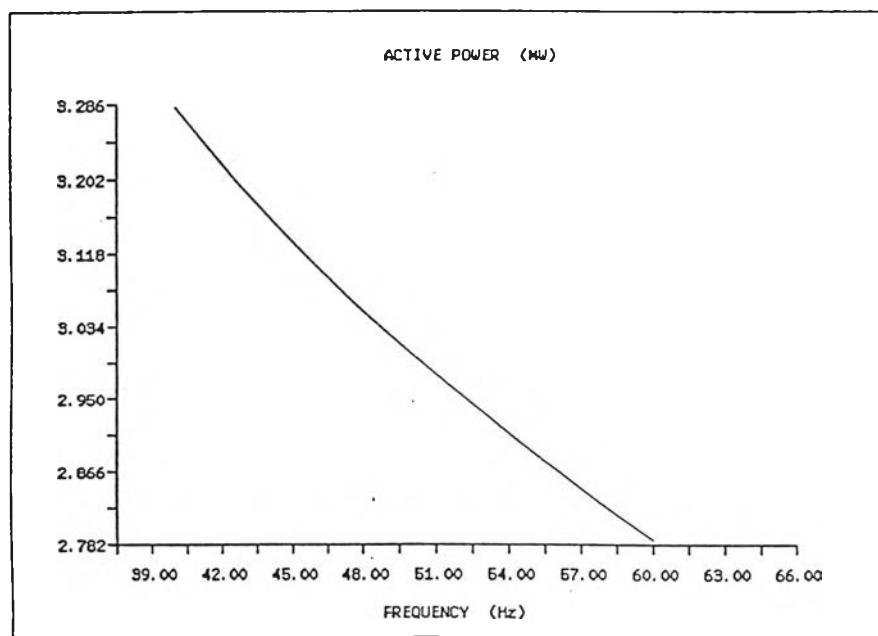


Fig. A4.22 Steady state frequency characteristic of load case 3 used aggregate model

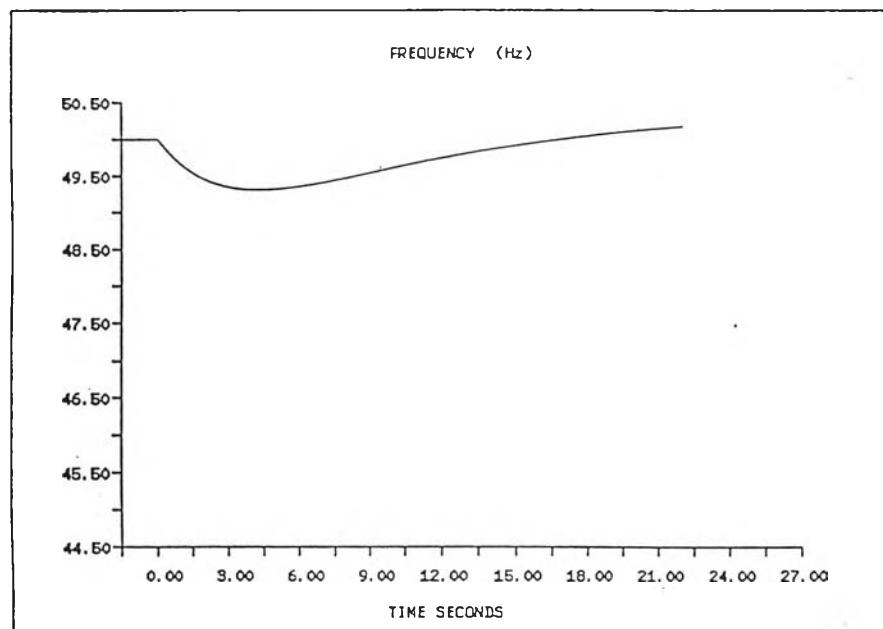
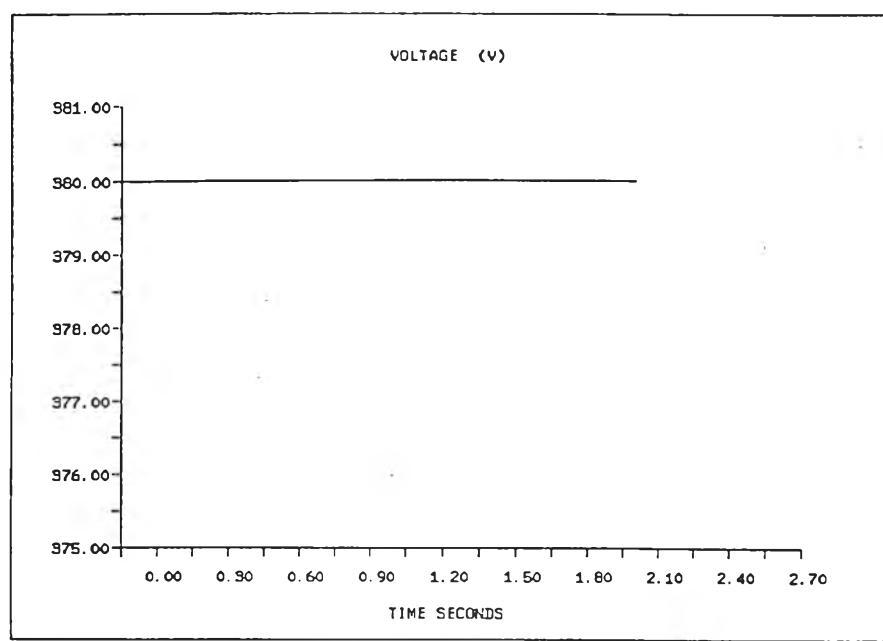


Fig. A4.23 Voltage and frequency at bus 3 during frequency dip

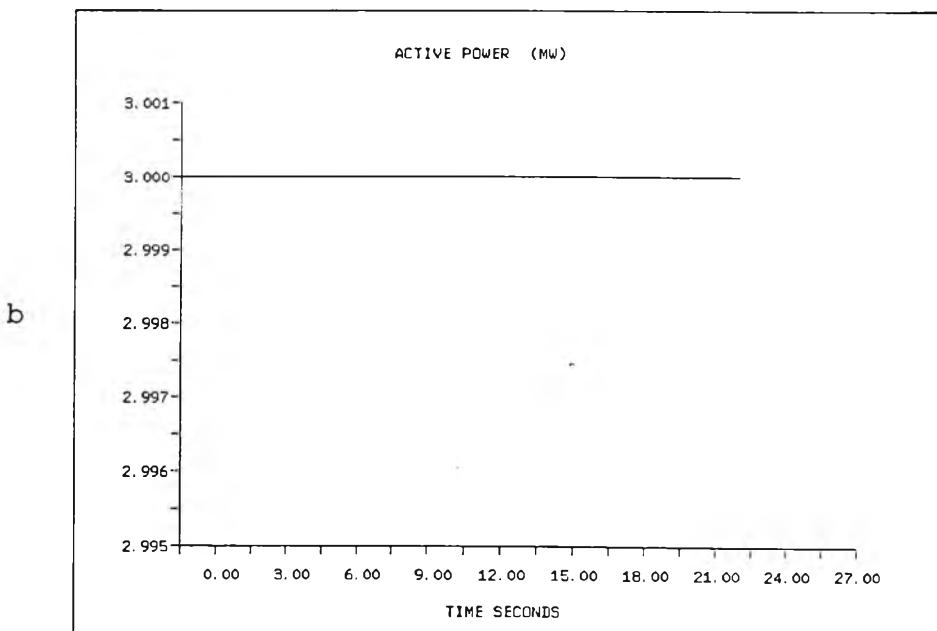
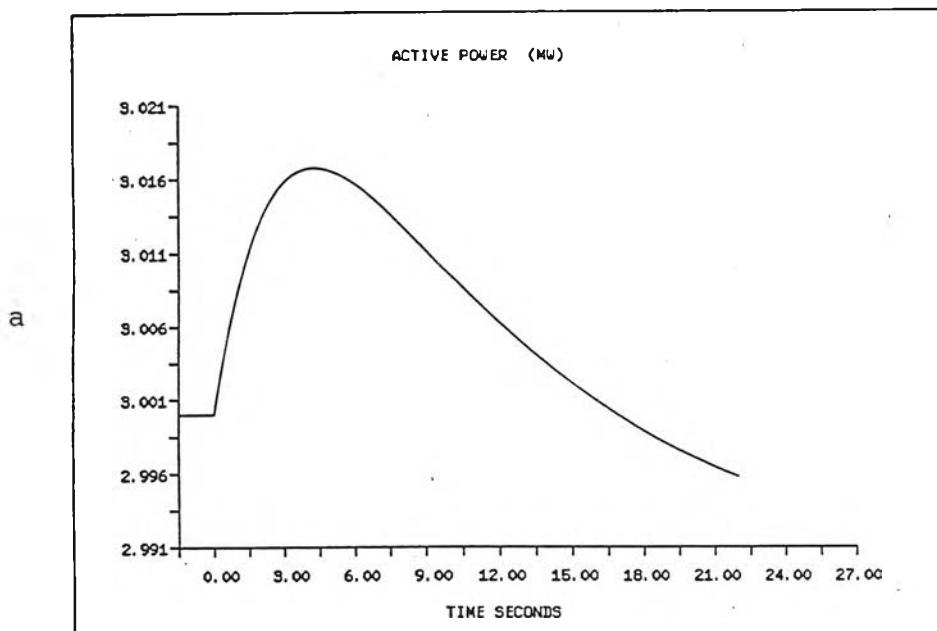


Fig. A4.24 Active power at bus 3 during frequency dip

- a. used aggregate model
- b. used constant MVA model

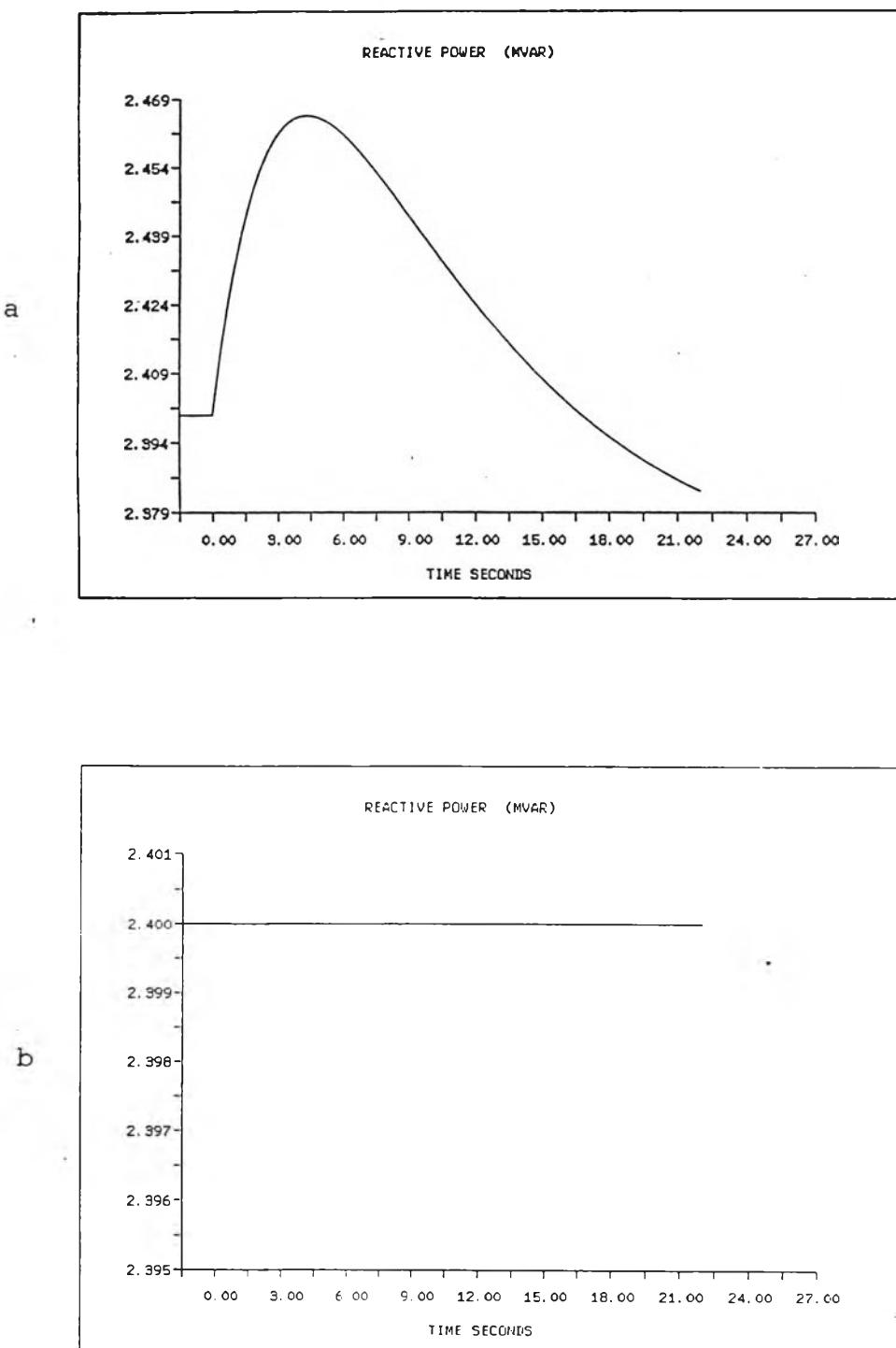


Fig. A4.25 Reactive power at bus 3 during frequency dip

- a. used aggregate model
- b. used constant MVA model

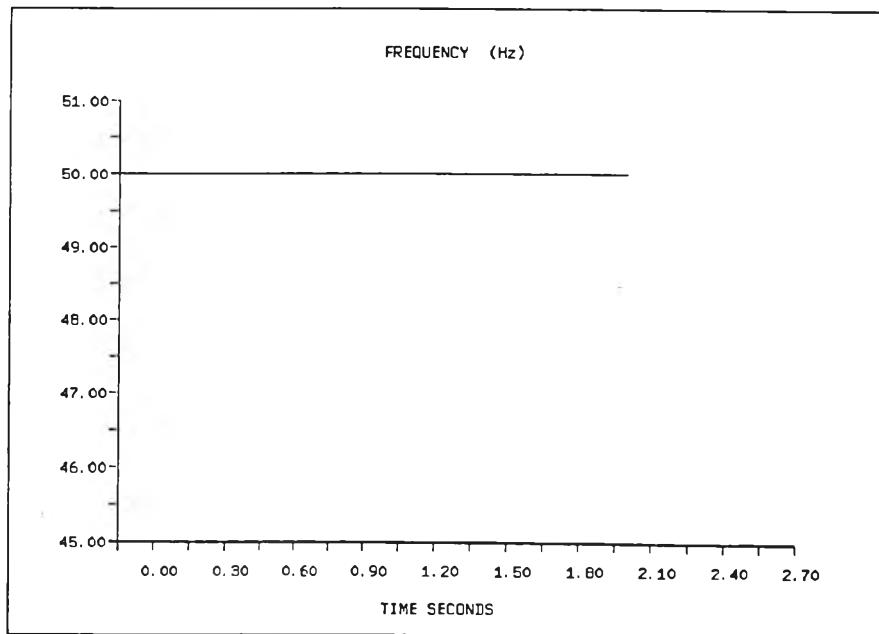
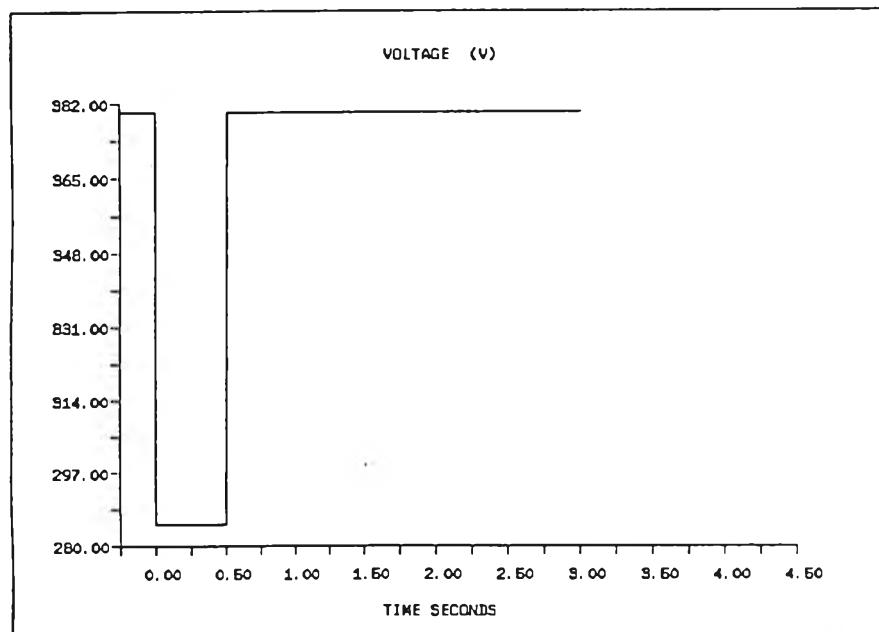


Fig. A4. 26 Voltage and frequency at bus 3 during voltage dip

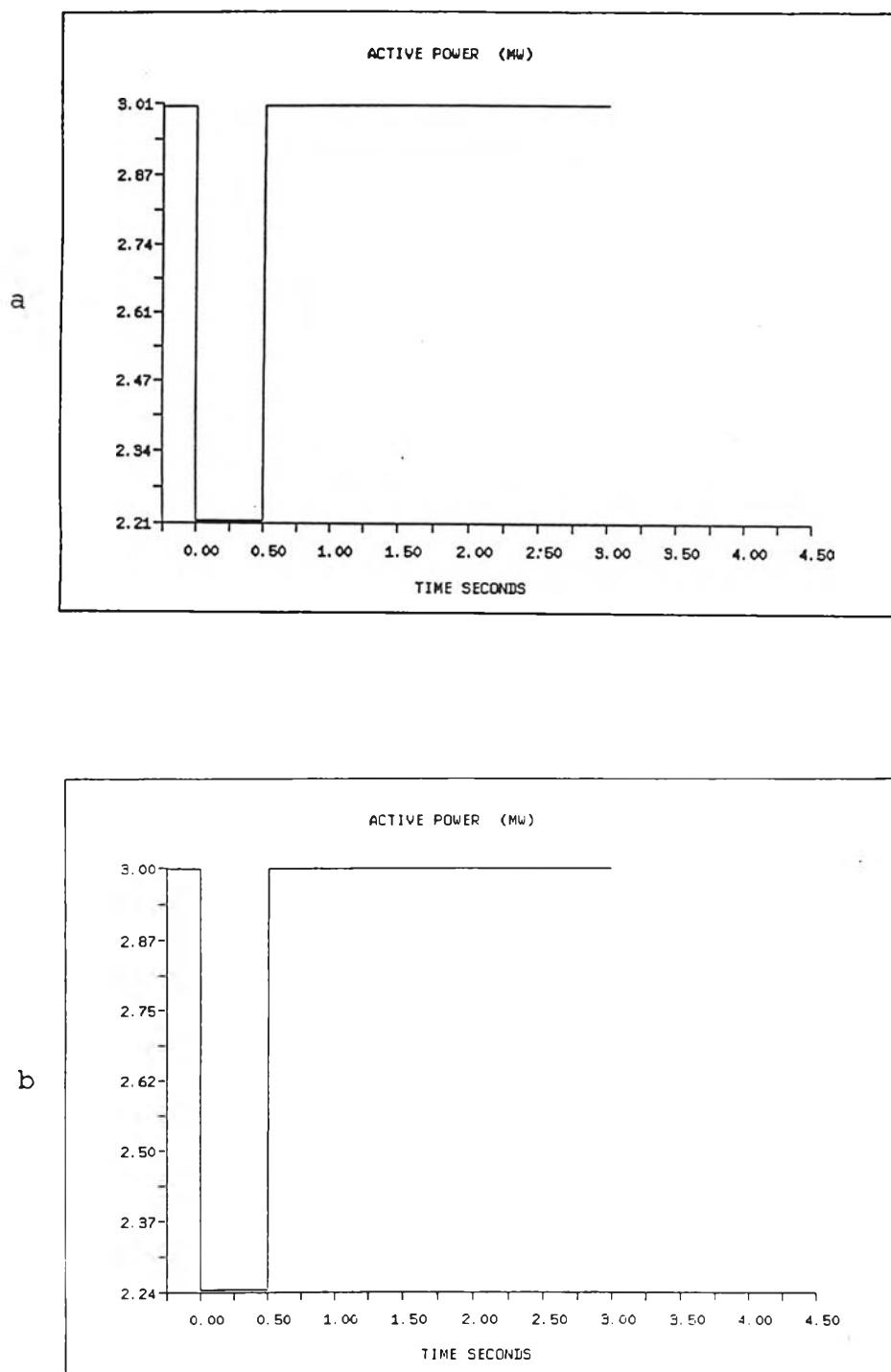


Fig. A4.27 Active power at bus 3 during voltage dip

- used aggregate model
- used constant current model

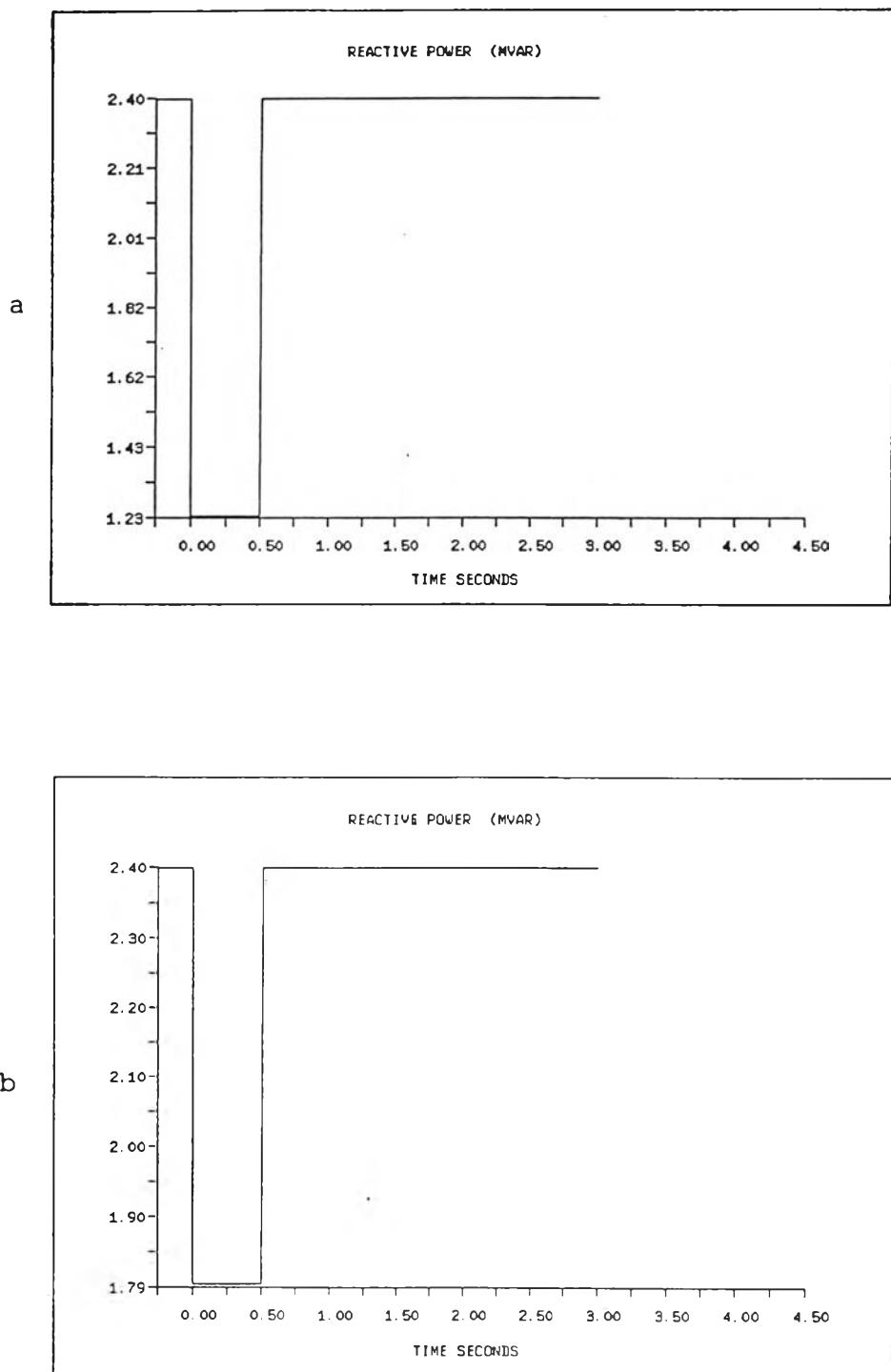


Fig. A4.28 Reactive power at bus 3 during voltage dip
a. used aggregate model
b. used constant current model

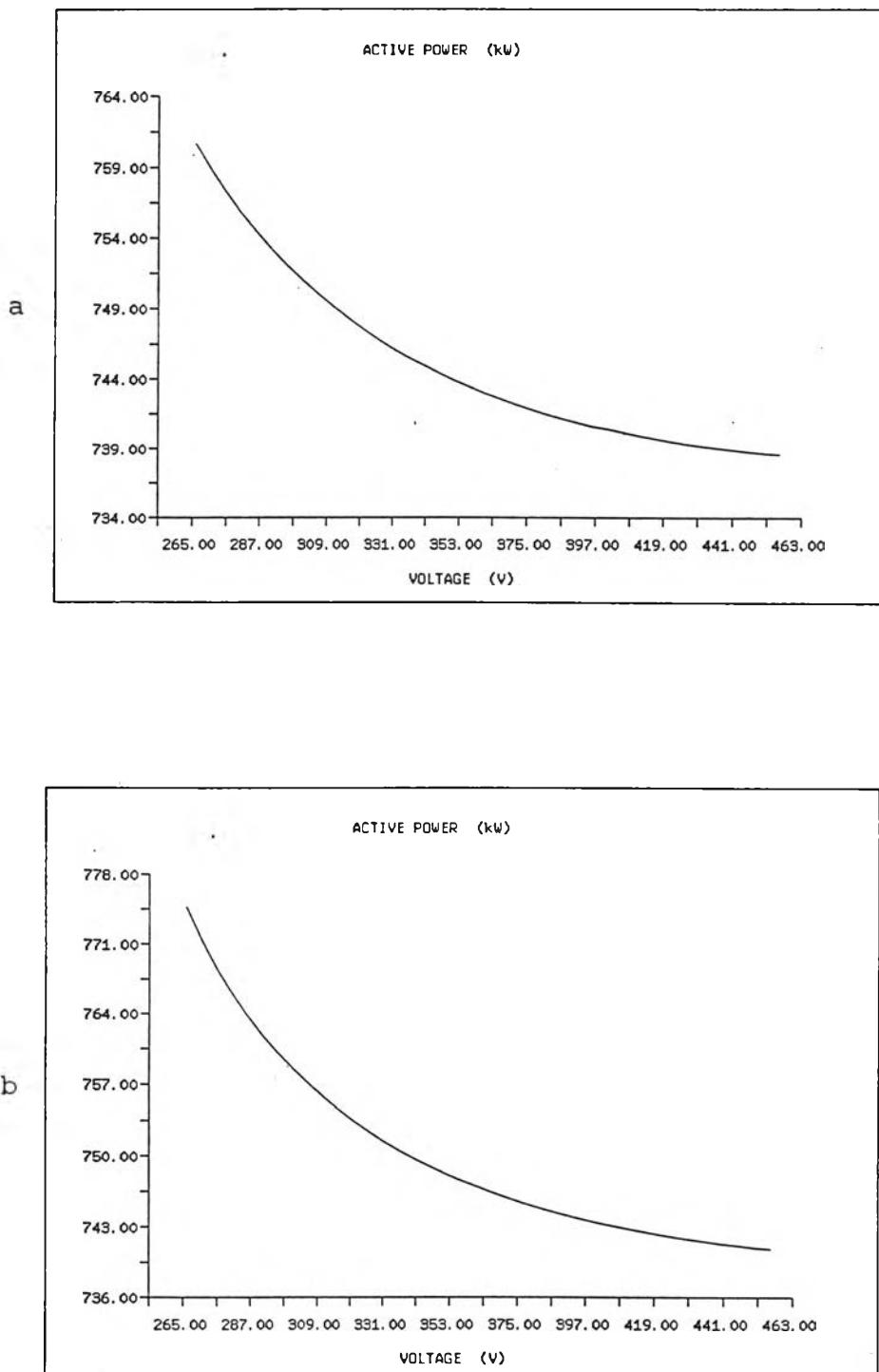


Fig. A4.29 steady state P-V characteristic of
load case 1

- a. used aggregate induction motor model
- b. used individual motor model

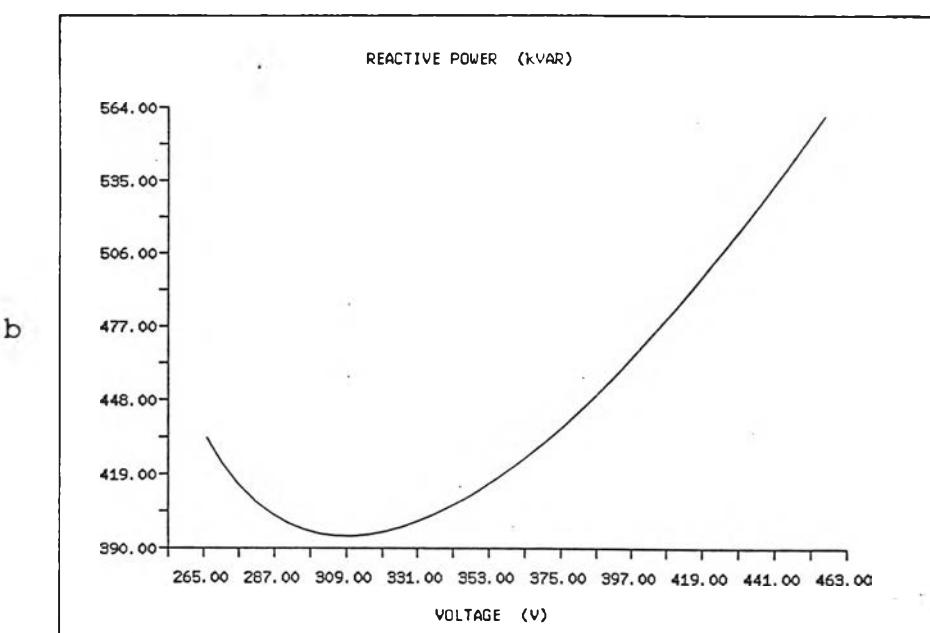
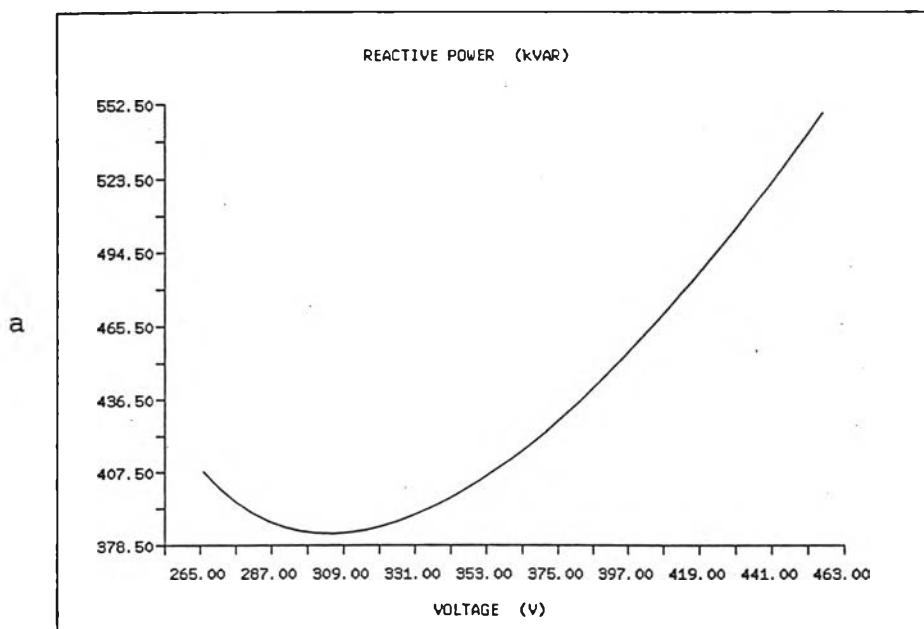


Fig. A4.30 Steady state Q-V characteristic of
load case 1

- a. used aggregate induction motor model
- b. used individual motor model

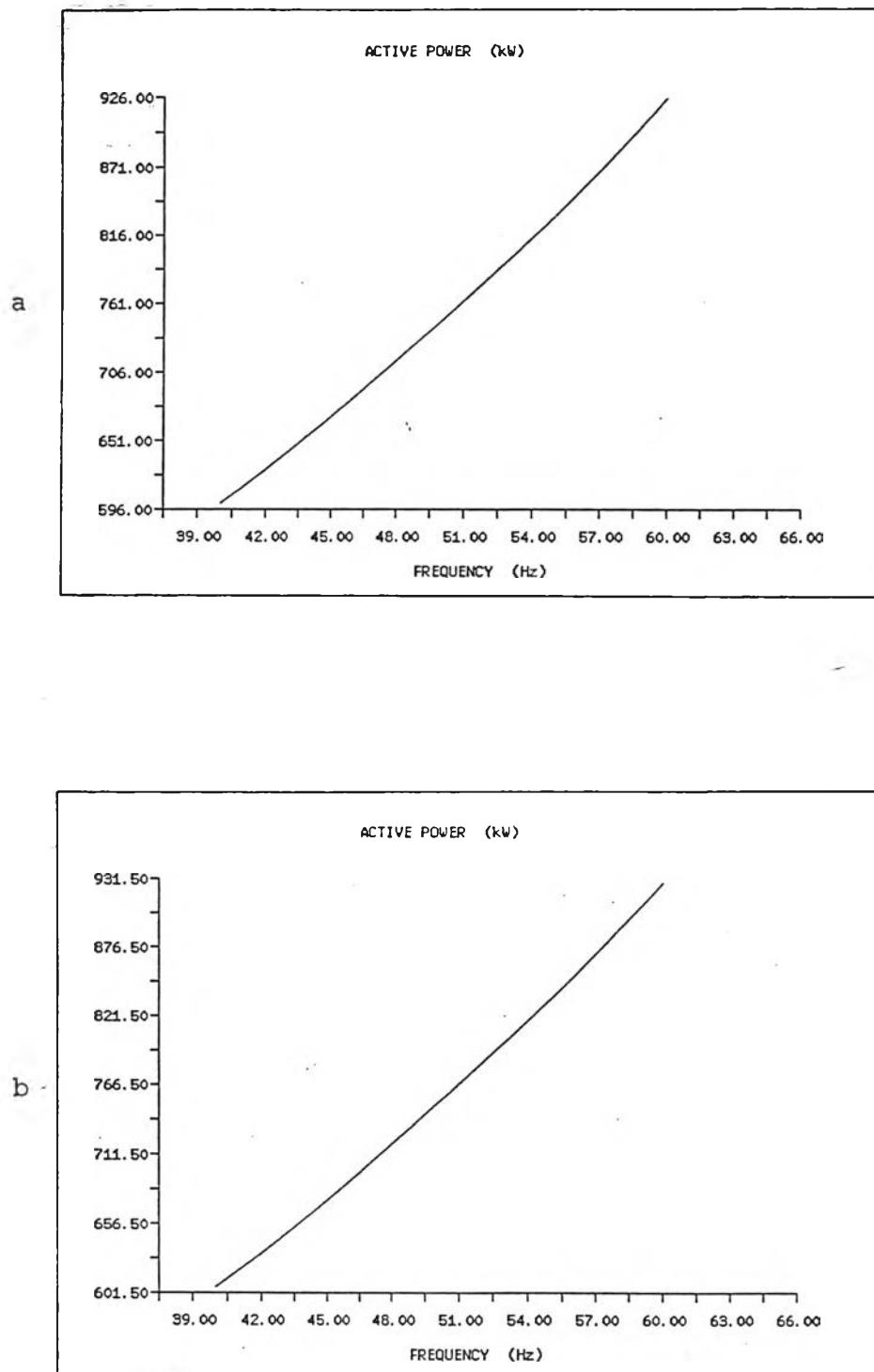


Fig. A4. 31 Steady state P-f characteristic of
load case 1

- used aggregate induction motor model
- used individual motor model

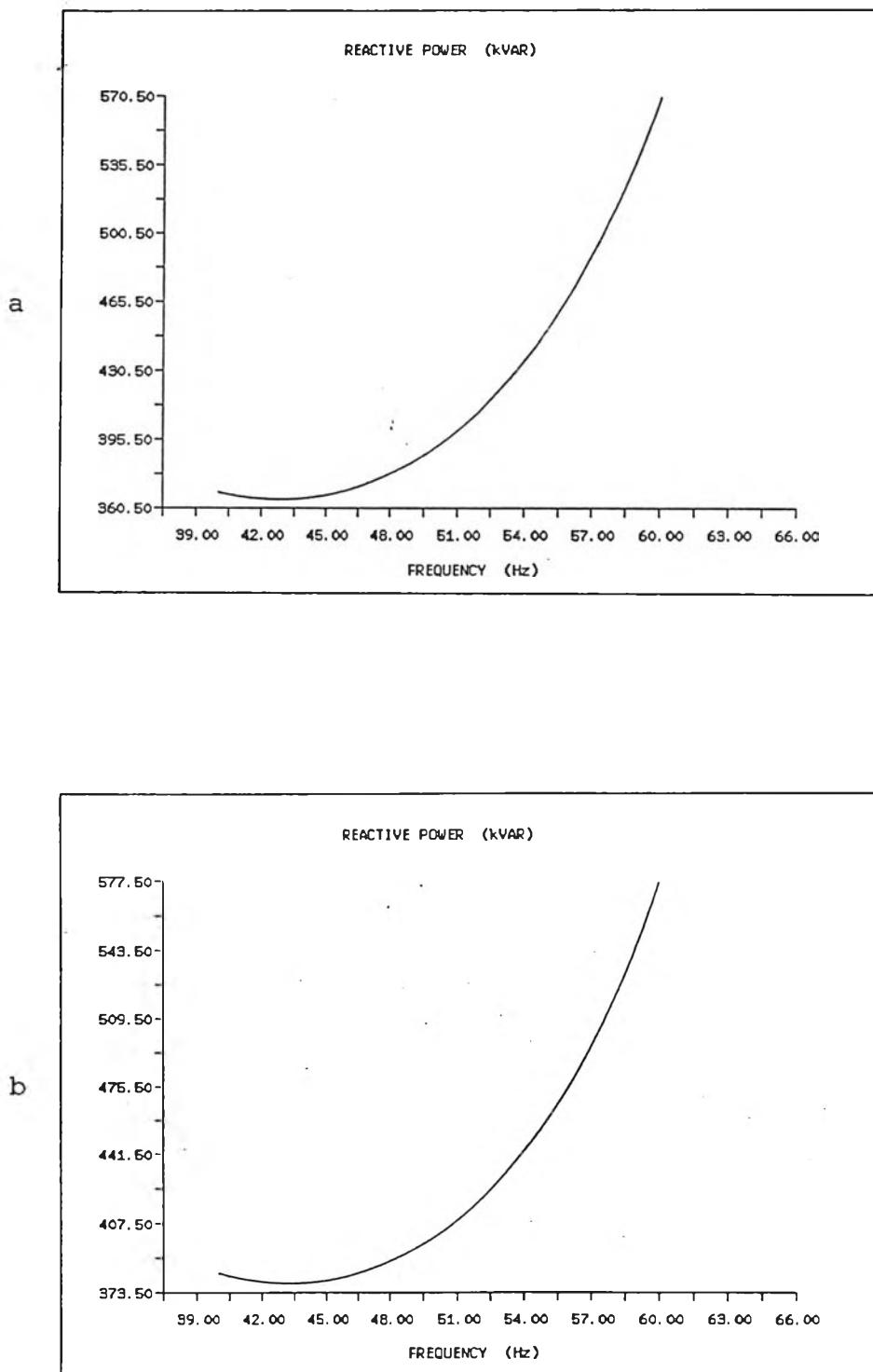


Fig. A4.32 Steady state Q-f characteristic of load case 1

- a. used aggregate induction motor model
- b. used individual motor model

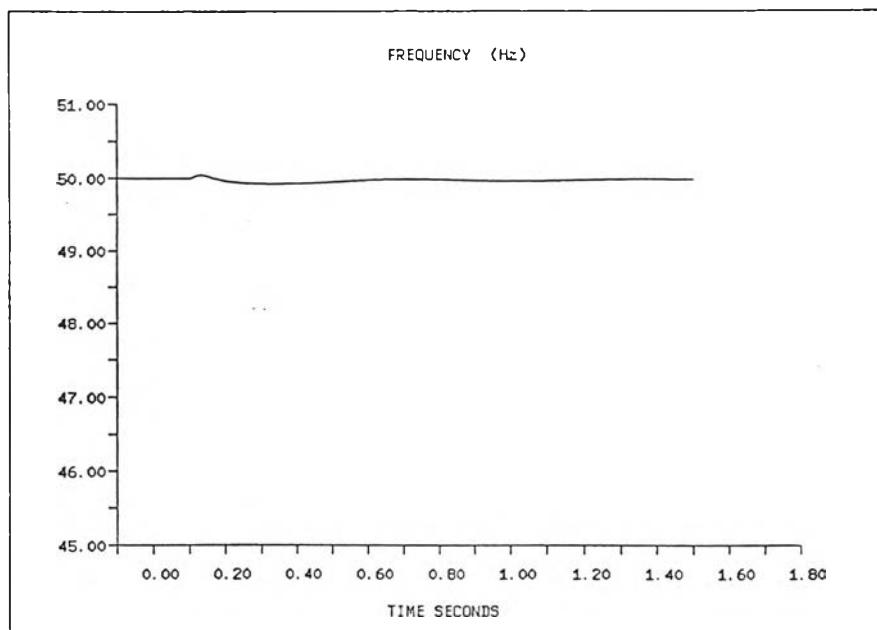
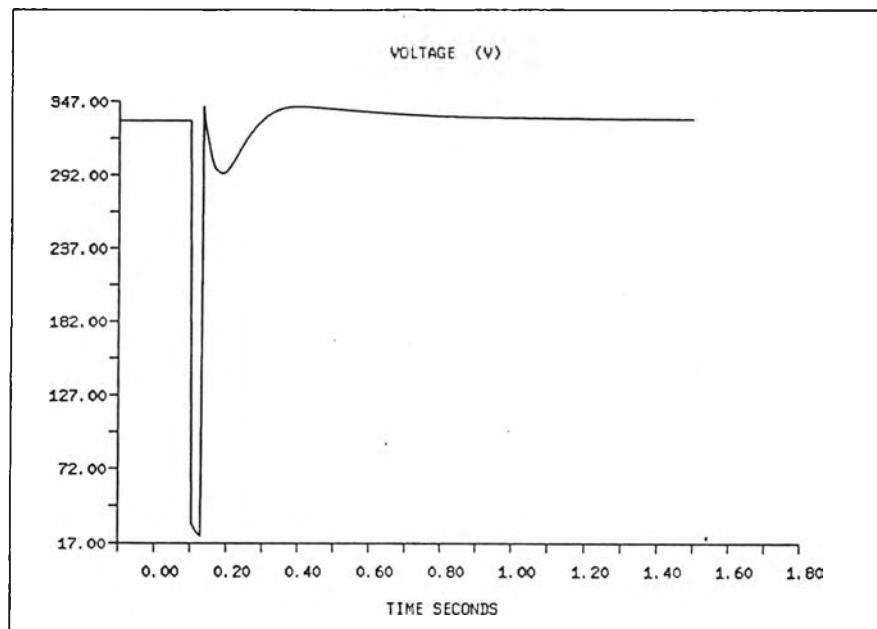


Fig. A4.33 Voltage and frequency at bus 4 during
3 phase fault at bus 3

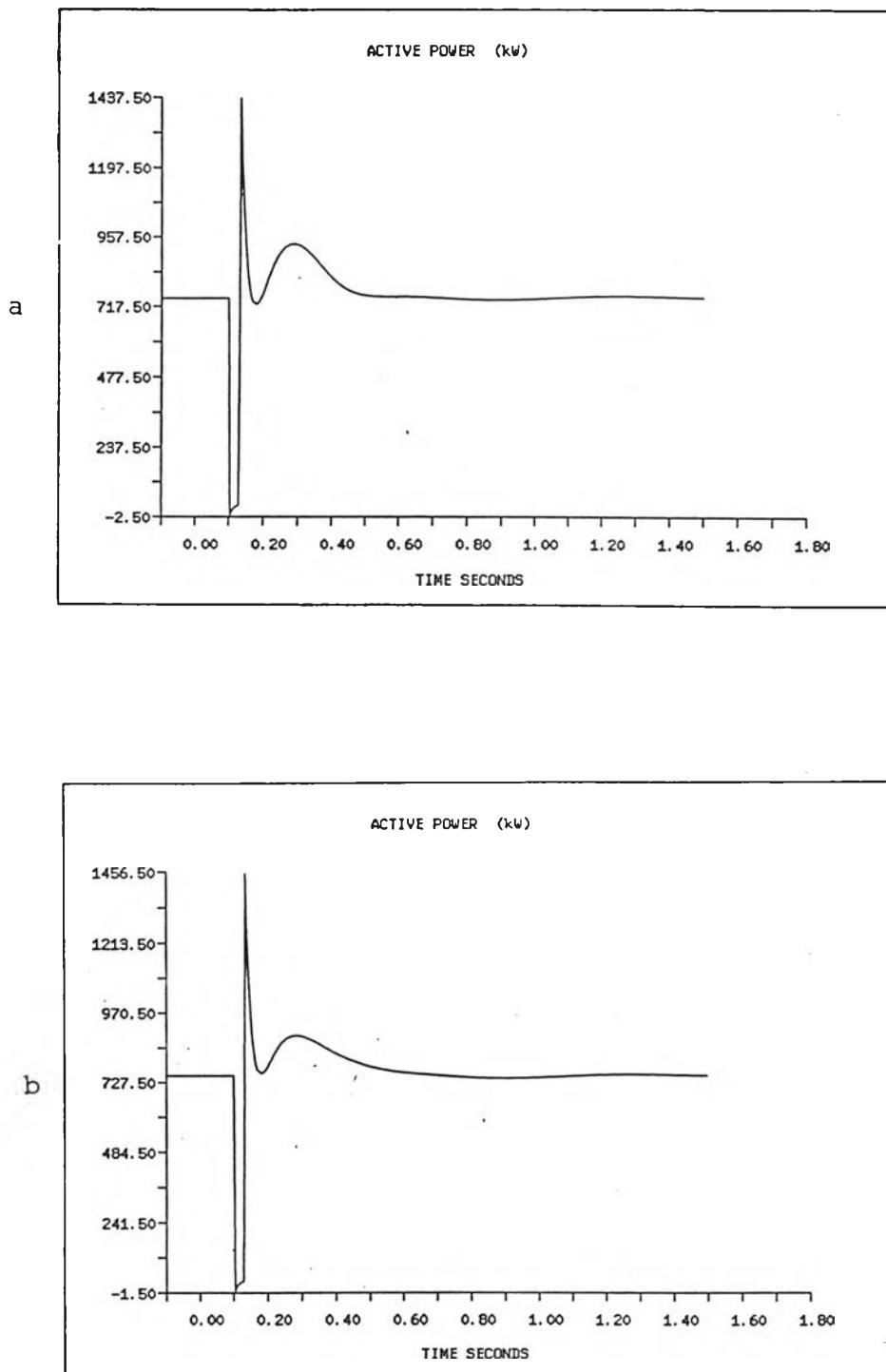


Fig. A4.34 Active power at bus 4 during 3 phase fault

a. used aggregate induction motor model

b. used individual motor model

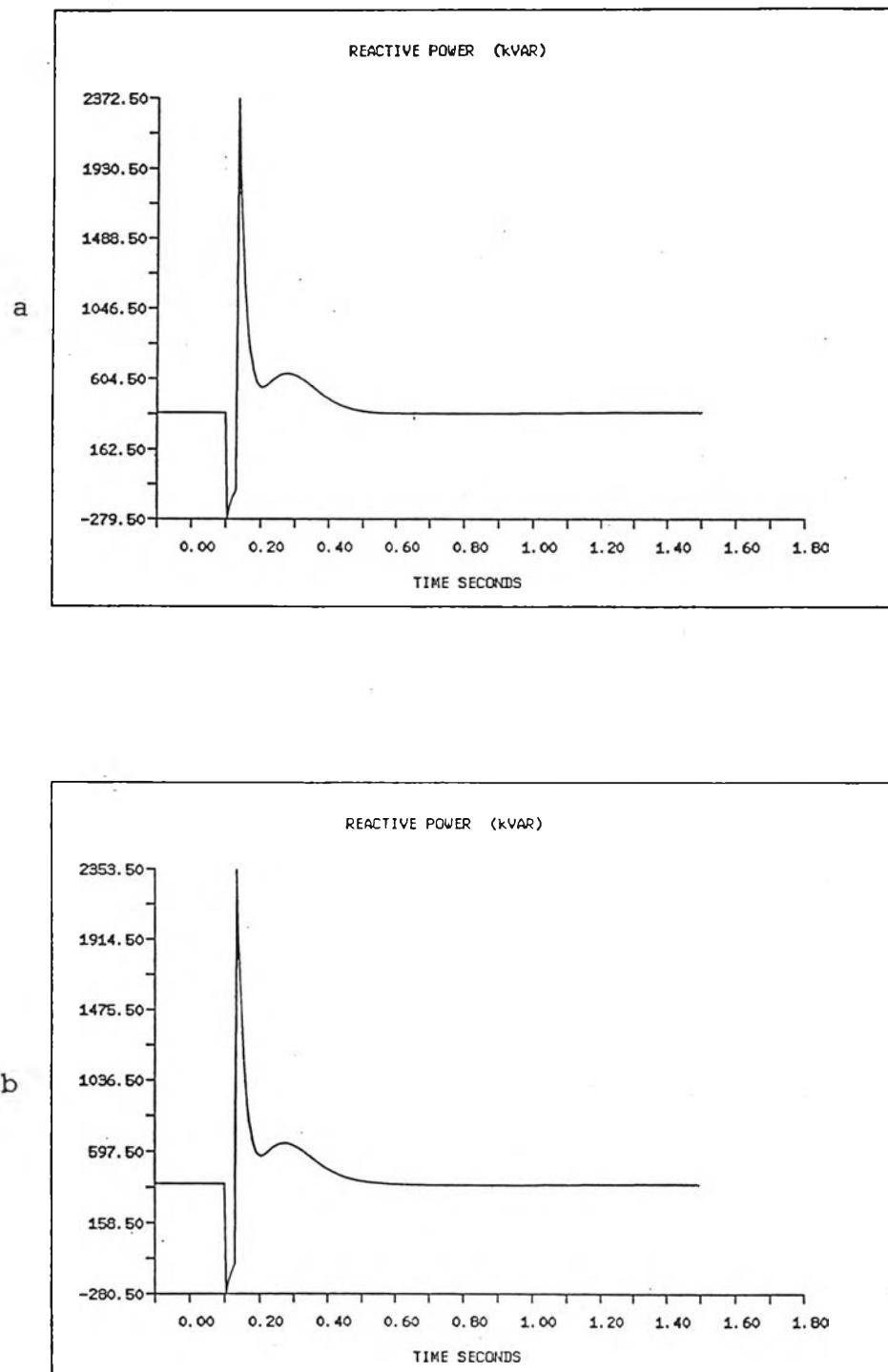


Fig. A4.35 Reactive power at bus 4 during 3 phase fault

- a. used aggregate induction motor model
- b. used individual motor model

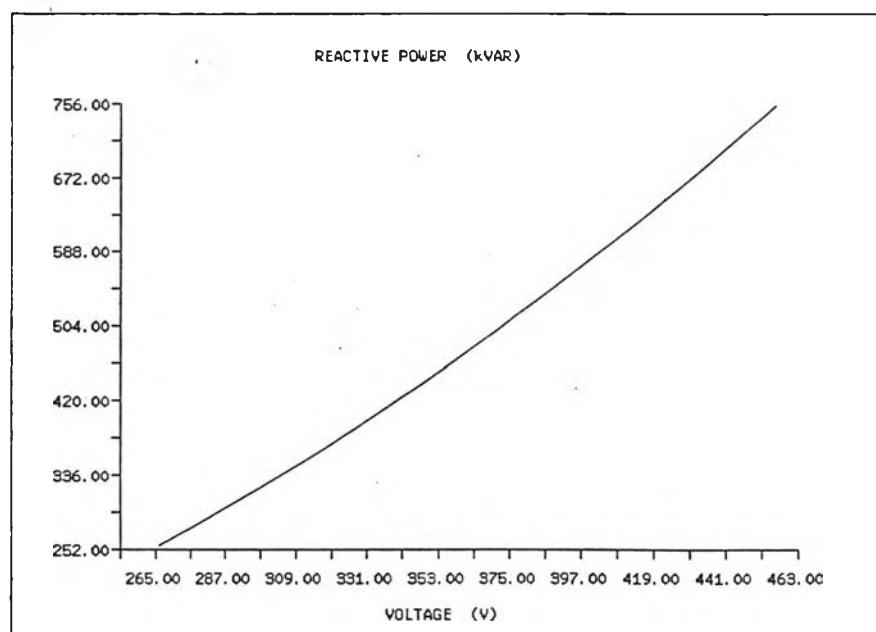
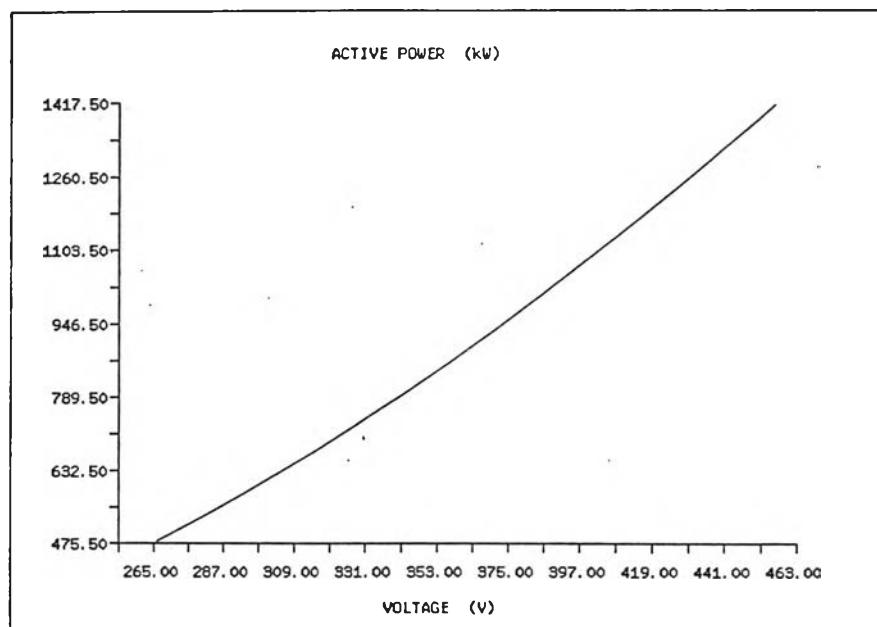


Fig. A4.36 Steady state voltage characteristic of load case 1 used constant impedance model

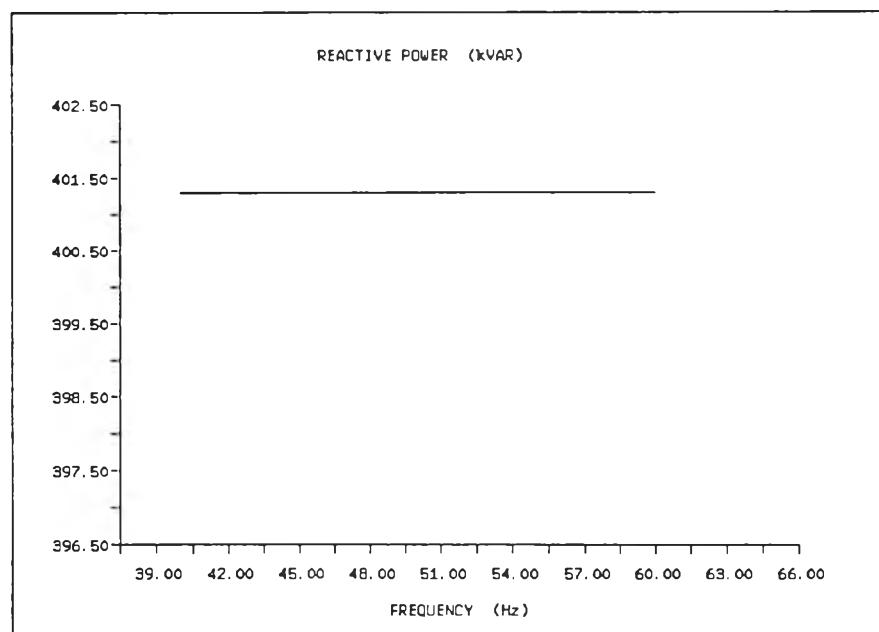
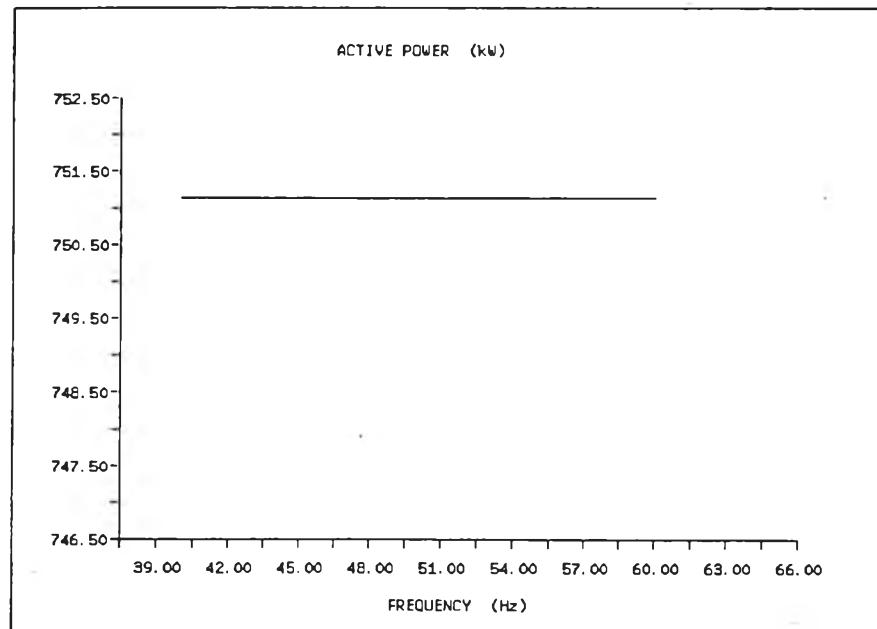


Fig. A4.37 Steady state frequency characteristic of load case 1 used constant impedance model

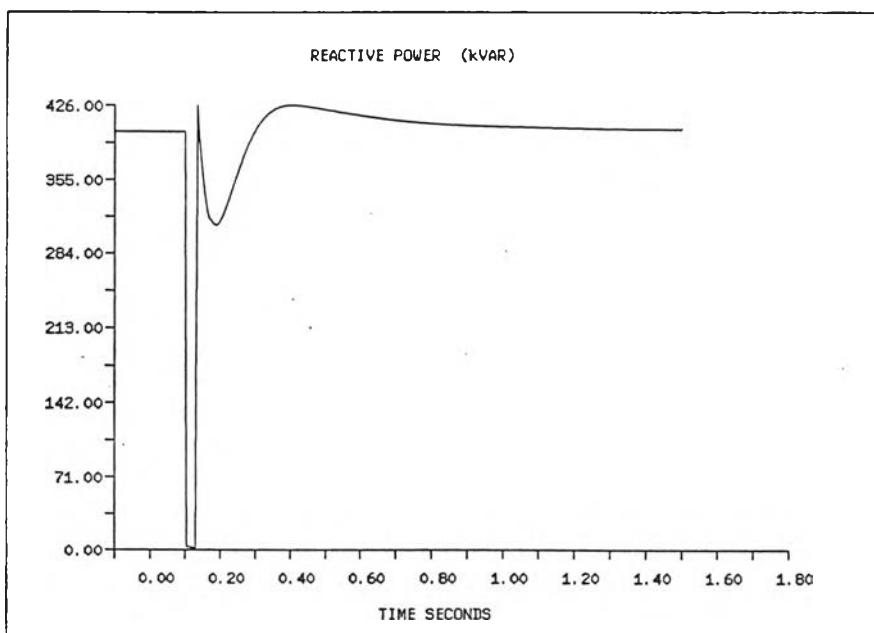
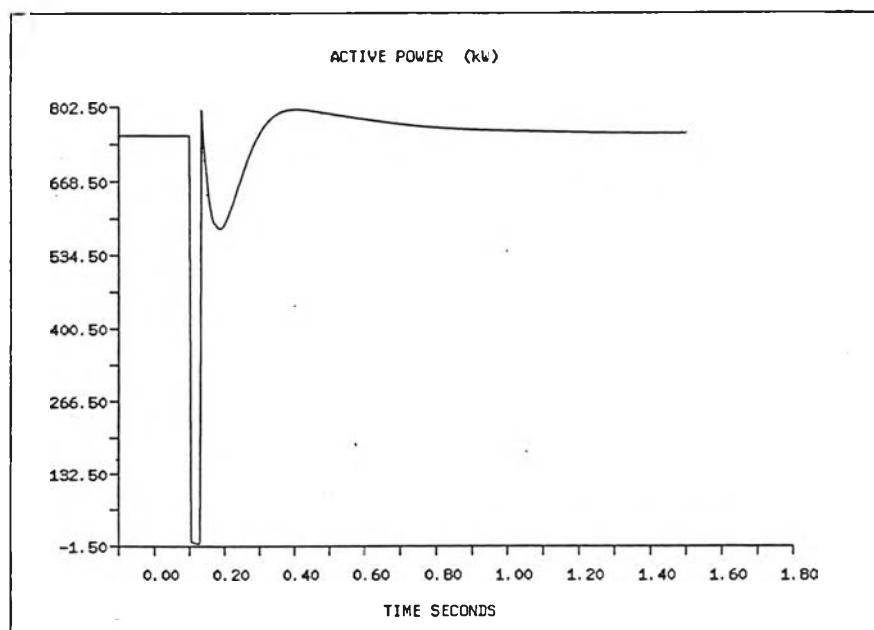


Fig. A4.38 Active and reactive power at bus 4 during
3 phase fault used constant impedance model

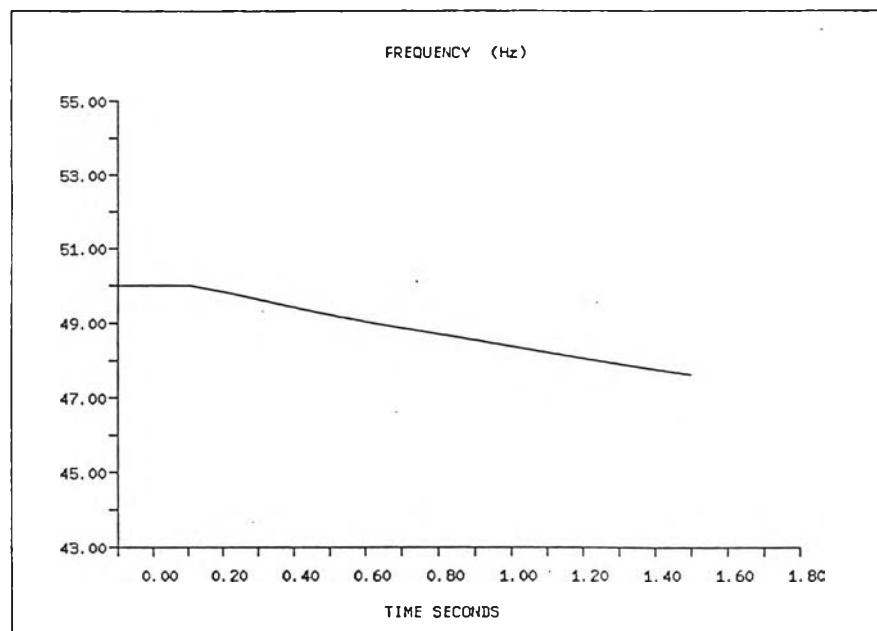
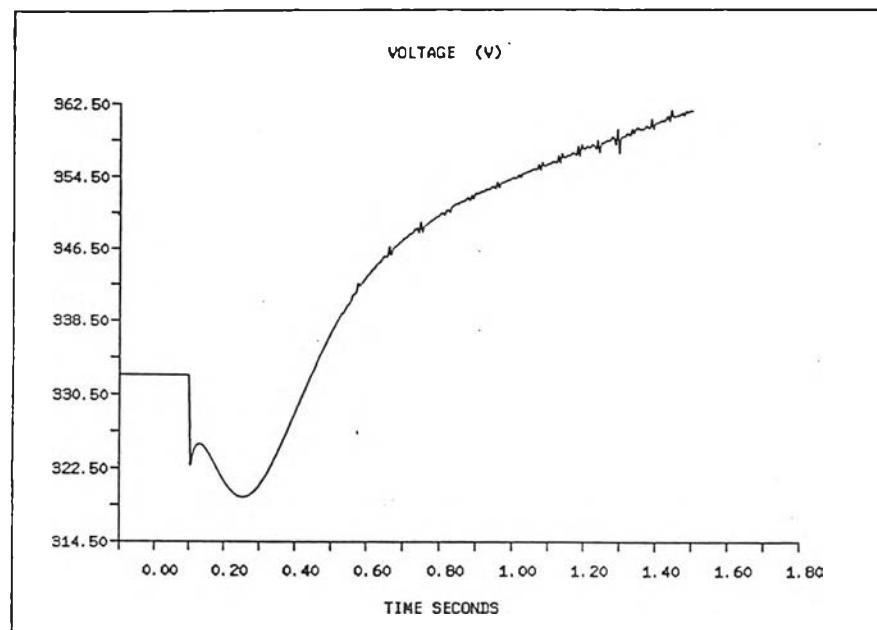


Fig. A4.39 Voltage and frequency at bus 4 during gen. 2 disconnected

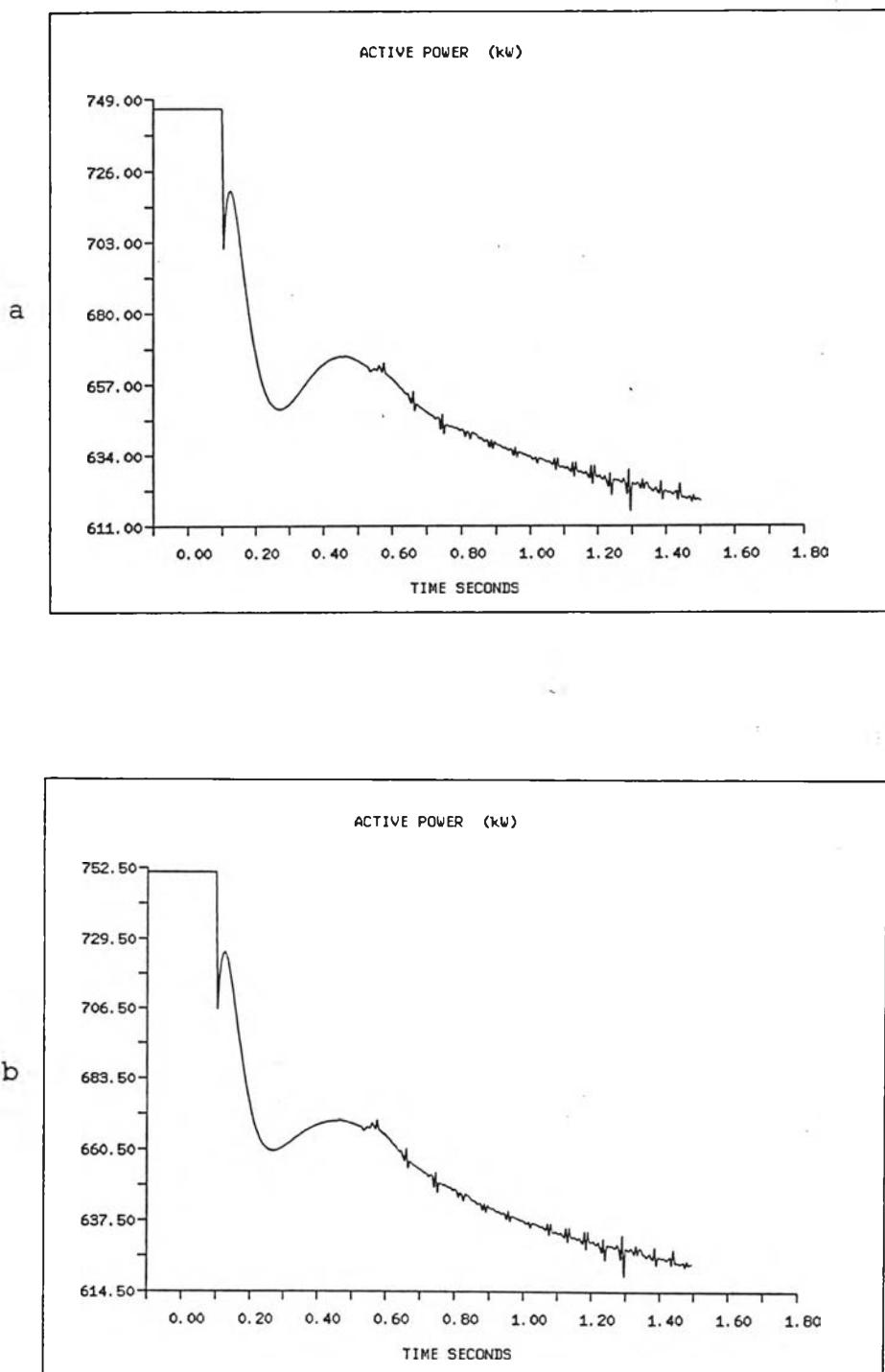


Fig. A4.40 Active power at bus 4 during gen. 2 disconnected

- a. used aggregate induction motor model
- b. used individual motor model

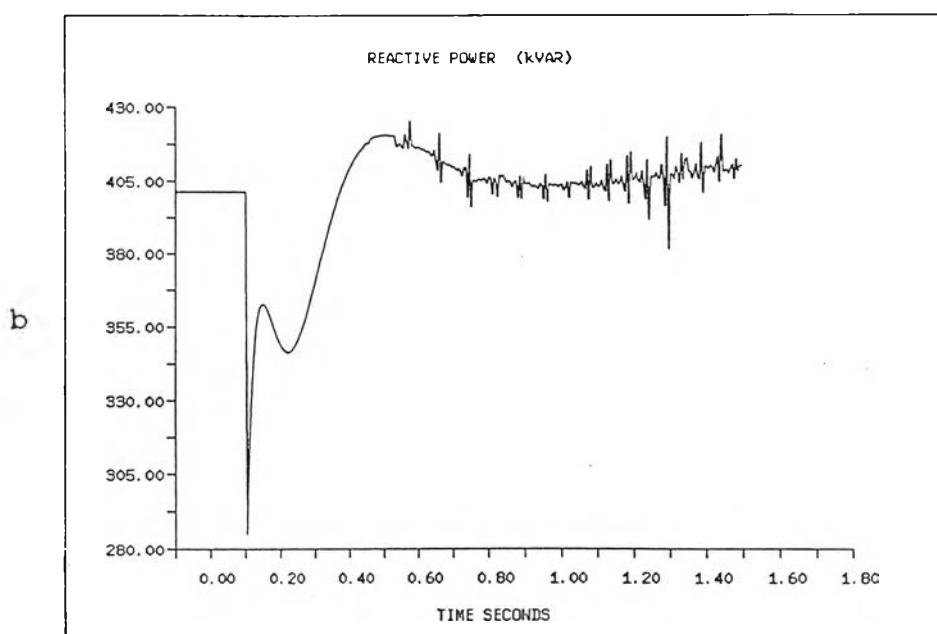
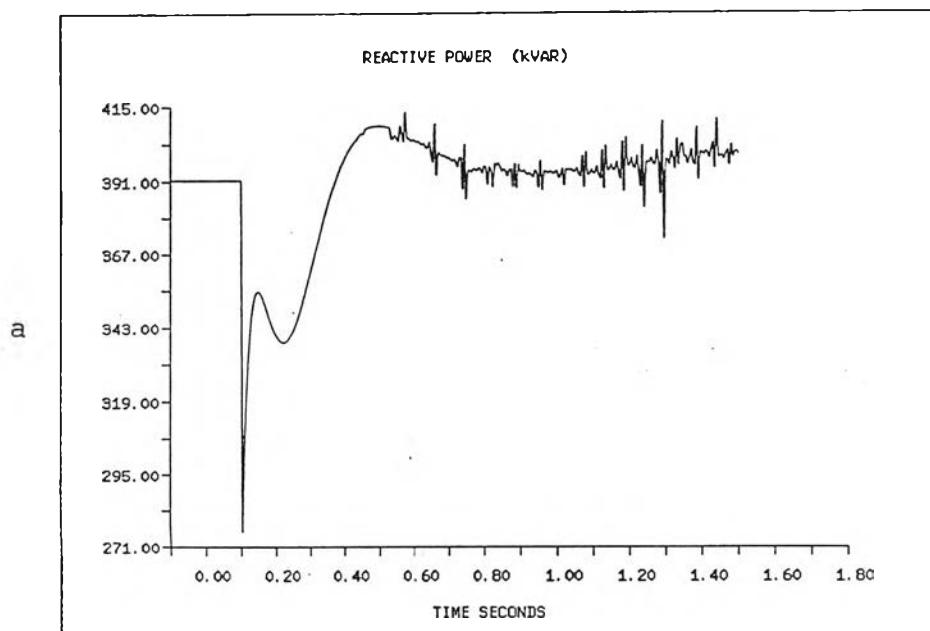


Fig. A4.41 Reactive power at bus 4 during gen. 2 disconnected

- a. used aggregate induction motor model
- b. used individual motor model

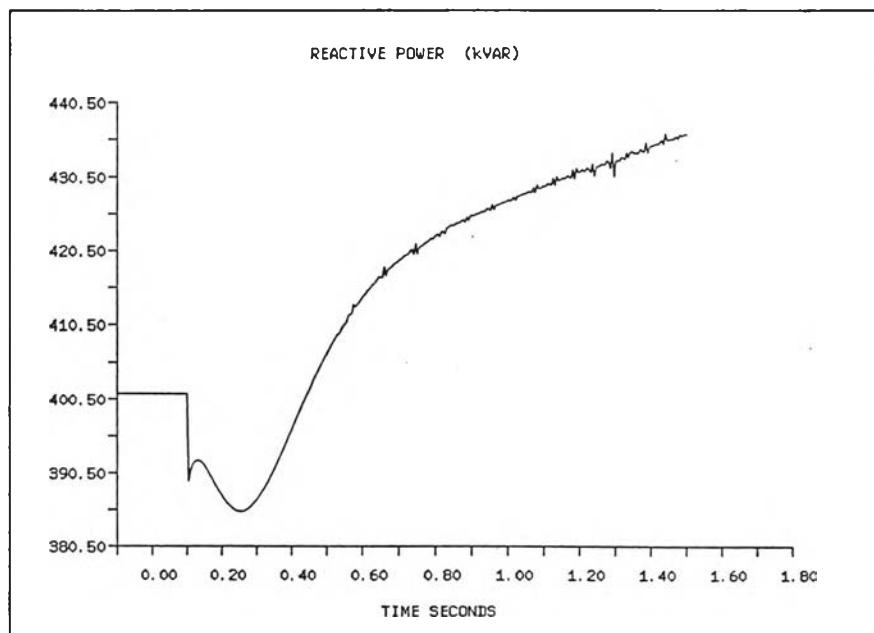
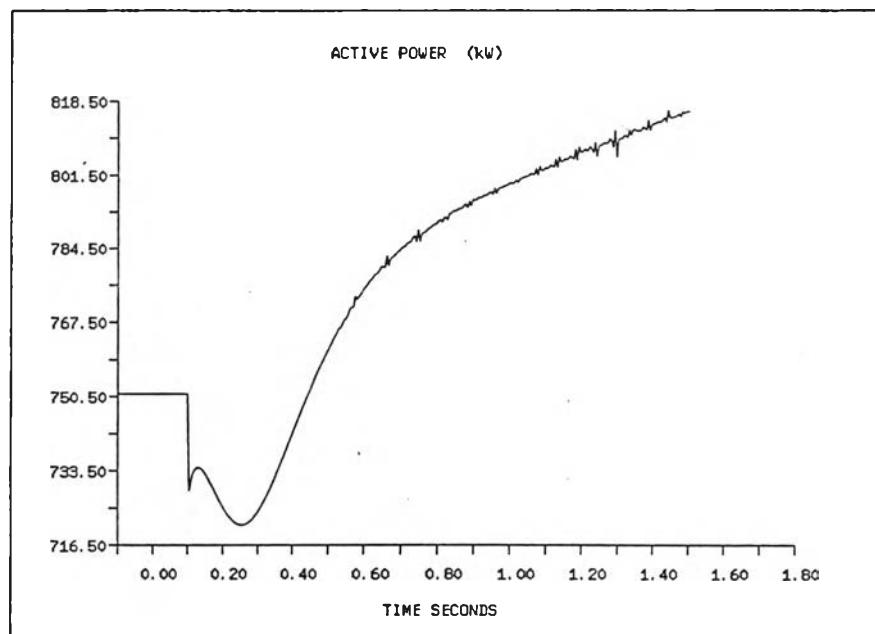


Fig. A4.42 Active and reactive power during gen. 2 disconnected used constant current model

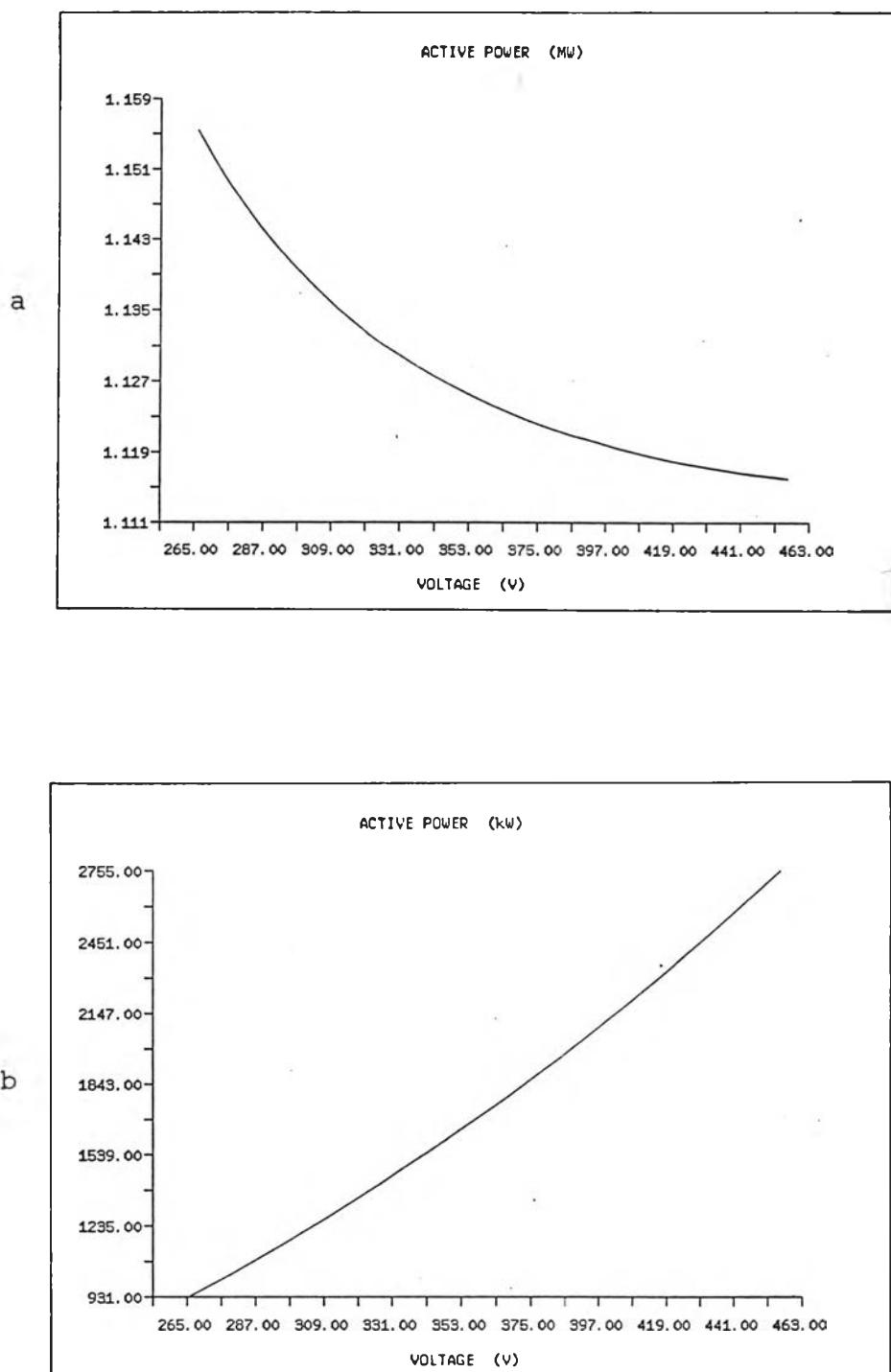


Fig.A4.43 Steady state P-V characteristic of load
case 2

- a. used aggregate induction motor model
- b. used constant impedance model

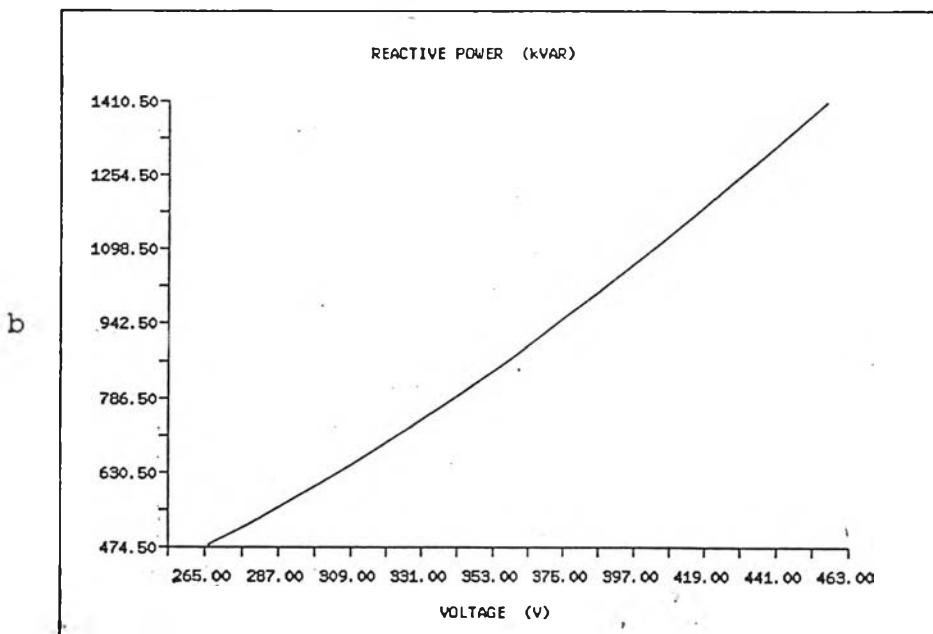
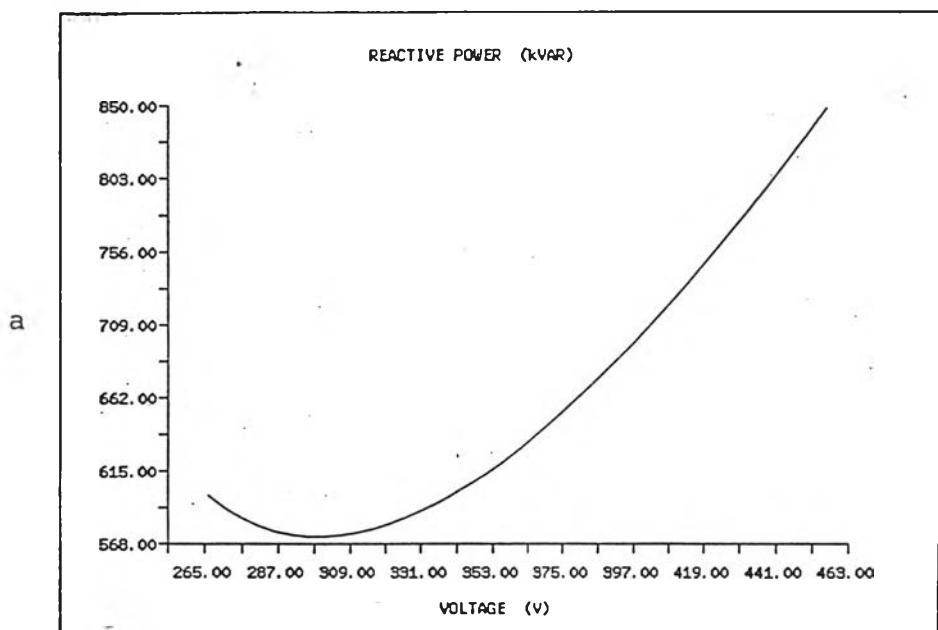


Fig. A4.44 Steady state Q-V characteristic of load
case 2

- a. used aggregate induction motor model
- b. used constant impedance model

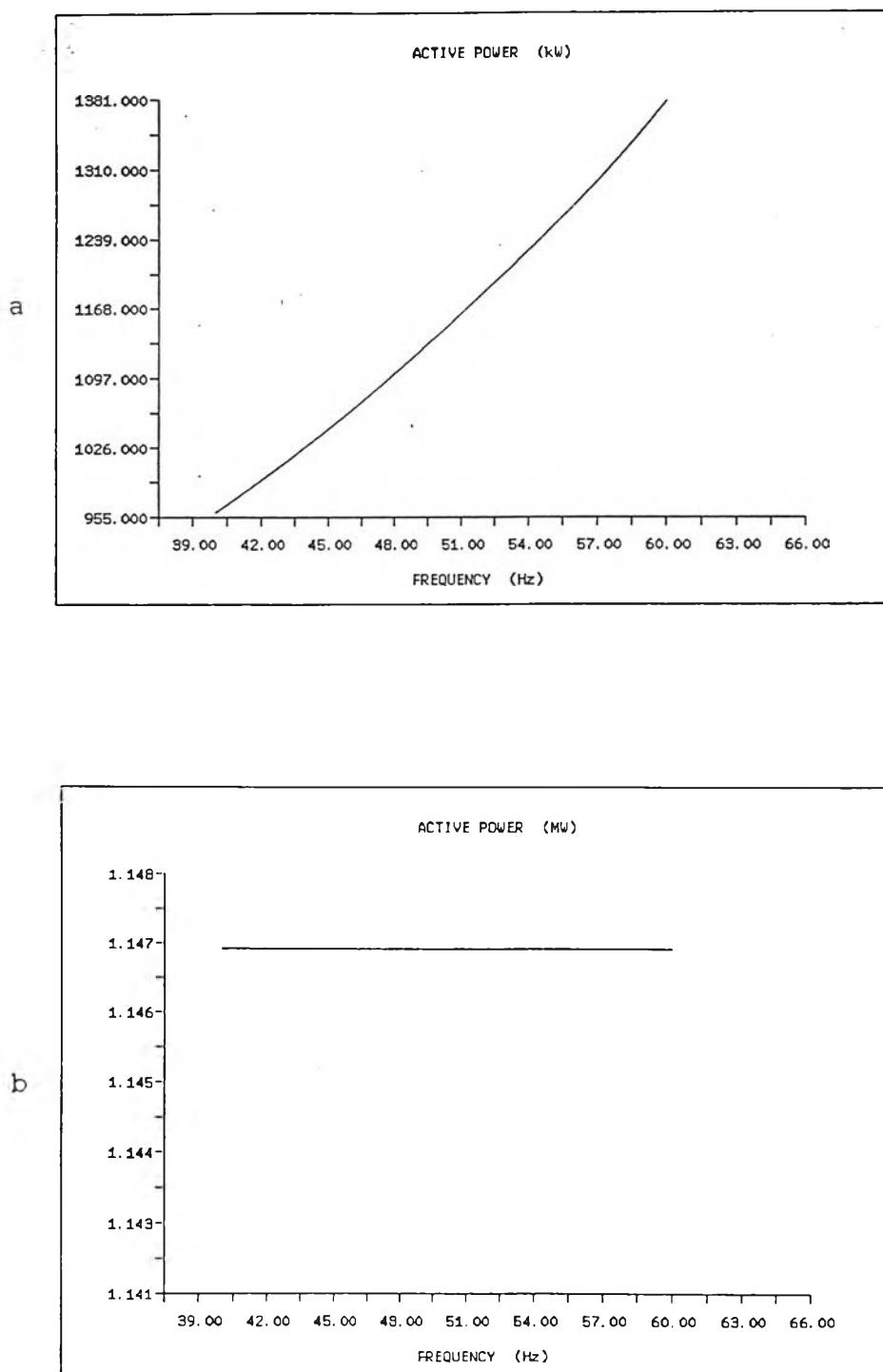


Fig. A4.45 Steady state P-f characteristic of load
case 2

- a. used aggregate induction motor model
- b. used constant impedance model

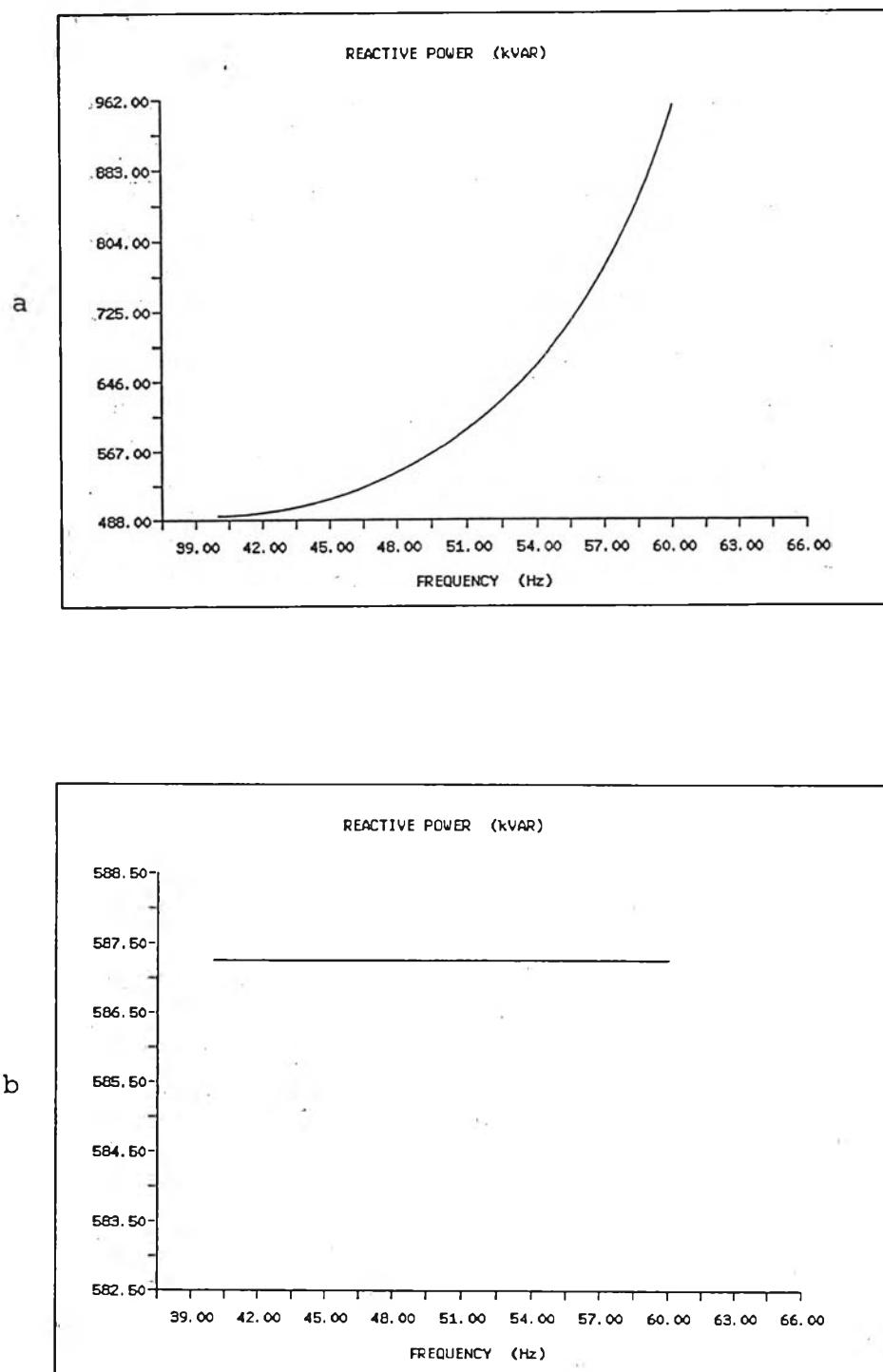


Fig. A4.46 Steady state Q-V characteristic of load

case 2

- a. used aggregate induction motor model
- b. used constant impedance model

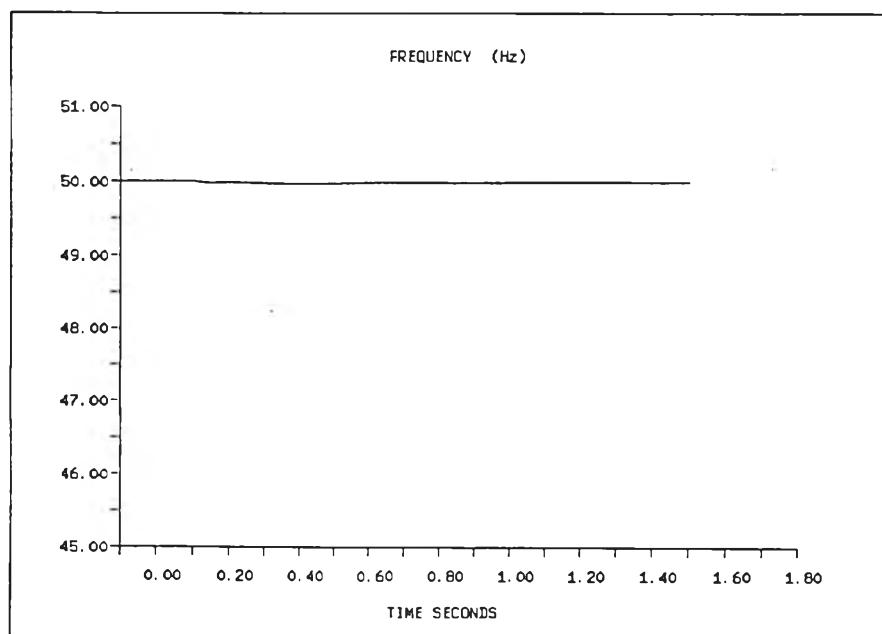
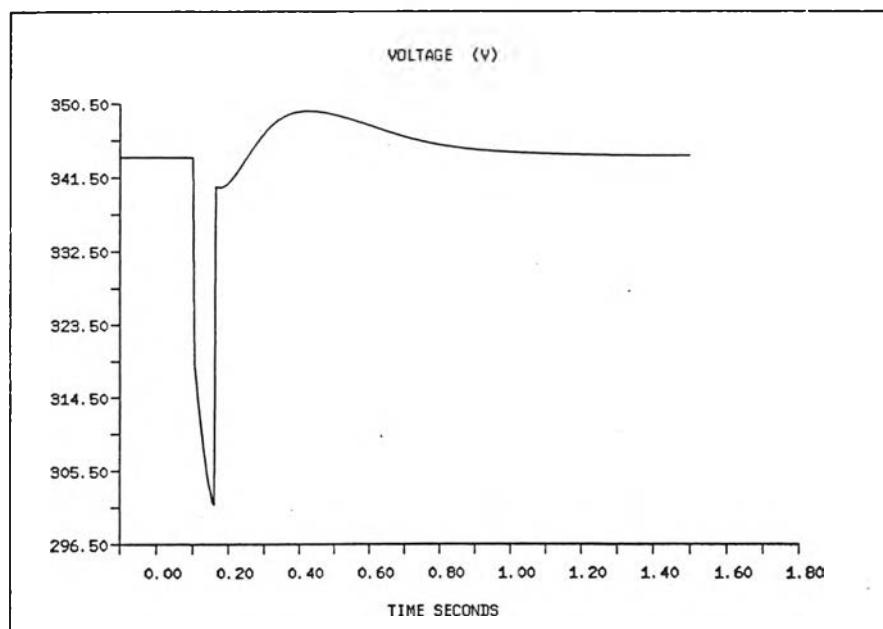


Fig. A4.47 Voltage and frequency at bus 4 when gen. 1 disconnected for 0.05 s

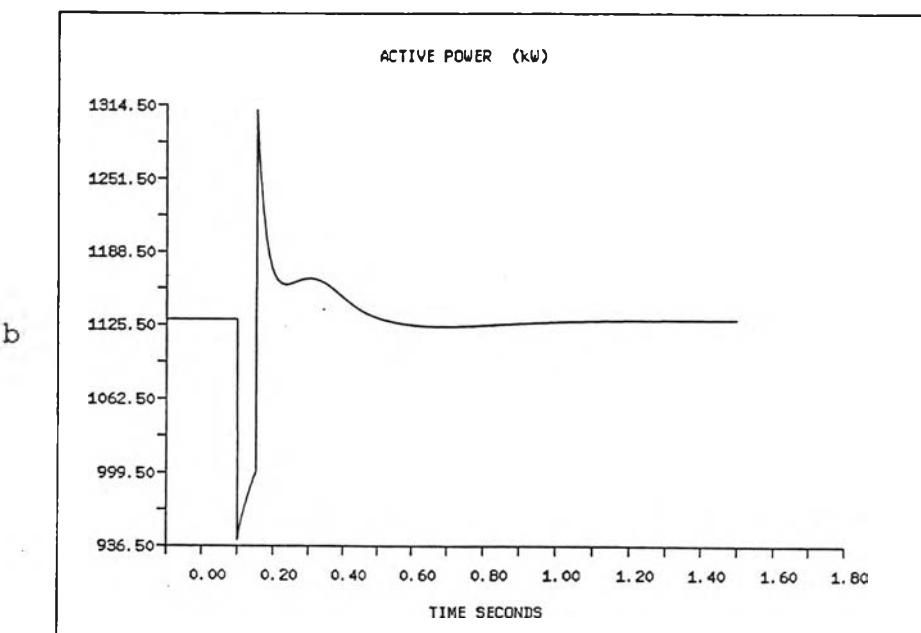
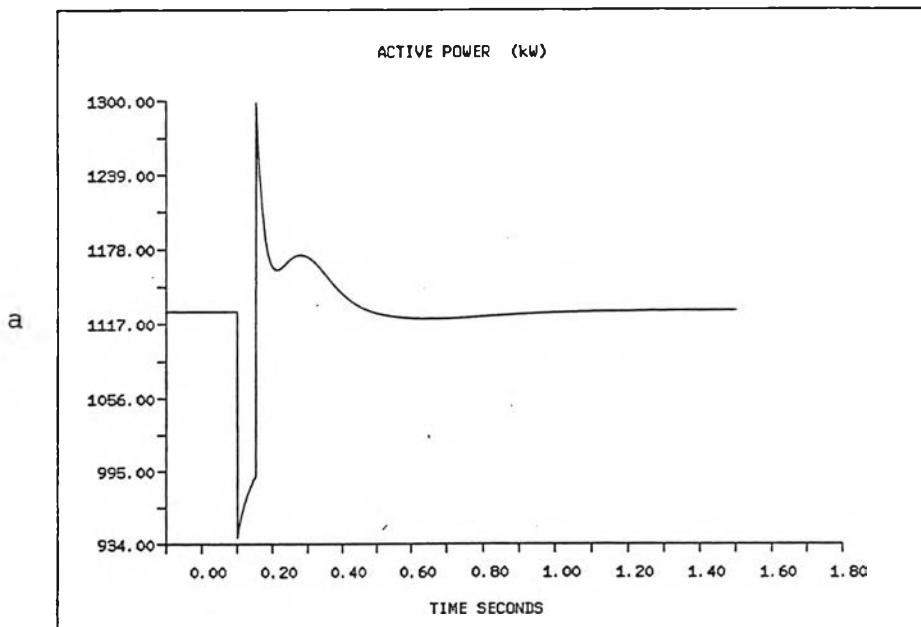


Fig. A4.48 Active power at bus 4 when gen. 1 disconnected

- a. used aggregate induction motor model
- b. used individual motor model

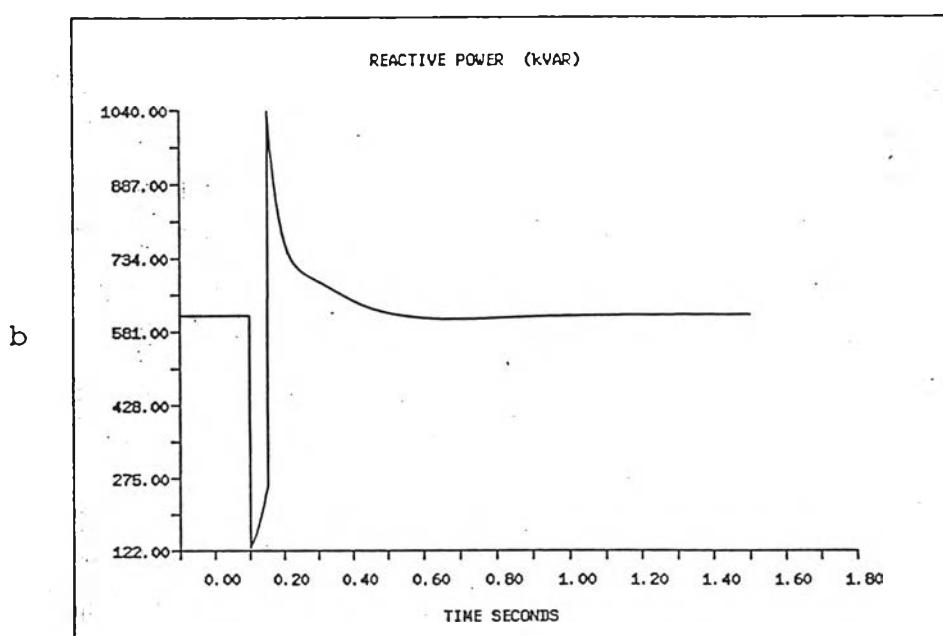
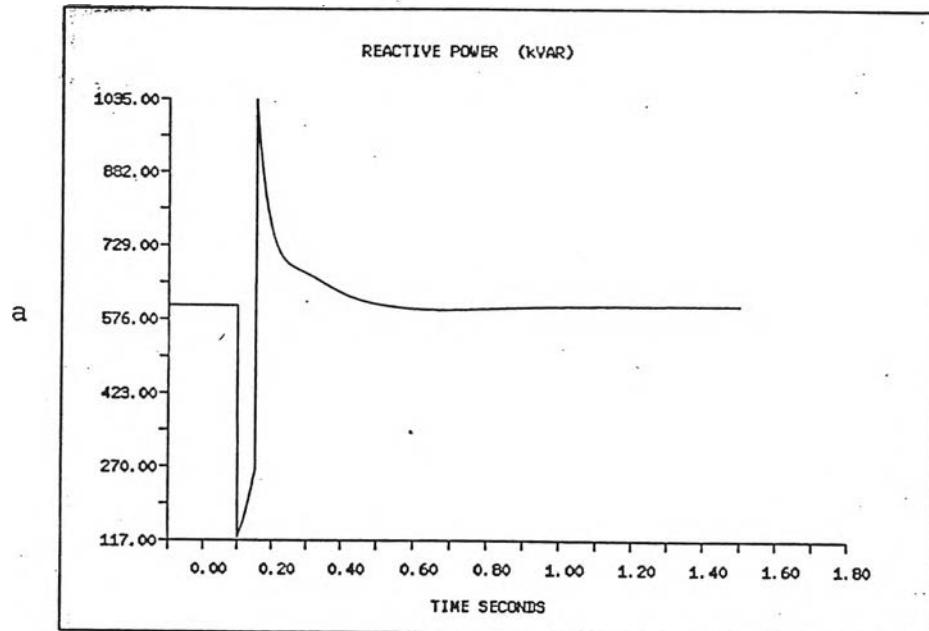


Fig. A4.49 Reactive power at bus 4 when gen. 1 disconnected

- a. used aggregate induction motor model
- b. used individual motor model

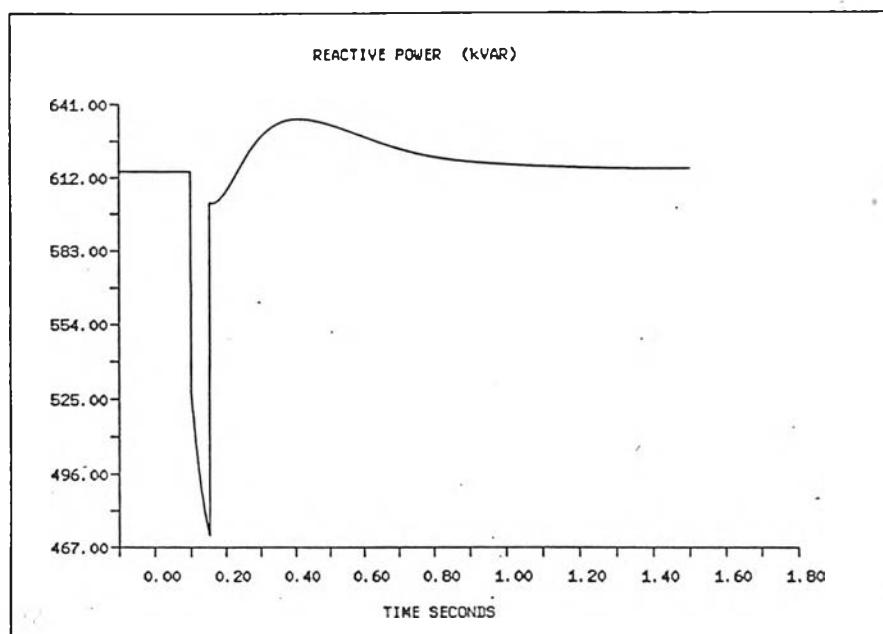
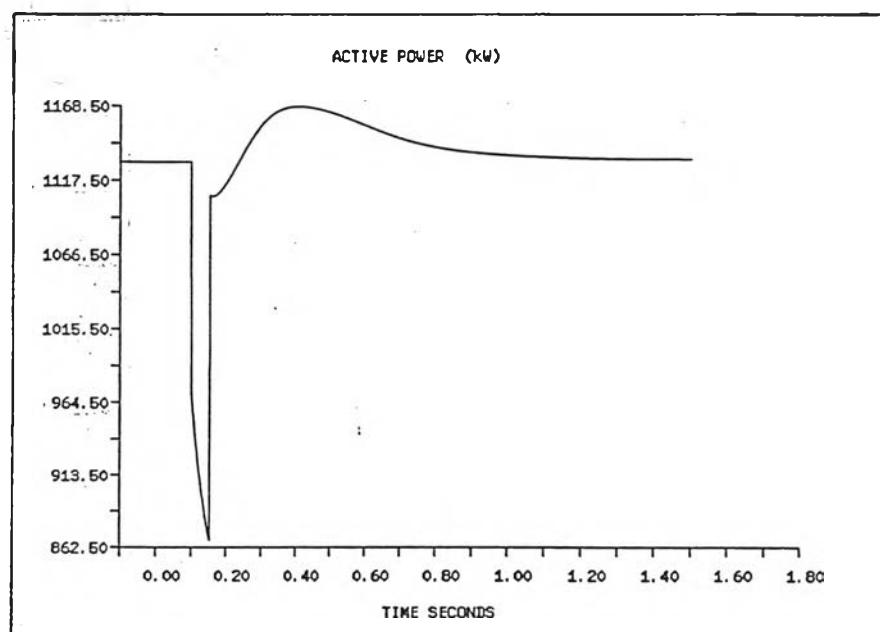


Fig. A4.50 Active and reactive power when gen. 1 disconnected used constant impedance model

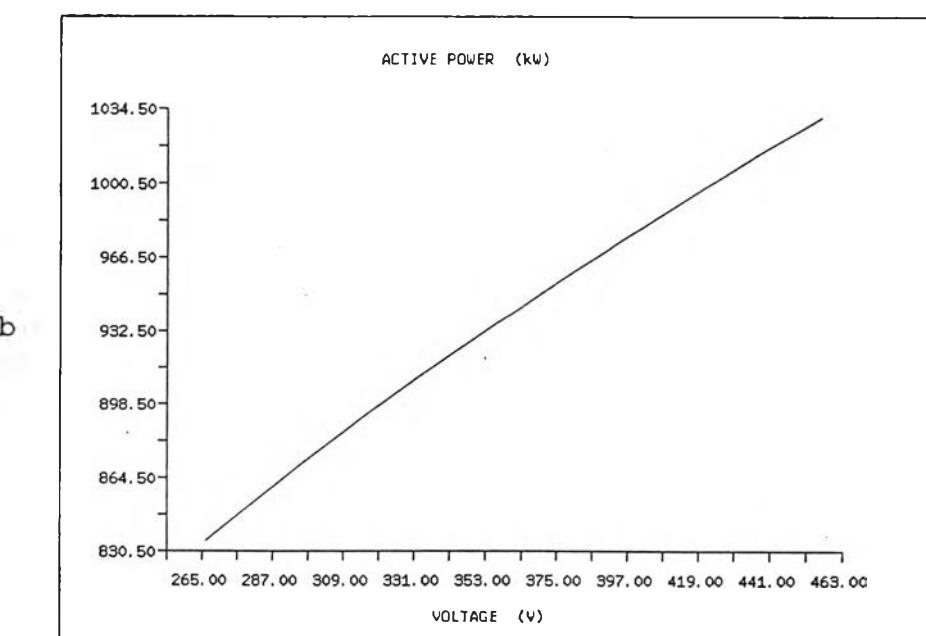
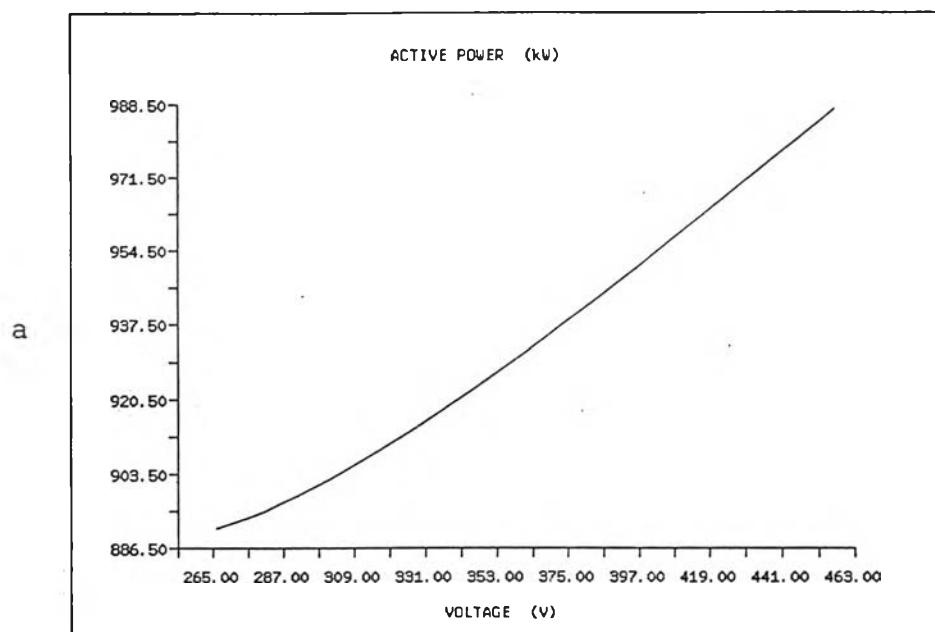


Fig. A4.51 Steady state P-V characteristic of composite load case 1

- a. used composite model
- b. used constant power model for motors and constant impedance model for other loads

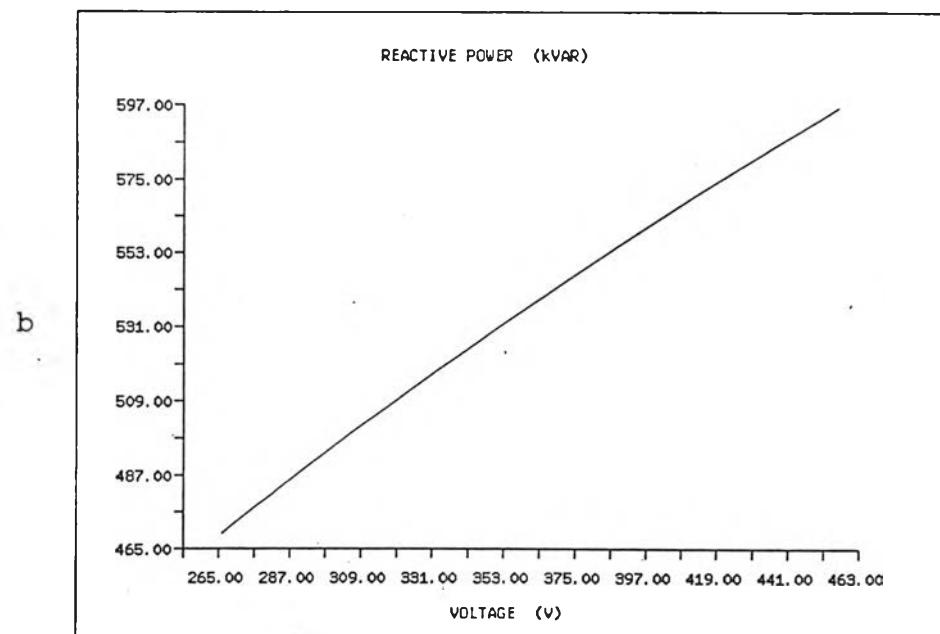
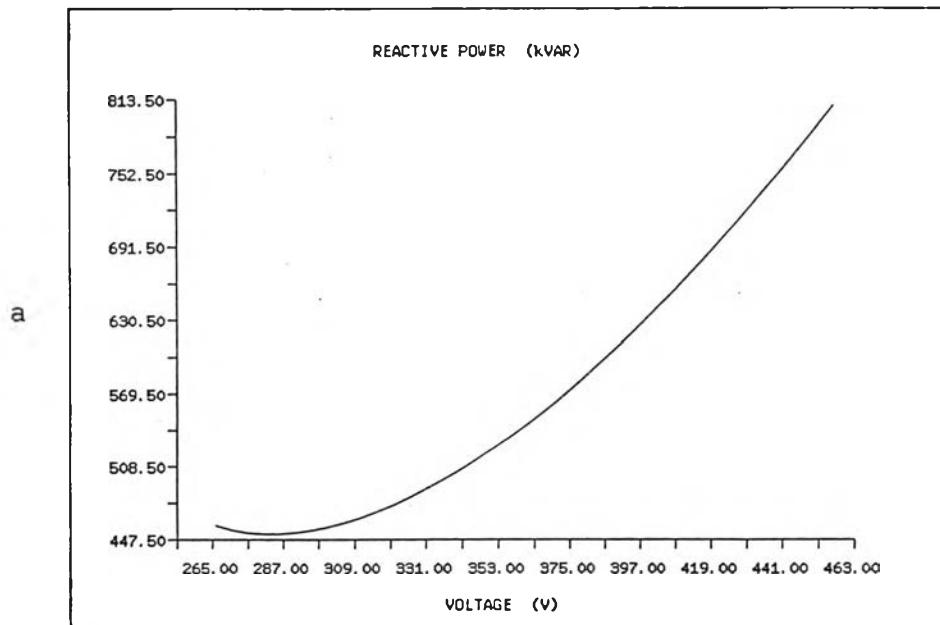


Fig. A4.52 Steady state Q-V characteristic of composite load case 1

- a. used composite model
- b. used constant power model for motors and constant impedance model for other loads

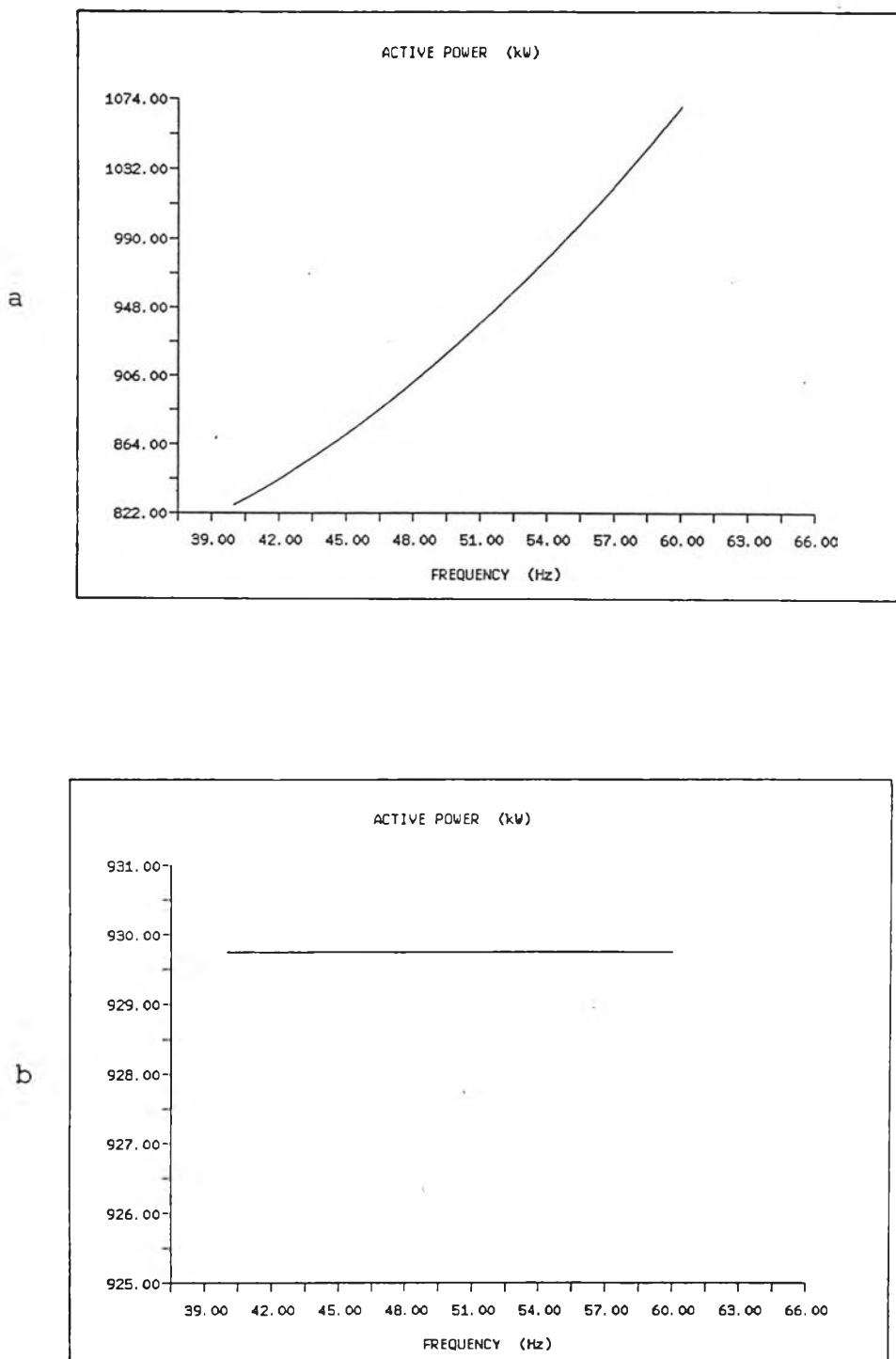


Fig. A4.53 Steady state P-f characteristic of composite load case 1

- a. used composite model
- b. used constant power model for motors and constant impedance model for other loads

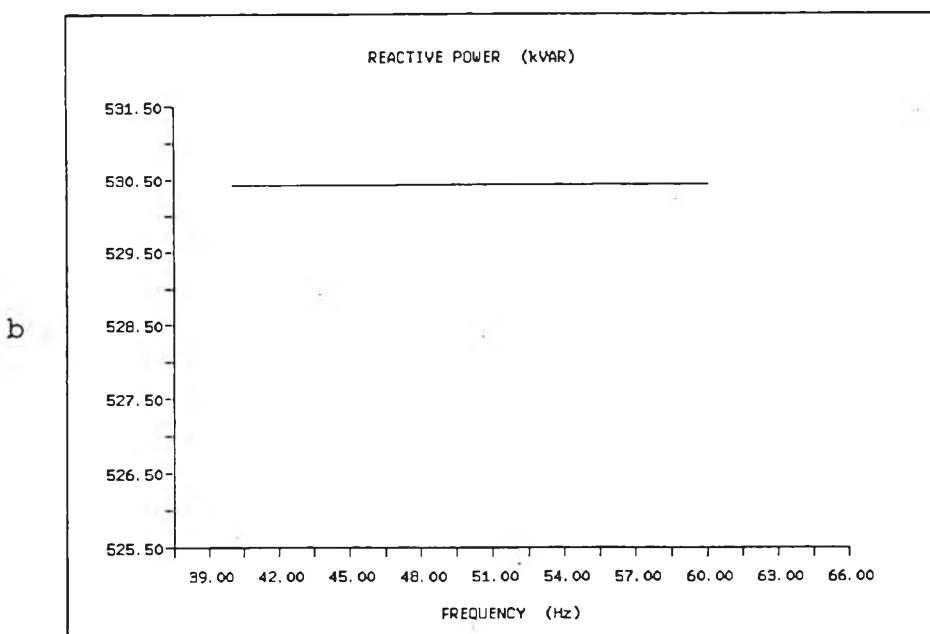
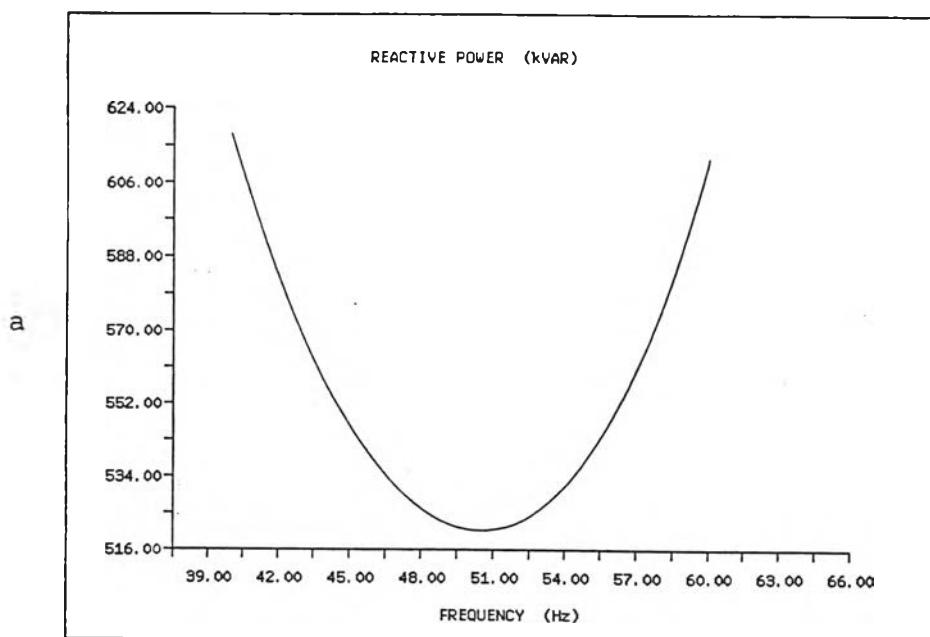


Fig. A4.54 Steady state Q-f characteristic of composite load case 1

- a. used composite model
- b. used constant power model for motors and constant impedance model for other loads

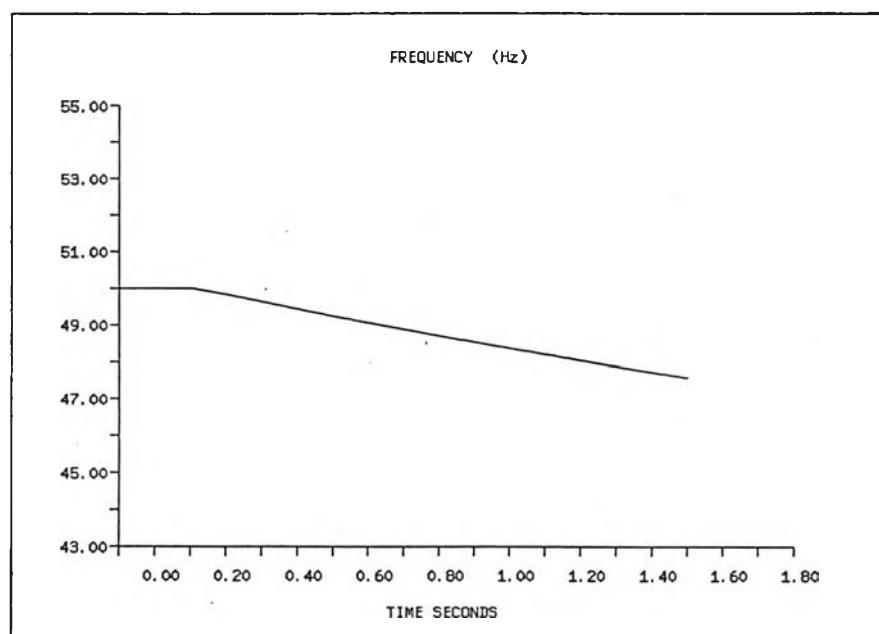
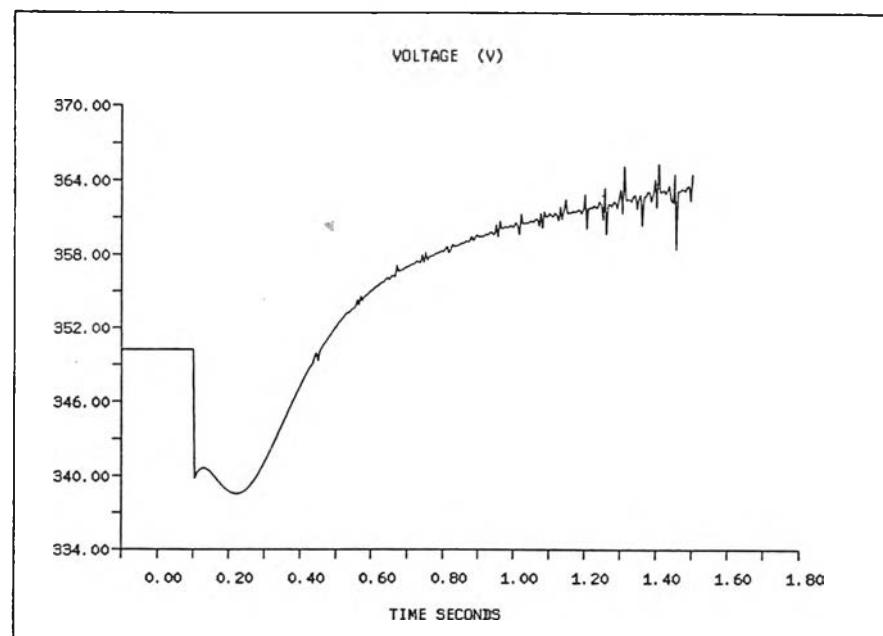


Fig. A4.55 Voltage and frequency at bus 4 during gen. 2 disconnected

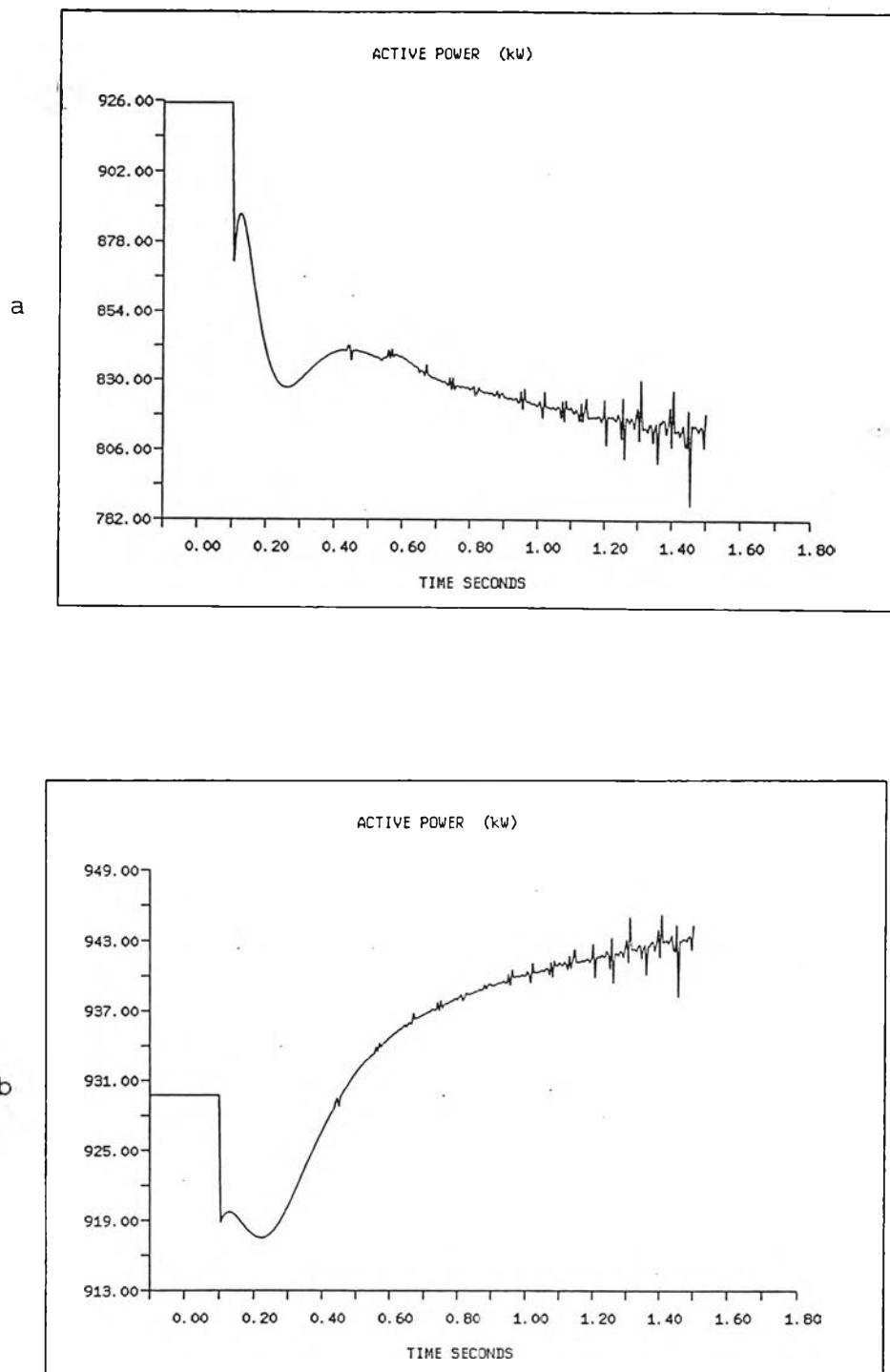


Fig. A4.56 Active power at bus 4 during gen. 2 disconnected

- a. used composite model
- b. used constant power model for motors and constant impedance model for other loads

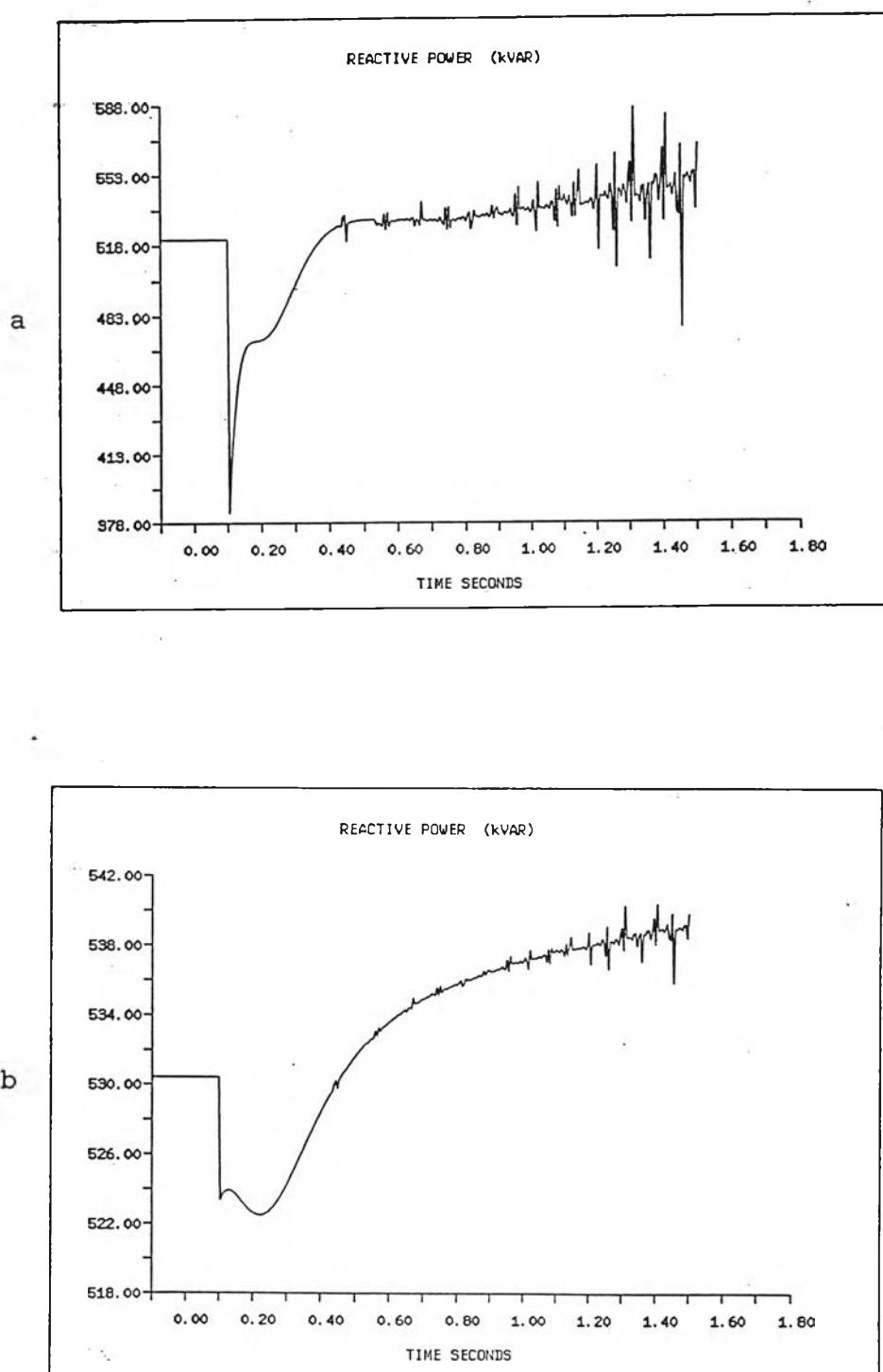


Fig. A4.57 Reactive power at bus 4 during gen. 2 disconnected

- a. used composite model
- b. used constant power model for motors and constant impedance model for other loads

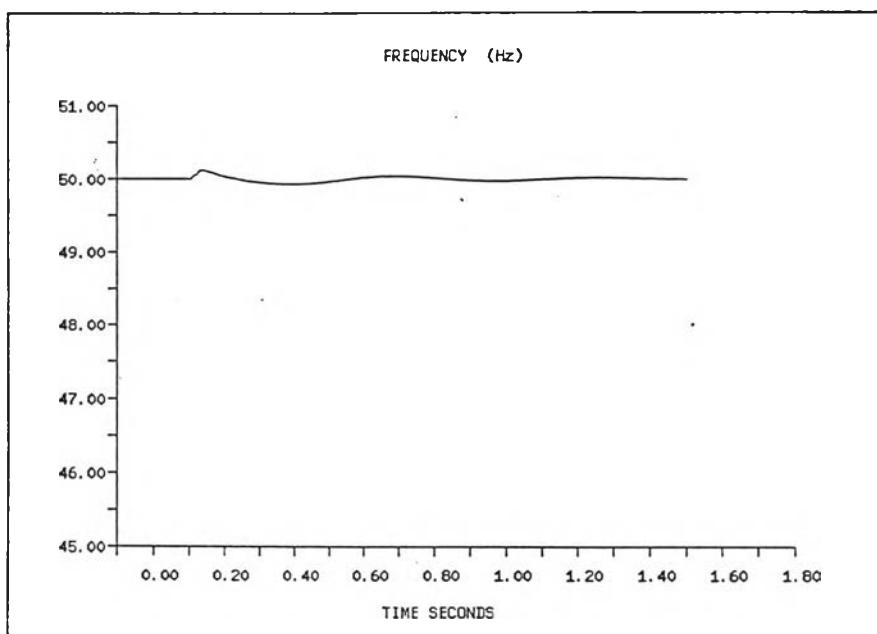
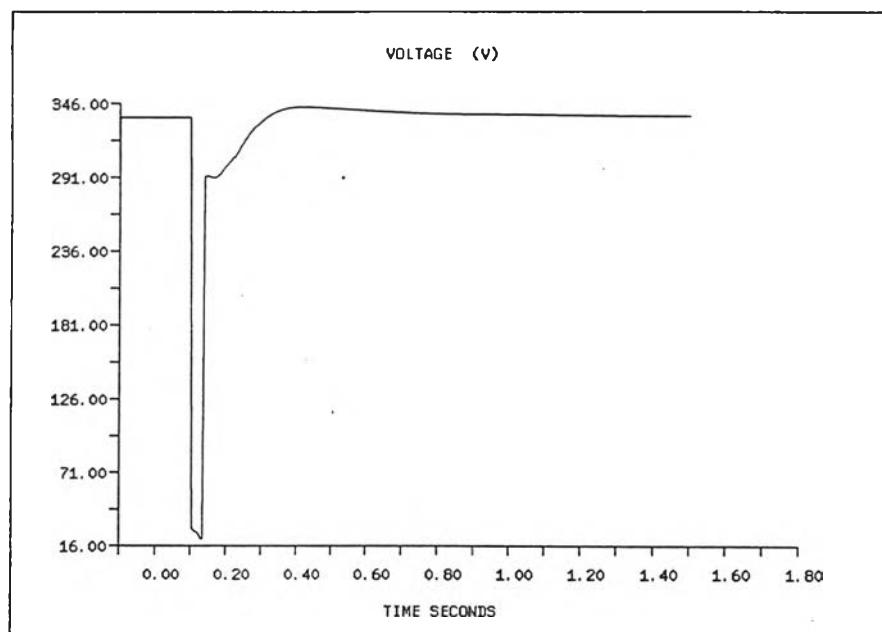


Fig.A4. 58 Voltage and frequency at bus 4 when 3 phase fault occurred at bus 3 (load case 2)

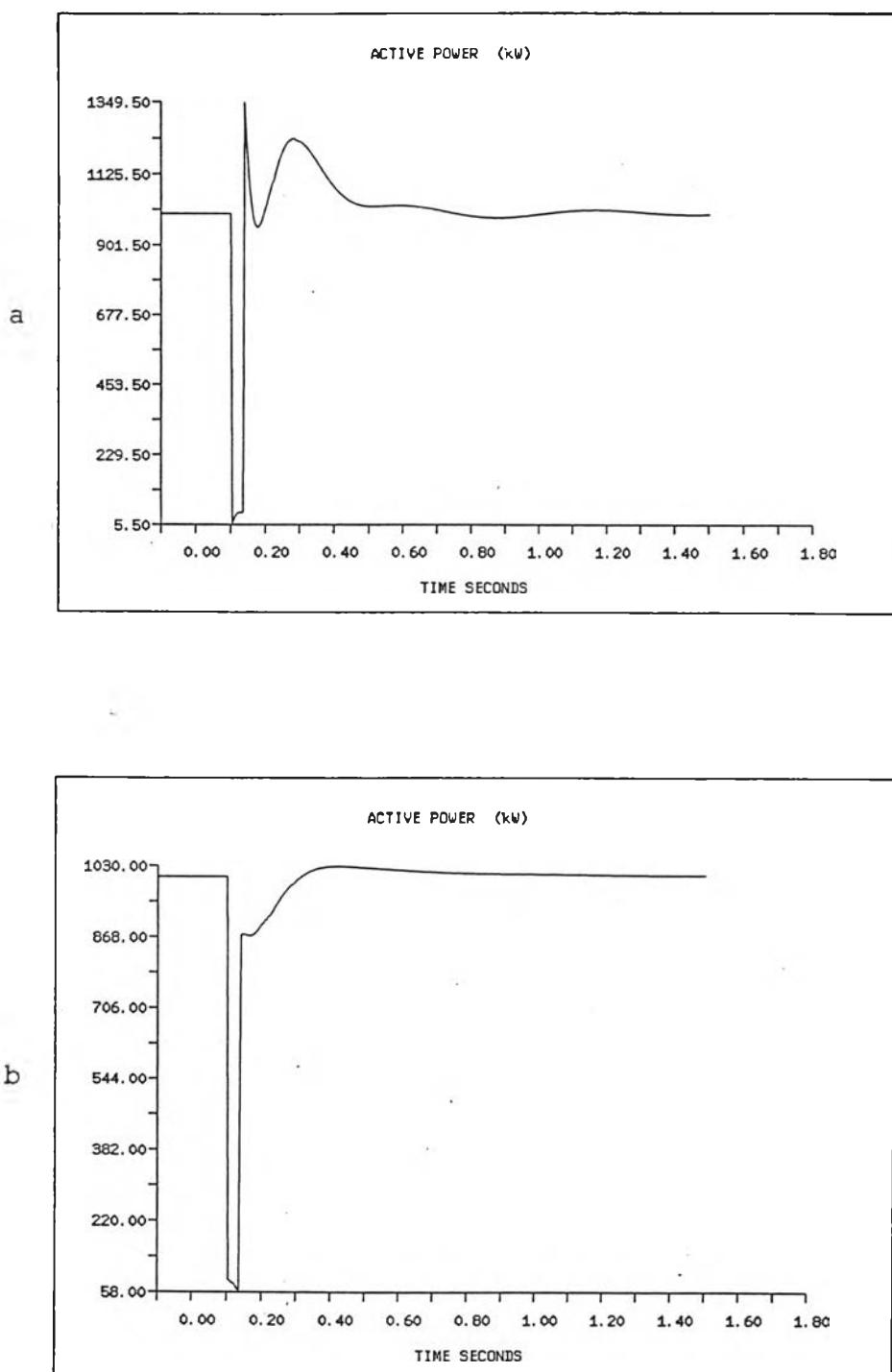


Fig.A4. 59 Active power at bus 4 when 3 phase fault occurred at bus 3

- a. used composite model
- b. used constant power model for motors and constant impedance model for other loads

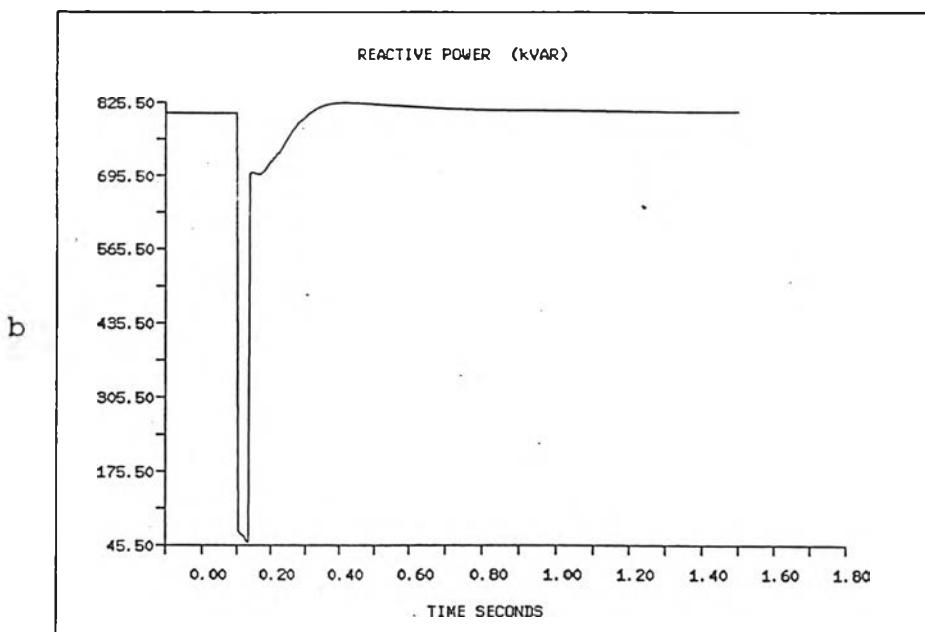
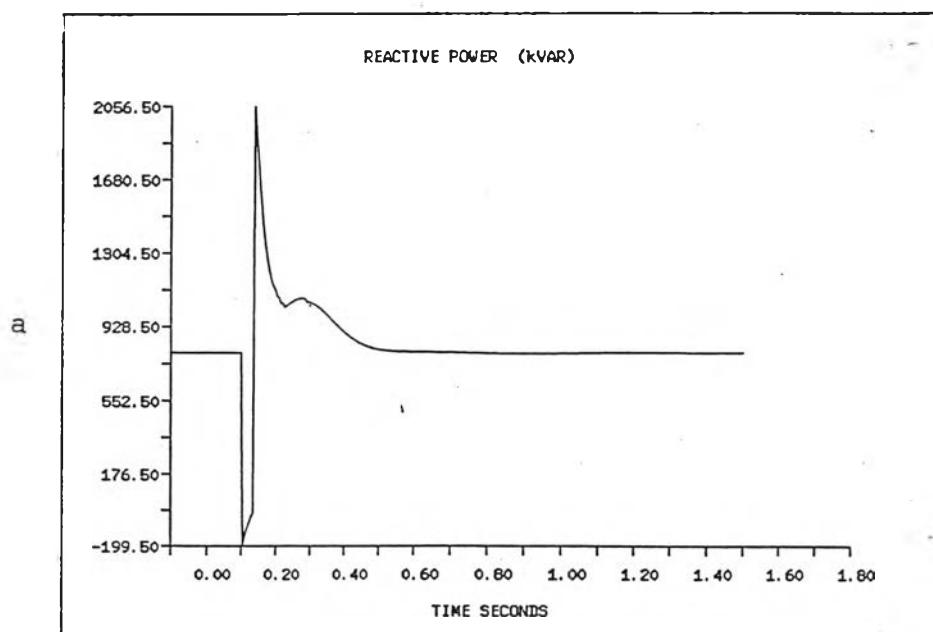


Fig. A4.60 Reactive power at bus 4 when 3 phase fault occurred at bus 3

- a. used composite model
- b. used constant power model for motors and constant impedance model for other loads

APPENDIX 5

EFFECT OF LOAD MODEL ON STABILITY STUDY

To illustrate effect of load model on stability study, the modified EGAT system was used to make stability test under two conditions of load model. The case studied is three phase fault on line. The two load models used are :

- 1) constant impedance
- 2) 20% constant current, 20% constant impedance,
60% constant power.

The results used the above two models are compared and shown in the figures on the following pages.

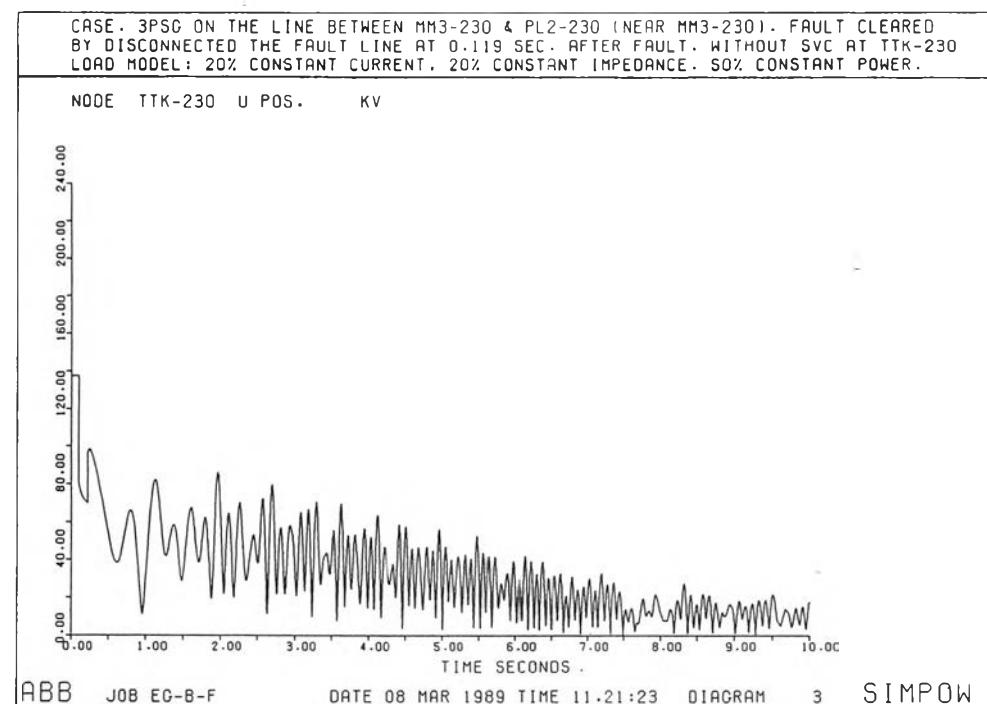
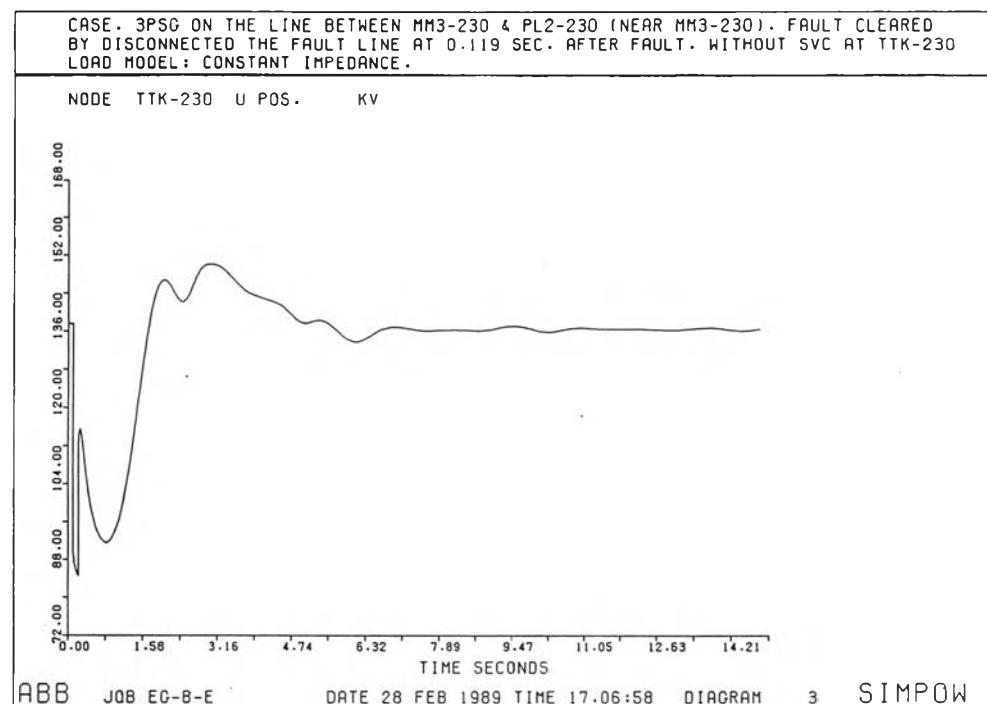


Figure A5.1 Comparison of voltage when used the two different load model

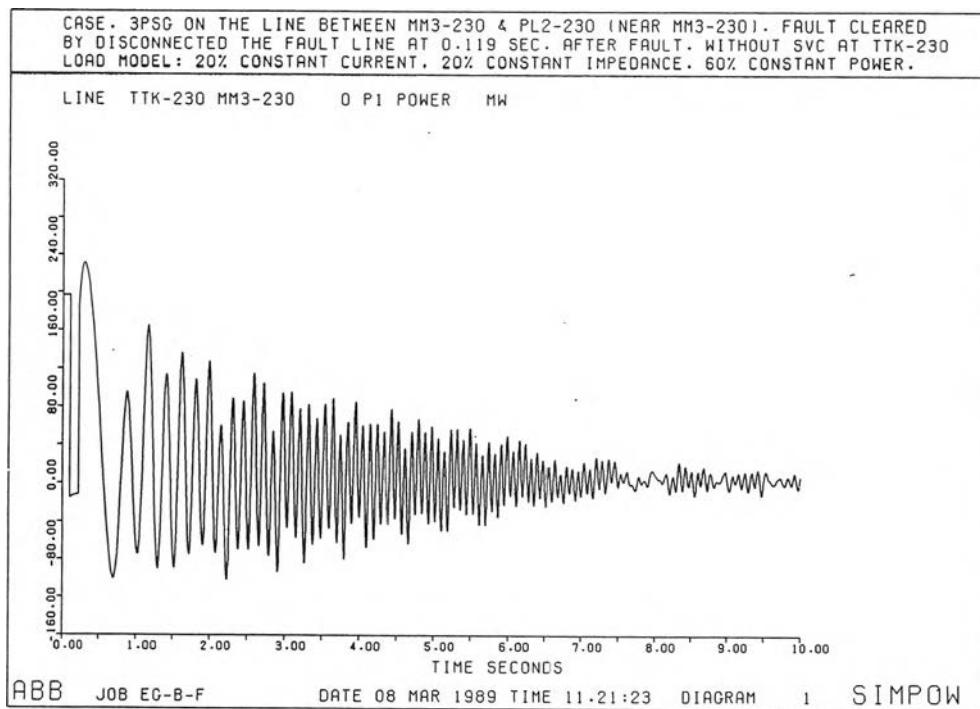
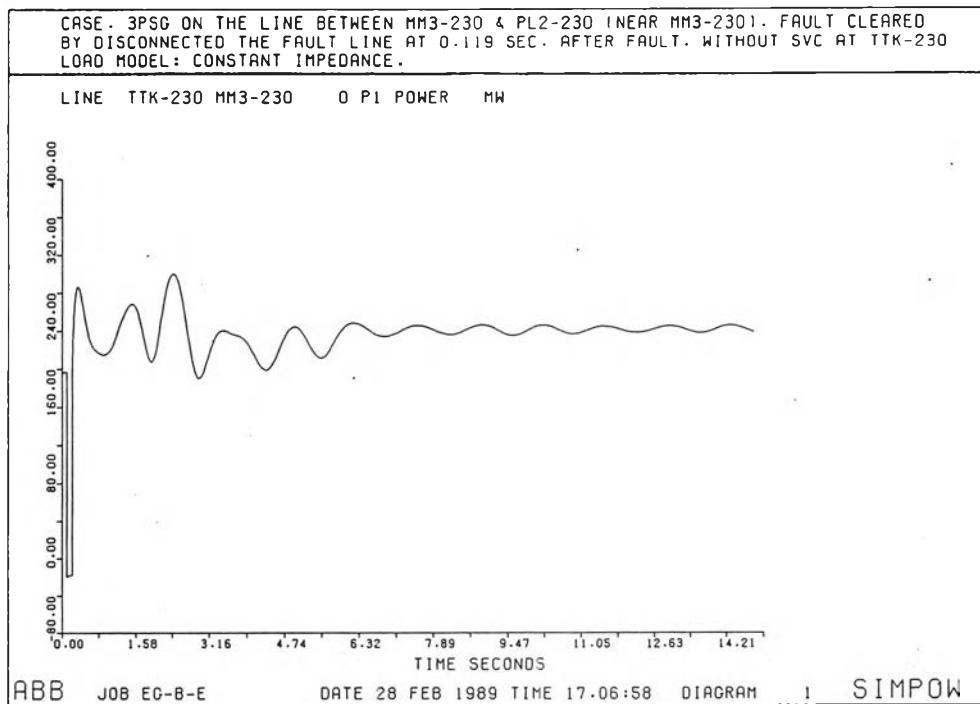


Figure A5.2 Comparison of active power

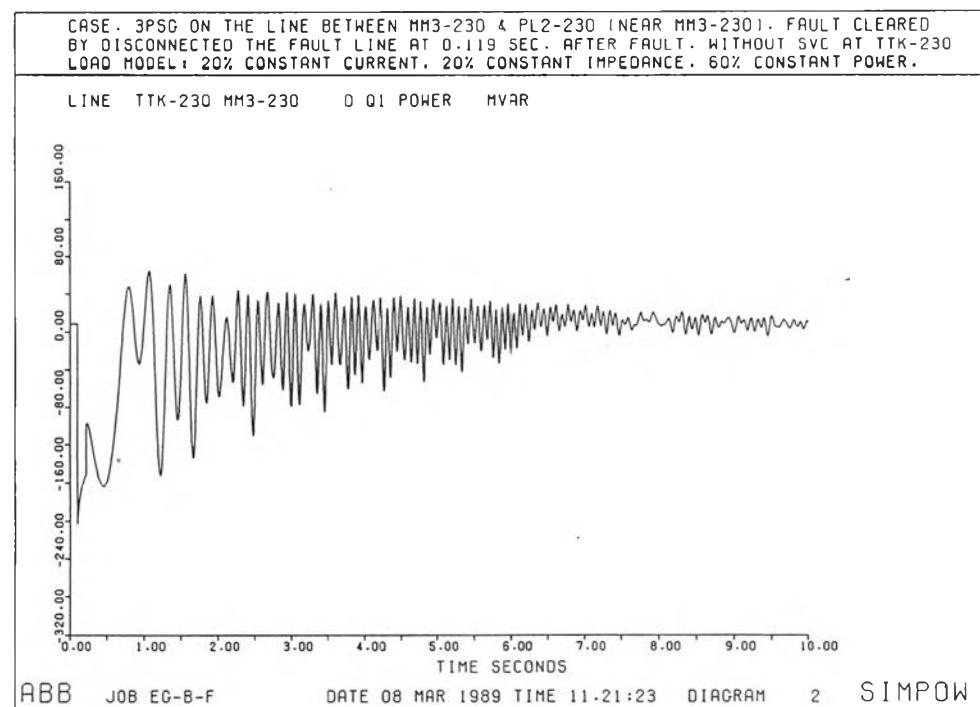
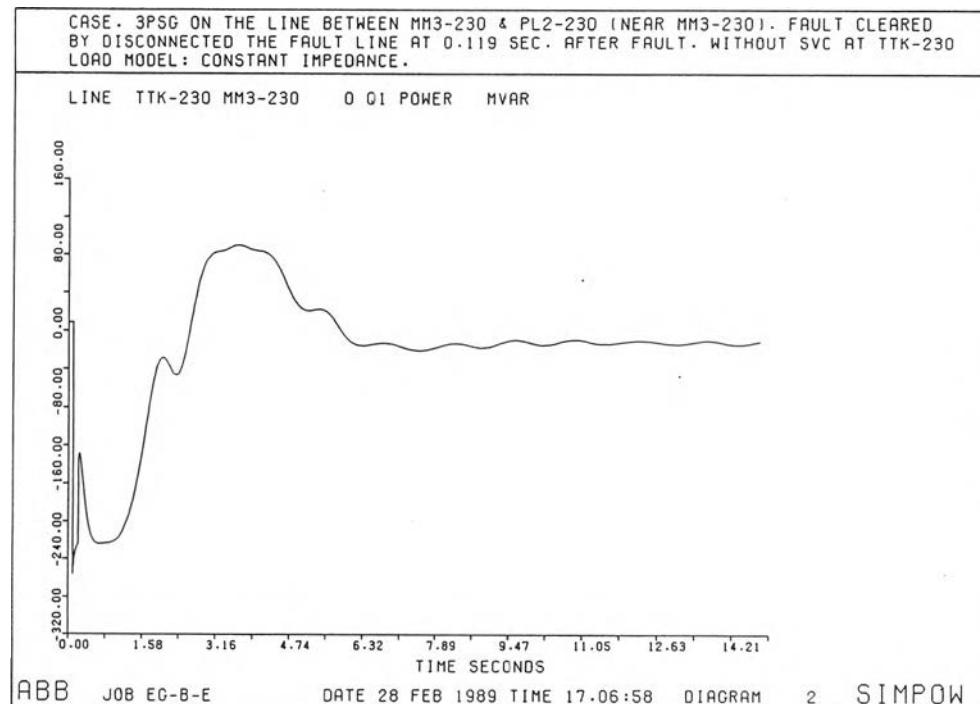


Figure A5.3 Comparison of reactive power

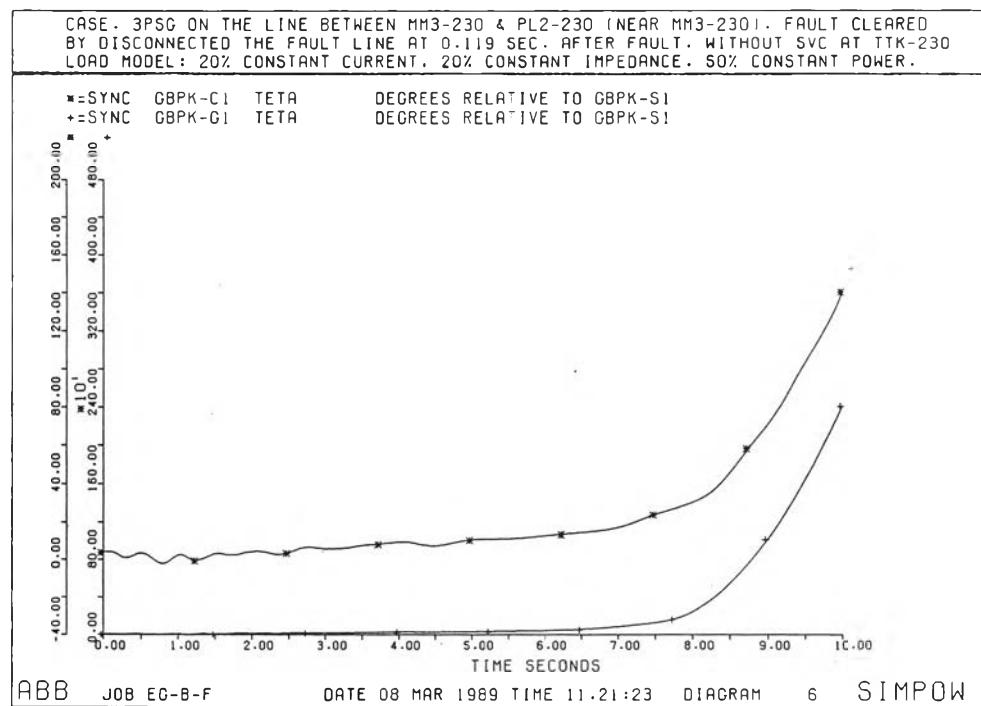
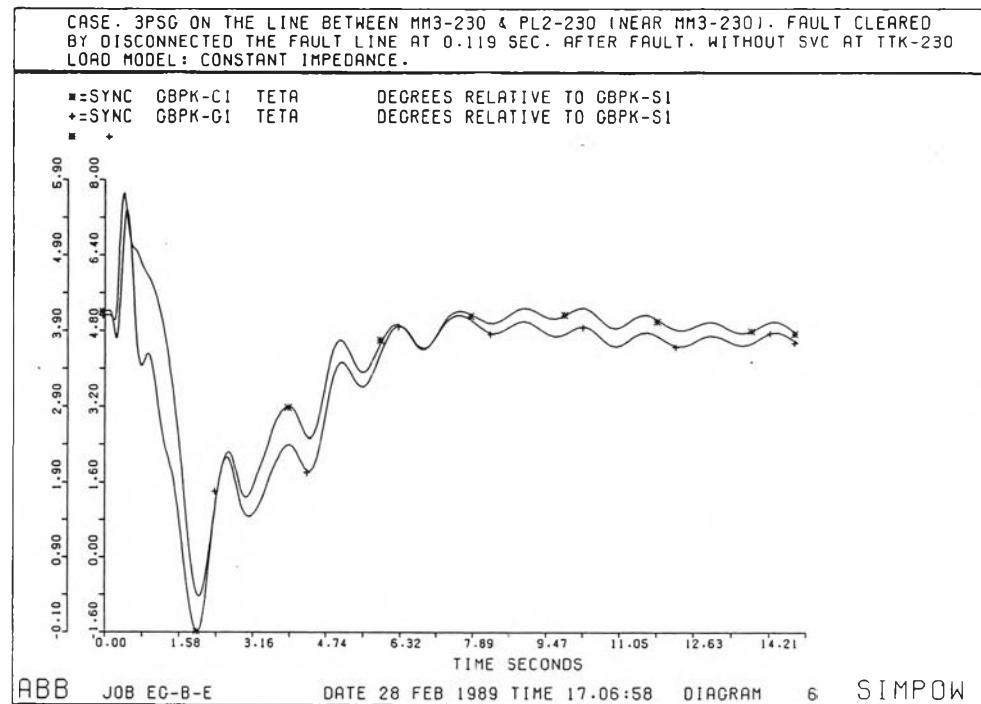


Figure A5.4 Relative angle of generators in the system

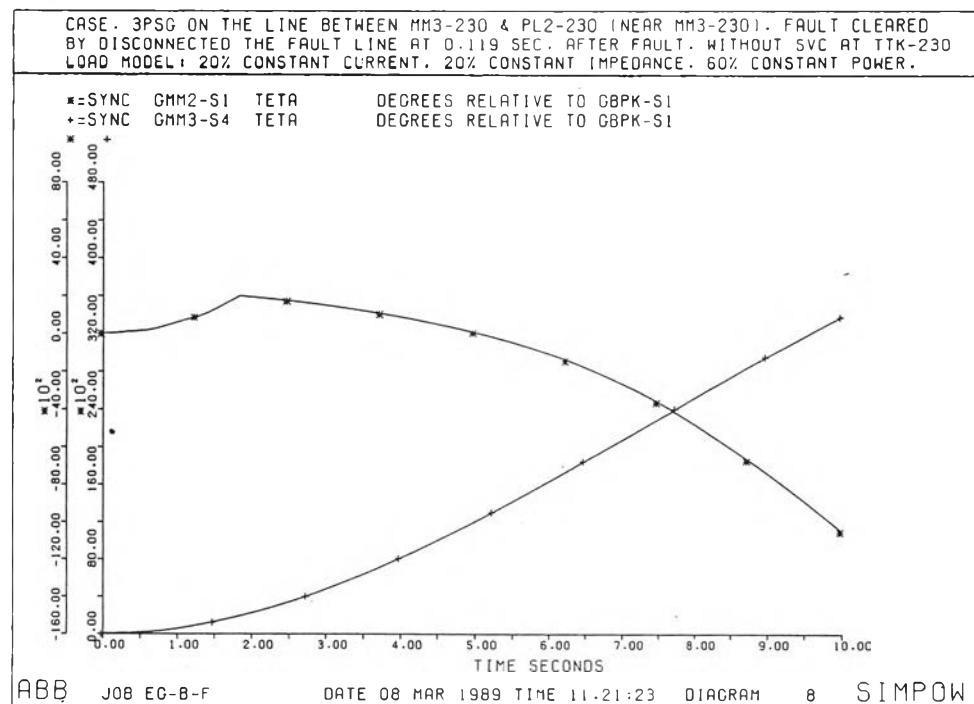
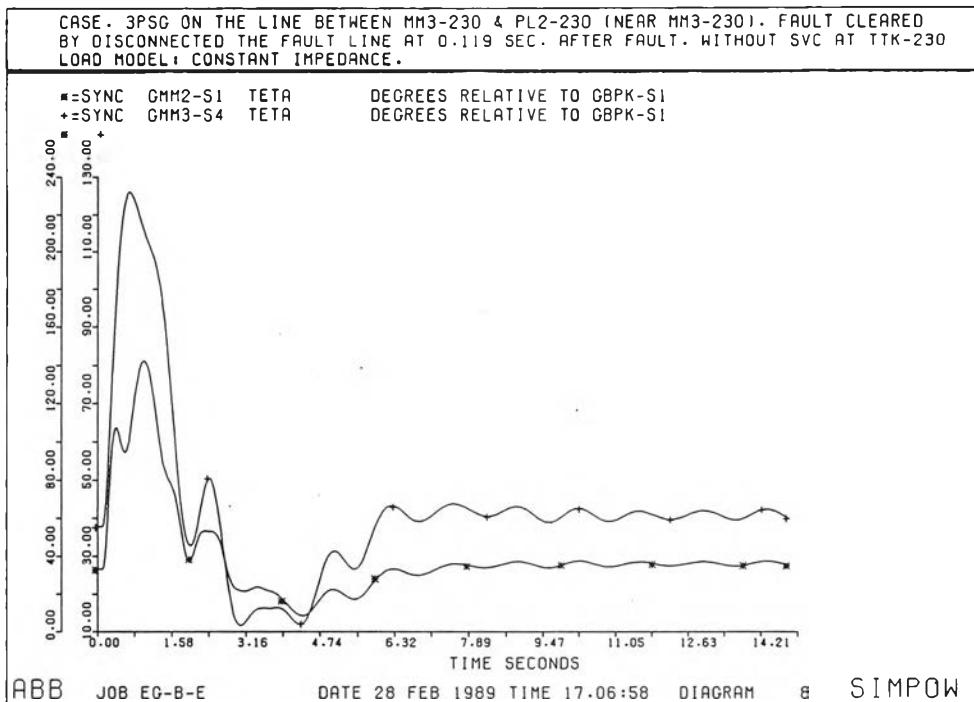


Figure A5.5 Relative angle of generators in the system

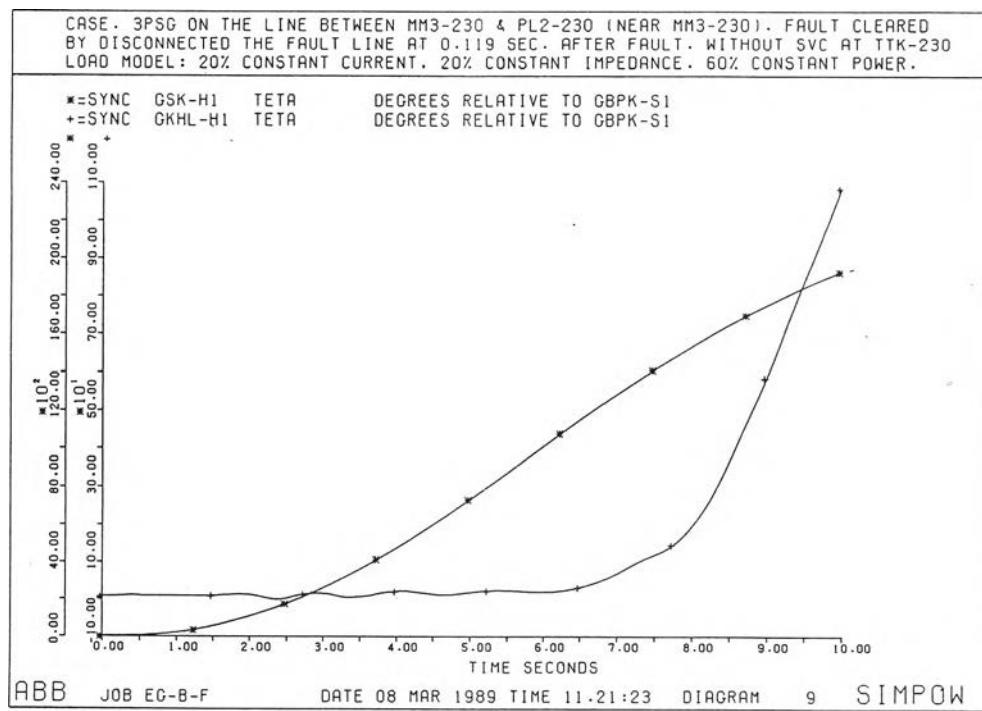
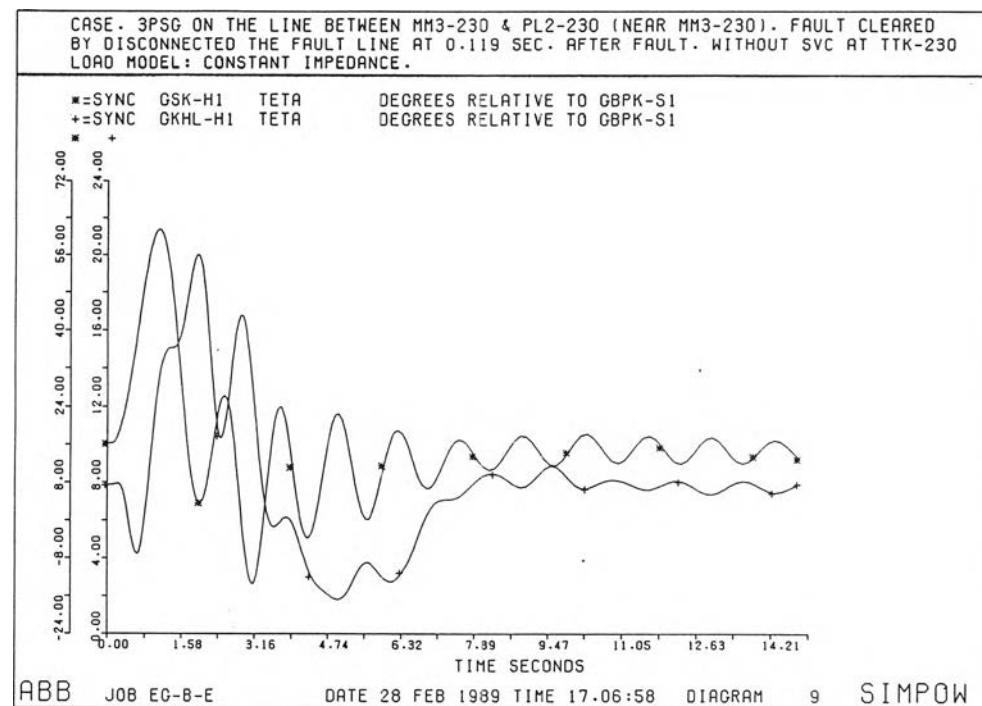


Figure A5.6 Relative angle of generators in the system

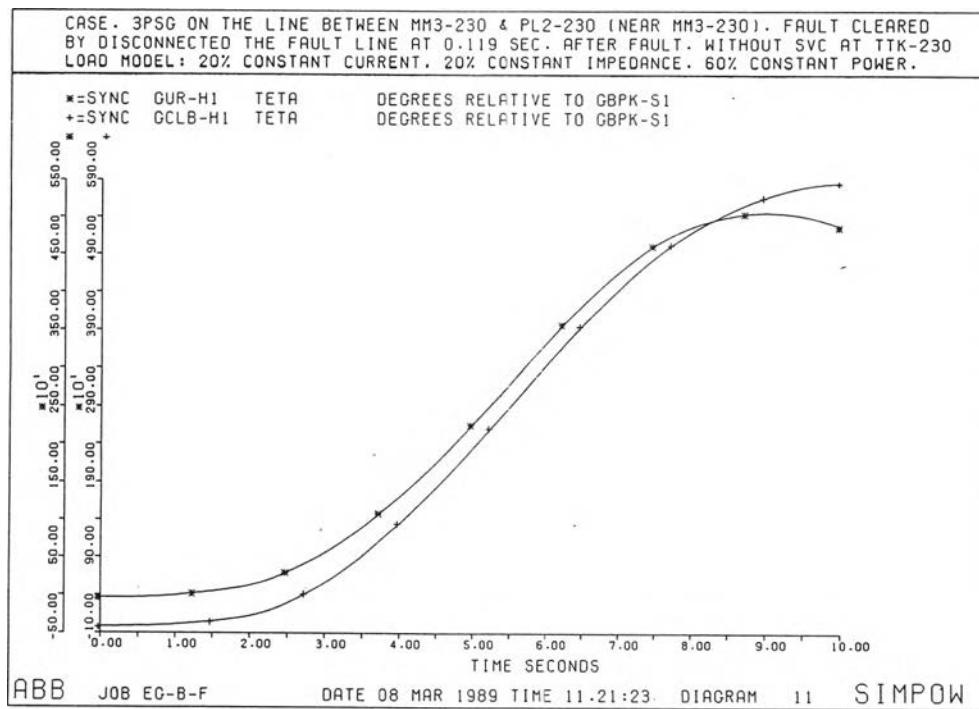
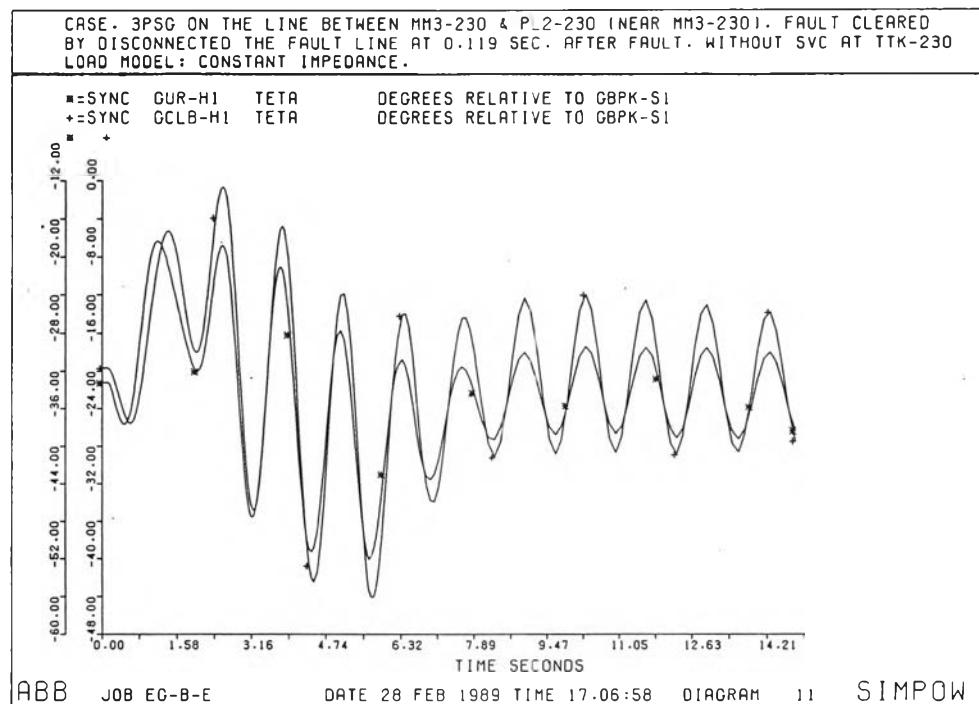


Figure A5.7 Relative angle of generators in the system

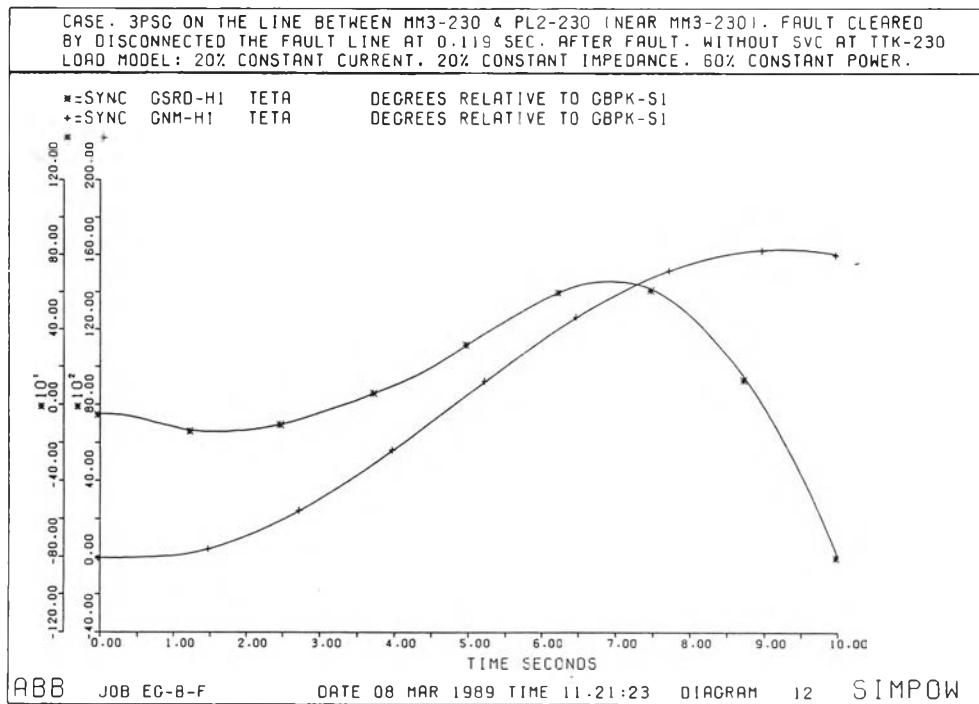
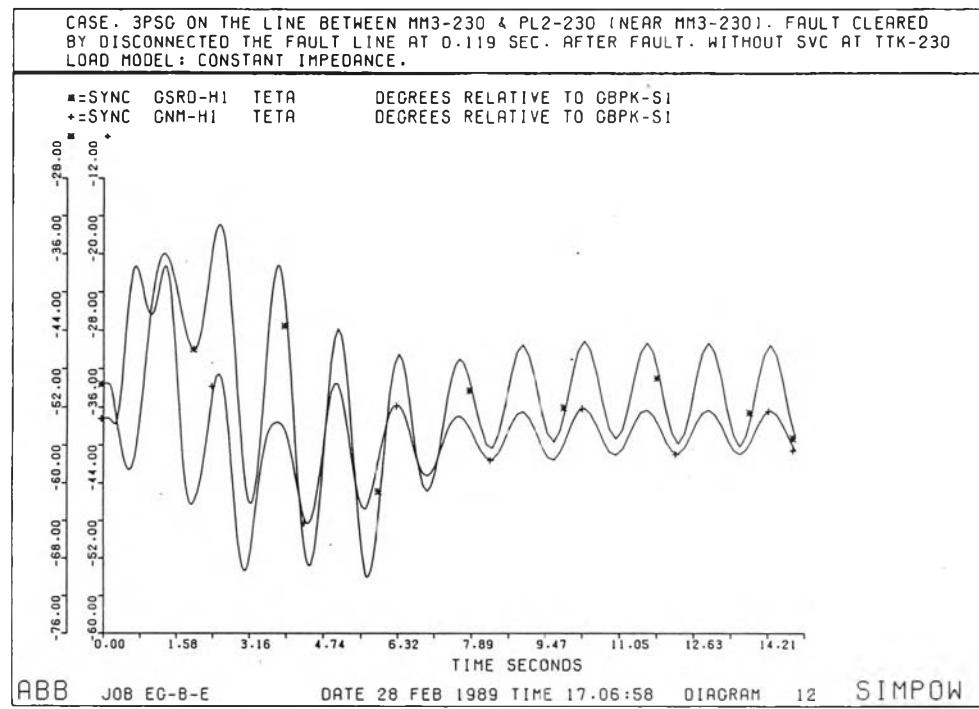


Figure A5.8 Relative angle of generators in the system

From the simulation results (e.g. Fig. A5.5, A5.8) it can be seen that when loads are modelled by constant impedance the system is stable and when loads are modelled by 20% constant current, 20% constant impedance and 60% constant power the system is unstable.

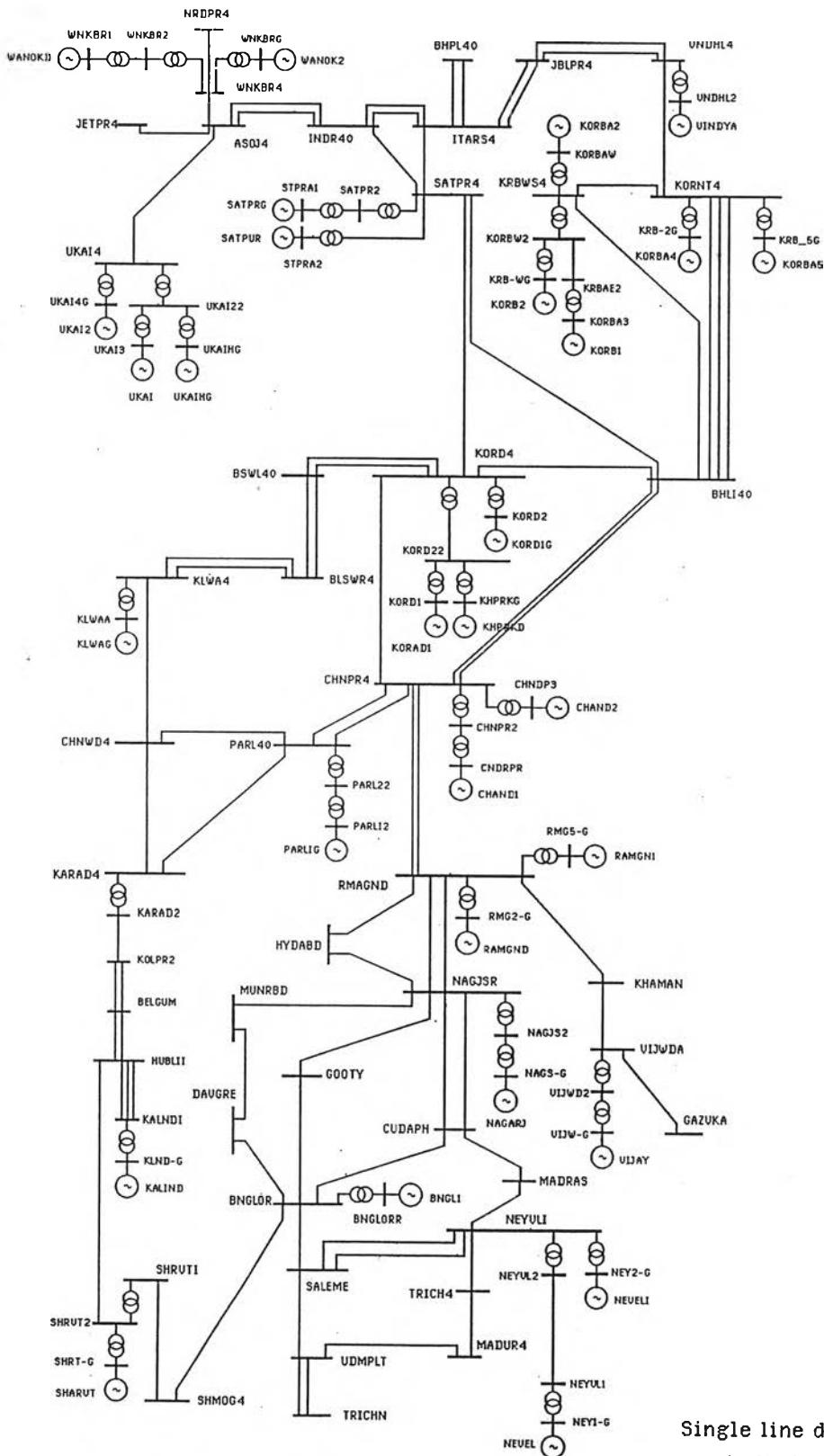
From this example it is clear that load model affects the stability results, so proper representation of loads is necessary.

APPENDIX 6

LOAD MODEL SIMULATION RESULTS OF A 91 BUSES SYSTEM

The studied system and its data are shown in figure A6.1 and table A6.1 respectively.

Bus MADRAS is selected for simulation study with 3 large motors connected to it. Dynamic responses of the motors when 3 phase fault occurred at bus CHNPR4 used induction motor model and constant current model are shown in figures on the following pages.



Single line diagram of studied system

Fig. A6.1 Studied system

Table A6.1 System Data

STUDIED SYSTEM DATA
 * *
 NODES

NAGS-G	UB	13.80	AREA 1
KLND-G	UB	11.00	AREA 1
KORBA3	UB=	13.80	AREA=1
KORBAW	UB=	15.75	AREA=1
STPRA1	UB=	15.75	AREA=1
STPRA2	UB=	15.75	AREA=1
KRB_2G	UB=	15.75	AREA=1
KRB_5G	UB=	21.00	AREA=1
VNDHL2	UB=	15.75	AREA=1
KRB_WG	UB=	15.75	AREA=1
KHPRKG	UB=	15.75	AREA=1
KORD1	UB=	14.30	AREA=1
KORD2	UB=	15.75	AREA=1
BNGLORR	UB=	13.8	AREA=1
KLWAA	UB=	13.8	AREA=1
CNDRPR	UB=	15.75	AREA=1
CHNDP3	UB=	15.75	AREA=1
WNKBR1	UB=	15.75	AREA=1
WNKBRG	UB=	15.75	AREA=1
UKAIHG	UB=	11.00	AREA=1
UKAI3	UB=	15.75	AREA=1
UKAI4G	UB=	15.75	AREA=1
VIJW-G	UB=	15.70	AREA=1
SHRT-G	UB=	11.00	AREA=1
NEY2-G	UB=	15.70	AREA=1
NEY1-G	UB=	11.00	AREA=1
RMG5-G	UB=	21.00	AREA=1
RMG2-G	UB	16.50	AREA 1
PARLI2	UB	15.75	AREA 1
VIJWD2	UB=	220.00	AREA=1
KALNDI	UB	220.00	AREA 1
HUBLII	UB	220.00	AREA 1
RMNGD2	UB	220.00	AREA 1
NEYVL1	UB	220.0	AREA 1
NEYVL2	UB	220.0	AREA 1
BELGUM	UB	220.00	AREA 1
SATPR2	UB=	220.00	AREA=1
KORBW2	UB=	220.00	AREA=1
KHPRKD	UB=	220.00	AREA=1
KORD22	UB=	220.00	AREA=1
CHNPR2	UB=	220.00	AREA=1
KARAD2	UB=	220.00	AREA=1
KOLPR2	UB=	220.00	AREA=1
WNKBR2	UB=	220.00	AREA=1
UKAI22	UB=	220.00	AREA=1
SHRVVT2	UB	220.0	AREA 1
NAGJS2	UB	220.0	AREA 1
KRBAE2	UB	220.0	AREA 1
PARL22	UB	220.0	AREA 1
RMAGND	UB=	400.00	AREA=1
KHAMAN	UB=	400.00	AREA=1
VIJWDA	UB=	400.00	AREA=1
GAZVKA	UB=	400.00	AREA=1
NAGJSR	UB=	400.00	AREA=1
CUDAPH	UB=	400.00	AREA=1
MADRAS	UB=	400.00	AREA=1
NEYVLI	UB=	400.00	AREA=1
SALEME	UB=	400.00	AREA=1
TRICH4	UB=	400.00	AREA=1
MADUR4	UB=	400.00	AREA=1
UDMPLT	UB=	400.00	AREA=1
BNGLOR	UB=	400.00	AREA=1
DAVGRE	UB=	400.00	AREA=1
MUNRBD	UB=	400.00	AREA=1
HYDABD	UB=	400.00	AREA=1
TRICHN	UB=	400.00	AREA=1
SHMOG4	UB=	400.00	AREA=1
SHRVTI	UB=	400.00	AREA=1
GOOTYY	UB=	400.00	AREA=1
KRBWS4	UB=	400.00	AREA=1
SATPR4	UB=	400.00	AREA=1
KORNT4	UB=	400.00	AREA=1
VNDHL4	UB=	400.00	AREA=1
BHLI40	UB=	400.00	AREA=1
JBLPR4	UB=	400.00	AREA=1
ITARS4	UB=	400.00	AREA=1
INDR40	UB=	400.00	AREA=1
BHPL40	UB=	400.00	AREA=1
KORD4	UB=	400.00	AREA=1
CHNPR4	UB=	400.00	AREA=1
PARL40	UB=	400.00	AREA=1
KARAD4	UB=	400.00	AREA=1
CHNW4D	UB=	400.00	AREA=1
KLWA4	UB=	400.00	AREA=1
BLSWR4	UB=	400.00	AREA=1
BSWL40	UB=	400.00	AREA=1
WNKBR4	UB=	400.00	AREA=1
UKAI4	UB=	400.00	AREA=1
ASOJ4	UB=	400.00	AREA=1
NRDPR4	UB=	400.00	AREA=1
JETPR4	UB=	400.00	AREA=1

END

TRANSFORMERS

NAGS-G	NAGJS2	SN 100	UN1 13.8	UN2 220.	EX12 0.0227
KLND-G	KALNDI	SN 100	UN1 11.0	UN2 220.	EX12 0.0188
NEY1-G	NEYVLI	SN 100	UN1 11.0	UN2 220.	EX12 0.0450
NEYVL2	NEYVLI	SN 100	UN1 220.	UN2 400.	EX12 0.0381
SHRT-G	SHRVT2	SN 100	UN1 11.00	UN2 220.00	EX12 0.0143
VIJW-G	VIJWD2	SN 100	UN1 15.75	UN2 220.00	EX12 0.0189
RMNGD2	RMAGND	SN 100	UN1 220.00	UN2 400.00	EX12 0.0381
WNKBR1	WNKBR2	SN 100	UN1 15.75	UN2 220.00	EX12 0.0280
UKAIHG	UKAI22	SN 100	UN1 11.00	UN2 220.00	EX12 0.0463
NEY2-G	NEYVLI	SN=100	UN1= 15.75	UN2=400.00	EX12=0.0189
RMG2-G	RMAGND	SN=100	UN1= 15.75	UN2=400.00	EX12=0.0199
RMG5-G	RMAGND	SN=100	UN1= 21.00	UN2=400.00	EX12=0.0119
SHRVT2	SHRVTI	SN=100	UN1=220.00	UN2=400.00	EX12=0.0381
VIJWD2	VIJWDA	SN=100	UN1=220.00	UN2=400.00	EX12=0.0381
NAGJS2	NAGJSR	SN=100	UN1=220.00	UN2=400.00	EX12=0.0381
KORBA3	KRBAE2	SN=100	UN1= 13.80	UN2=220.00	EX12=0.0481
KORBW	KRBWS4	SN=100	UN1= 15.80	UN2=400.00	EX12=0.0580
STPRA1	SATPR2	SN=100	UN1= 15.80	UN2=220.00	EX12=0.0560
STPRA2	SATPR4	SN=100	UN1= 15.75	UN2=400.00	EX12=0.0580
KRB_2G	KORNT4	SN=100	UN1= 15.80	UN2=400.00	EX12=0.0187
KRB_5G	KORNT4	SN=100	UN1= 21.00	UN2=400.00	EX12=0.0125
VNDHL2	VNDHL4	SN=100	UN1= 15.75	UN2=400.00	EX12=0.0112
KRB_WG	KORBW2	SN=100	UN1= 15.80	UN2=220.00	EX12=0.0580
SATPR2	SATPR4	SN=100	UN1=220.00	UN2=400.00	EX12=0.0300
KORBW2	KRBWS4	SN=100	UN1=220.00	UN2=400.00	EX12=0.0300
KHPRKG	KHPRKD	SN=100	UN1= 15.75	UN2=220.00	EX12=0.0290
KORD1	KORD22	SN=100	UN1= 14.30	UN2=220.00	EX12=0.0278
CNDRPR	CHNPR2	SN=100	UN1= 15.75	UN2=220.00	EX12=0.0600
PARL12	PARL22	SN=100	UN1= 15.75	UN2=220.00	EX12=0.0300
CHNDP3	CHNPR4	SN=100	UN1= 15.75	UN2=400.00	EX12=0.0250
KORD22	KORD4	SN=100	UN1=220.00	UN2=400.00	EX12=0.0417
CHNPR2	CHNPR4	SN=100	UN1=220.00	UN2=400.00	EX12=0.0500
PARL22	PARL40	SN=100	UN1=220.00	UN2=400.00	EX12=0.0500
KARAD2	KARAD4	SN=100	UN1=220.00	UN2=400.00	EX12=0.0397
WNKBRG	WNKBR4	SN=100	UN1= 15.75	UN2=400.00	EX12=0.0187
UKAI3	UKAI22	SN=100	UN1= 15.75	UN2=220.00	EX12=0.0280
UKAI4G	UKAI4	SN=100	UN1= 15.75	UN2=400.00	EX12=0.0560
WNKBR2	WNKBR4	SN=100	UN1=220.00	UN2=400.00	EX12=0.0300
UKAI22	UKAI4	SN=100	UN1=220.00	UN2=400.00	EX12=0.0500
KORD2	KORD4	SN 100	UN1=15.75	UN2=400.00	EX12=0.025
KLWA4	KLWA4	SN 100	UN1=13.8	UN2=400.00	EX12=0.025
BNGLORR	BNGLOR	SN 100	UN1=13.8	UN2=400.00	EX12=0.025

END

SHUNT IMPEDANCES

KOLPR2	Q = -80.0	UN 220.
BELGUM	Q = -40.0	UN 220.0
KARAD2	Q = -50.	UN 220.
KHAMAN	Q = 90.	UN 400.00
RMAGND	Q = 90.	UN 400.00
VIJWDA	Q = 114.	UN 400.00
GAZVKA	Q = 57.	UN 400.00 NCON 1
NAGJSR	Q = 282.	UN 400.00
CUDAPH	Q = 180.	UN 400.00
MADRAS	Q = 90.	UN 400.00
NEYVLI	Q = 90.00	UN 400.00
SALEME	Q = 135.00	UN 400.00
TRICH4	Q = 45.00	UN 400.00
MADUR4	Q = 90.00	UN 400.00
UDMPLT	Q = 90.00	UN= 400.00
BNGLOR	Q = 236.00	UN= 400.00
DAVGRE	Q = 135.00	UN= 400.00
MUNRBD	Q = 102.00	UN= 400.00
HYDABD	Q = 45.00	UN= 400.00
TRICHN	Q = 45.00	UN= 400.00
SHMOG4	Q = 090.00	UN= 400.00
SHRVTI	Q = 45.00	UN= 400.00
GOOTYY	Q = 90.00	UN= 400.00
SATPR4	Q = 90.60	UN= 400.00
KORNT4	Q = 45.30	UN= 400.00
VNDHL4	Q = 159.70	UN= 400.00
BHLI40	Q = 415.80	UN= 400.00
JBLPR4	Q = 205.00	UN= 400.00
ITARS4	Q = 181.20	UN= 400.00
INDR40	Q = 226.50	UN= 400.00
KORD4	Q = 320.30	UN= 400.00
CHNPR4	Q = 280.60	UN= 400.00
PARL40	Q = 150.00	UN= 400.00
KARAD4	Q = 50.00	UN= 400.00
CHNW4	Q = 90.60	UN= 400.00
KLWA4	Q = -50.00	UN= 400.00
BSWL40	Q = 50.00	UN= 400.00
UKAI4	Q = 45.30	UN= 400.00
ASOJ4	Q = 135.90	UN= 400.00
JETPR4	Q = 45.00	UN= 400.00
NEY1-G	Q = 100.00	UN= 11.00

END

```

LOADS
RMNGD2 P 246.659 Q 55.1927
KALNDI P 68.96 Q 32.93
VIJWD2 P 383.732 Q 65.6334
GAZVKA P 110.464 Q 86.0714
NAGJS2 P 397.437 Q -41.6214
TRICH4 P 122.318 Q 40.0019
HUBLII P 170.840 Q 98.65
CUDAPH P 57.1499 Q 83.3083
NEYVLL P 251.9 Q 137.86
BELGUM P 224.586 Q 109.080
DAVGRE P 67.9404 Q 10.7775
NEYVLL P 172.92 Q -21.9557
SALEME P 118.415 Q 46.1621
MADUR4 P -46.2338 Q -10.9572
UDMPLT P -76.3952 Q 13.2961
BNGLOR P 476.305 Q 116.046
MUNRBD P 94.9931 Q 40.4092
HYDABD P 219.866 Q 128.971
TRICHN P 173.574 Q 30.2406
SHMOG4 P 75.4110 Q 14.7217
GOOTYY P 46.5738 Q 79.7417
SATPR2 P 199.6361 Q 47.64
KORBW2 P 107.467 Q -66.2437
BHLI40 P 322.924 Q 168.279
JBLPR4 P 78.6299 Q 0.937877
ITARS4 P 54.7943 Q 6.69695
INDR40 P 357.065 Q 129.786
BHPL40 P 217.341 Q 41.6994
KHPRKD P 311.866 Q 19.
KORD22 P 334.207 Q -3.0846
CHNPR2 P 303.617 Q 50.0
KARAD2 NO 1 P 191.85 Q 85.41
KARAD2 NO 2 P -232.402 Q -56.9
KOLPR2 NO 2 P 126. Q -6.4
KOLPR2 NO 1 P 213.52 Q 95.06
CHNW4 P 361.155 Q 62.3821
KLWA4 P 678.513 Q 7.27344
BLSWR4 P 267.607 Q 69.6942
BSWL40 P 230.177 Q 34.7787
WNKBR2 P 271.236 Q 18.7
UKA122 P 528.940 Q -6.0
ASOJ4 P 254.725 Q 6.73997
NRDPR4 P 421.725 Q 63.9703
JETPR4 P 247.859 Q 53.8415
SHRVT2 P 543.7 Q -33.0
PARL22 P 380.4988 Q 12.0
KRBAE2 P 515.472 Q 151.0
END
ASYNCHRONOUS MACHINES
MOT1 MADRAS SN .375 UN 400. H 19.4689
R1 .008 X1S .0767 X2S .0947
XM 4.2148 LOAD 1 RTAB 1 TYPE 1A
MOT2 MADRAS SN .95 UN 400. H 1.7453
R1 .0052 X1S .0732 X2S .1417
XM 7.3524 LOAD 2 RTAB 2 TYPE 1A
MOT3 MADRAS SN .95 UN 400. H 1.7453
R1 .0052 X1S .0732 X2S .1417
XM 7.3524 LOAD 2 RTAB 2 TYPE 1A
END
MLOADS
1 K 1.12 N 2 TYPE 0
2 K 1.03 N 2 TYPE 0
END
TABLES
1 TYPE 2 F 0.0 0.0083
1.0 0.0083
2 TYPE 2 F 0.0 0.0047
1.0 0.0047
END

```

SYNCHRONOUS MACHINES

BNGL1 BNGLORR TYPE 4 SN 1000. UN 13.8 RA 0. XD 0.9 XDP 0.23 XQ 0.9 H 2.95
 KORD1G KORD2 TYPE 4 SN 500. UN 15.75 RA 0. XD 0.9 XDP 0.23 XQ 0.9 H 2.95
 KLWAG KLWAA TYPE 4 SN 500. UN 13.8 RA 0. XD 0.9 XDP 0.23 XQ 0.9 H 2.95
 KORBI KORBA3 TYPE 4 SN 282. UN 13.8 RA 0. XD 0.9 XDP 0.23 XQ 0.9 H 2.95
 KORAD1 KORD1 TYPE 4 SN 408. UN 14.3 RA 0. XD 0.9 XDP 0.21 XQ 0.9 H 2.95
 UKAIGH UKAIHG TYPE 4 SN 375. UN 11. RA 0. XD 0.9 XDP 0.32 XQ 0.9 H 2.71
 KORBA2 KORBW TYPE 1A SN 247. UN 15.7 RA 0. XD 2.12 XDP 0.253 XQ 1.85 H 2.8
 D 0. XA 0.165 XDB 0.179 XQB 0.197 TDOP 7. TQOP 0.5 TDOB 0.04 TQOB 0.078
 VREG 0 XQP 0.506
 SATPRG STPRA1 TYPE 1A SN 247. UN 15.7 RA 0. XD 2.12 XDP 0.253 XQ 1.85 H 2.8
 D 0. XA 0.165 XDB 0.179 XQB 0.197 TDOP 7. TQOP 0.5 TDOB 0.04 TQOB 0.078
 VREG 0 XQP 0.506
 SATPUR STPRA2 TYPE 1A SN 247. UN 15.7 RA 0. XD 2.12 XDP 0.253 XQ 1.85 H 2.8
 D 0. XA 0.165 XDB 0.179 XQB 0.197 TDOP 7. TQOP 0.5 TDOB 0.04 TQOB 0.078
 VREG 0 XQP 0.506
 VINDYA VNLDHL2 TYPE 1A SN 988. UN 15.7 RA 0. XD 2.12 XDP 0.253 XQ 1.85 H 2.8
 D 0. XA 0.165 XDB 0.179 XQB 0.197 TDOP 7. TQOP 0.5 TDOB 0.04 TQOB 0.078
 VREG 0 XQP 0.506
 KORB2 KRB WG TYPE 1A SN 247. UN 15.7 RA 0. XD 2.12 XDP 0.253 XQ 1.85 H 2.8
 D 0. XA 0.165 XDB 0.179 XQB 0.197 TDOP 7. TQOP 0.5 TDOB 0.04 TQOB 0.078
 VREG 0 XQP 0.506
 KHPRKD KHPRKG TYPE 1A SN 494. UN 15.7 RA 0. XD 2.12 XDP 0.253 XQ 1.85 H 2.8
 D 0. XA 0.165 XDB 0.179 XQB 0.197 TDOP 7. TQOP 0.5 TDOB 0.04 TQOB 0.078
 VREG 0 XQP 0.506
 CHAND1 CNDRPR TYPE 1A SN 247. UN 15.7 RA 0. XD 2.12 XDP 0.253 XQ 1.85 H 2.8
 D 0. XA 0.165 XDB 0.179 XQB 0.197 TDOP 7. TQOP 0.5 TDOB 0.04 TQOB 0.078
 VREG 0 XQP 0.506
 PARLIG PARLI2 TYPE 1A SN 494. UN 15.7 RA 0. XD 2.12 XDP 0.253 XQ 1.85 H 2.8
 D 0. XA 0.165 XDB 0.179 XQB 0.197 TDOP 7. TQOP 0.5 TDOB 0.04 TQOB 0.078
 VREG 0 XQP 0.506
 CHAND2 CHNDP3 TYPE 1A SN 741. UN 15.7 RA 0. XD 2.12 XDP 0.253 XQ 1.85 H 2.8
 D 0. XA 0.165 XDB 0.179 XQB 0.197 TDOP 7. TQOP 0.5 TDOB 0.04 TQOB 0.078
 VREG 0 XQP 0.506
 WANOKB WNKBR1 TYPE 1A SN 247. UN 15.7 RA 0. XD 2.12 XDP 0.253 XQ 1.85 H 2.8
 D 0. XA 0.165 XDB 0.179 XQB 0.197 TDOP 7. TQOP 0.5 TDOB 0.04 TQOB 0.078
 VREG 0 XQP 0.506
 WANOK2 WNKBRG TYPE 1A SN 741. UN 15.7 RA 0. XD 2.12 XDP 0.253 XQ 1.85 H 2.8
 D 0. XA 0.165 XDB 0.179 XQB 0.197 TDOP 7. TQOP 0.5 TDOB 0.04 TQOB 0.078
 VREG 0 XQP 0.506
 UKAI UKAI3 TYPE 1A SN 470. UN 15.7 RA 0. XD 2.12 XDP 0.253 XQ 1.85 H 2.8
 D 0. XA 0.165 XDB 0.179 XQB 0.197 TDOP 7. TQOP 0.5 TDOB 0.04 TQOB 0.078
 VREG 0 XQP 0.506
 UKAI2 UKAI4G TYPE 1A SN 247. UN 15.7 RA 0. XD 2.12 XDP 0.253 XQ 1.85 H 2.8
 D 0. XA 0.165 XDB 0.179 XQB 0.197 TDOP 7. TQOP 0.5 TDOB 0.04 TQOB 0.078
 VREG 0 XQP 0.506
 VIJAY VIJW-G TYPE 1A SN 741. UN 15.7 RA 0. XD 2.12 XDP 0.253 XQ 1.85 H 2.8
 D 0. XA 0.165 XDB 0.179 XQB 0.197 TDOP 7. TQOP 0.5 TDOB 0.04 TQOB 0.078
 VREG 0 XQP 0.506
 NEVELI NEY2-G TYPE 1A SN 741. UN 15.7 RA 0. XD 2.12 XDP 0.253 XQ 1.85 H 2.8
 D 0. XA 0.165 XDB 0.179 XQB 0.197 TDOP 7. TQOP 0.5 TDOB 0.04 TQOB 0.078
 VREG 0 XQP 0.506
 RAMGND RMG2-G TYPE 1A SN 470. UN 15.7 RA 0. XD 2.12 XDP 0.253 XQ 1.85 H 2.8
 D 0. XA 0.165 XDB 0.179 XQB 0.197 TDOP 7. TQOP 0.5 TDOB 0.04 TQOB 0.078
 VREG 0 XQP 0.506
 KORBA4 KRB 2G TYPE 1A SN 470. UN 15.7 RA 0. XD 2.12 XDP 0.253 XQ 1.85 H 2.8
 D 0. XA 0.165 XDB 0.179 XQB 0.197 TDOP 7. TQOP 0.5 TDOB 0.04 TQOB 0.078
 VREG 0 XQP 0.506
 KORBAS KRB 5G TYPE 1A SN 1176. UN 21. RA 0. XD 2.35 XDP 0.270 XQ 2.15 H 3.07
 D 0. XA 0.16 XDB 0.190 XQB 0.210 TDOP 8.6 TQOP 2.5 TDOB 0.04 TQOB 0.08
 VREG 0 XQP 0.67
 RAMGN1 RMG5-G TYPE 1A SN 1176. UN 21. RA 0. XD 2.35 XDP 0.270 XQ 2.15 H 3.07
 D 0. XA 0.16 XDB 0.190 XQB 0.210 TDOP 8.6 TQOP 2.5 TDOB 0.04 TQOB 0.08
 VREG 0 XQP 0.506
 NEVEL NEY1-G TYPE 1A SN 352. UN 11.0 RA 0. XD 1.95 XDP 0.263 XQ 1.95 H 2.74
 D 0. XA 0.110 XDB 0.183 XQB 0.183 TDOP 6.50 TQOP 0.63 TDOB 0.04 TQOB 0.076
 VREG 0 XQP 0.750
 NAGARJ NAGS-G TYPE 2A SN 945. UN 11.0 RA 0. XD 0.9 XDP 0.23 XQ 0.59 H 4.0
 D 0. XA 0.089 XDB 0.185 XQB 0.185 TDOP 8.90 TDOB 0.04 TQOB 0.08 VREG 1
 SHARVT SHRT-G TYPE 2A SN 890. UN 11.0 RA 0. XD 0.9 XDP 0.27 XQ 0.59 H 3.8
 D 0. XA 0.09 XDB 0.185 XQB 0.185 TDOP 8.9 TDOB 0.03 TQOB 0.04 VREG 1
 KALIND KLNDS-G TYPE 2A SN 600. UN 11. RA 0. XD 0.91 XDP 0.24 XQ 0.54 H 4.4
 D 0. XA 0.09 XDB 0.145 XQB 0.145 TDOP 8.9 TDOB 0.03 TQOB 0.03 VREG 1
 END

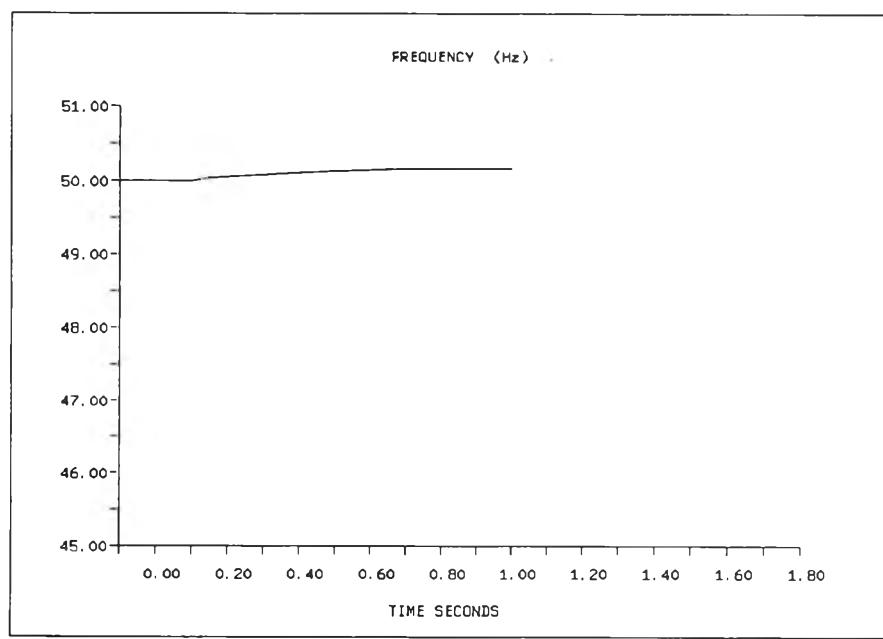
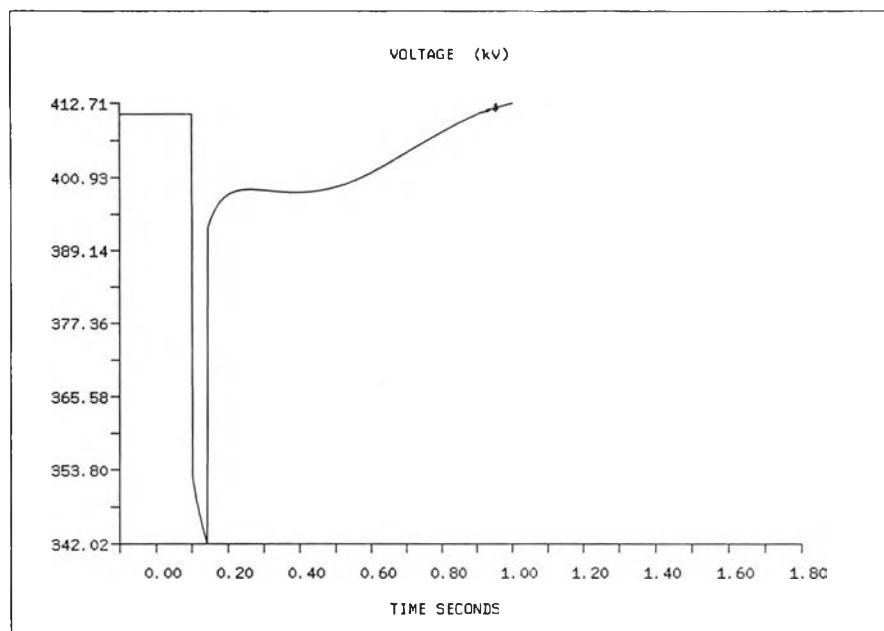
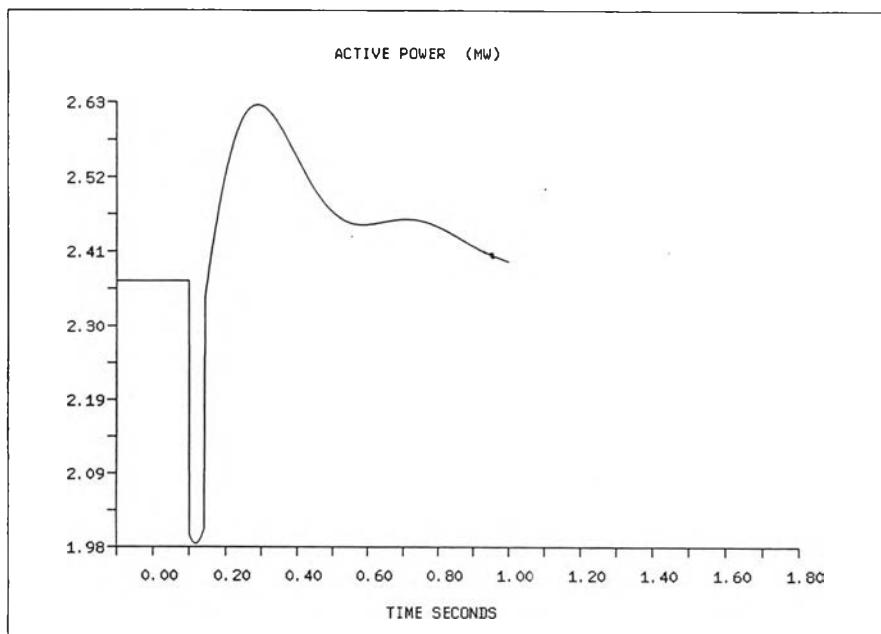


Fig. A6.2 Voltage and frequency at bus MADRAS when three phase fault occurred at bus CHNPR4

a.



b.

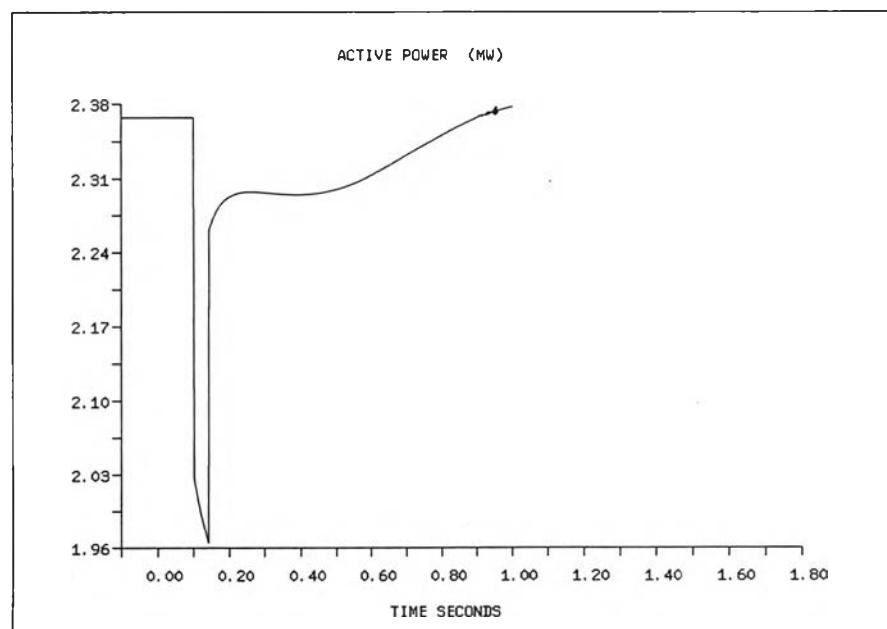
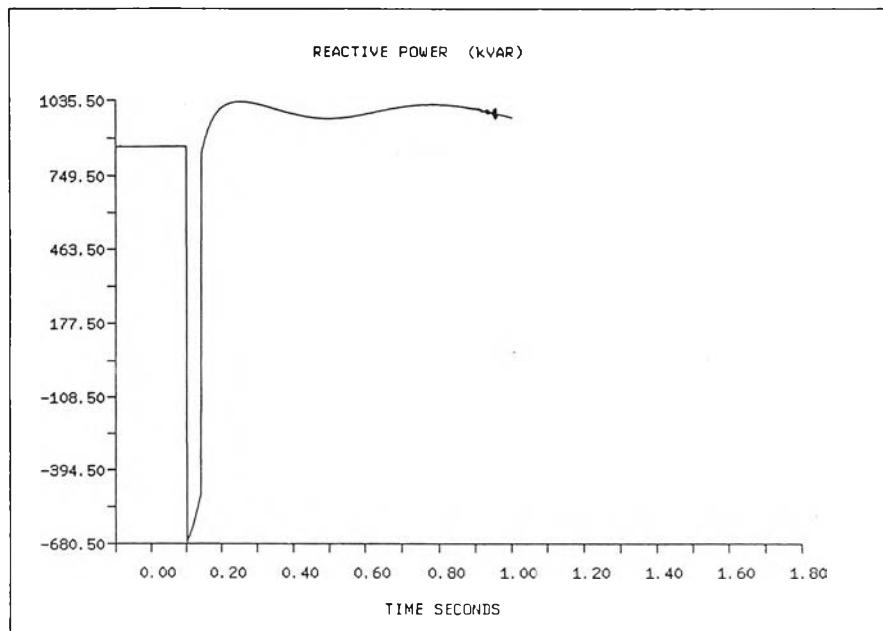


Fig. A6.3 Active power of motor loads at bus

MADRAS

- a. aggregate induction motor model
- b. constant current model

a.



b.

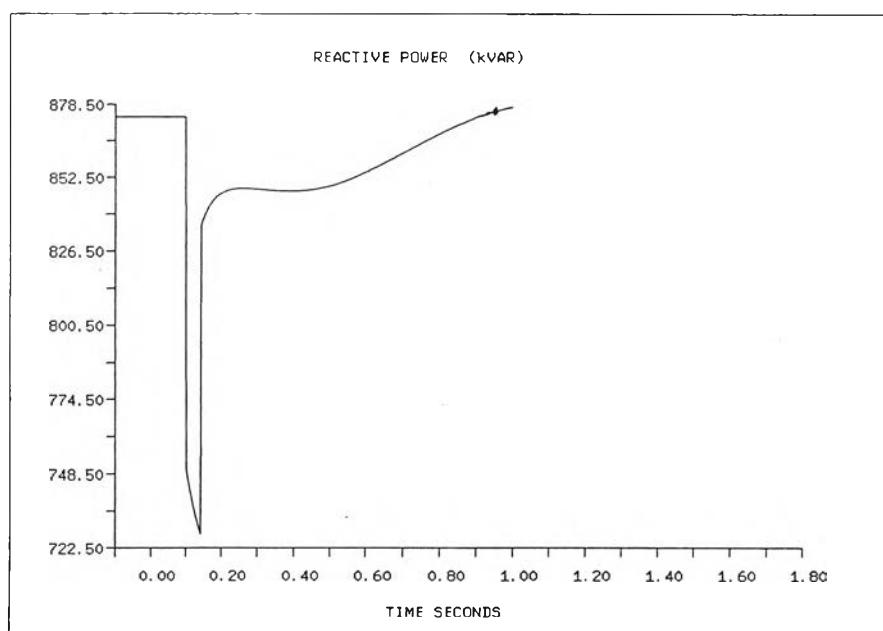


Fig. A6.4 Reactive power of motor loads at bus MADRAS

- a. aggregate induction motor model
- b. constant current model

Compare the simulation results obtained from the two models, large difference can be observed. This is because constant current model neglects the dynamic nature of motor loads while induction motor model includes.



VITA

The author, Mr. Somnuk Chayapornkul was born in Bangkok, Thailand on February 21, 1965. He received a Bachelor of Engineering degree in Electrical Engineering from Chulalongkorn University in 1986.